

QCX
Avro
CF105
R-7-0558-8



TECHNICAL REPORT



A. V. ROE CANADA LIMITED
MALTON - ONTARIO

ANALYZED

TECHNICAL DEPARTMENT (Aircraft)

AIRCRAFT: C-105

REPORT NO. 7-0558-8

FILE NO.

NO. OF SHEETS 77

TITLE:

~~CONFIDENTIAL~~

Classification cancelled / Changed to UNCLASS
By authority of AVES
Date 30 Sept 56
Signature J. Bully
Unit / Rank / Appointment NRSS

DEFLECTION FORMULAS

FORMER - SPAR - STRUT ASSEMBLY.



PREPARED BY V.J. Mason

DATE 5-17-55

CHECKED BY

DATE

SUPERVISED BY J. Bully

DATE MAY 55

APPROVED BY

DATE

ISSUE NO.	REVISION No.	REVISED BY	APPROVED BY	DATE	REMARKS

15865820



AIRCRAFT:

PREPARED BY

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5-17-55

CHECKED BY

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1, 16	Derivation of Deflection Formulas
17, 30	Sample Calculation
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C1, C12	Check Analysis & Calculation
F3, F15	Tabulated Formulas
31, 32	Alternate Spar Energy Function (Not used)

Calculations:

7425-F3, F15	Station 742.5	Case 1 (Realistic spar and struts)
7425-F9A, F15A	Station 742.5	Case 2 (Infinitely stiff spar and struts)
7425-F13B, F16B	Station 742.5	Case 3 (Infinitely stiff spar but no struts)



AVRO AIRCRAFT LIMITED

TECHNICAL DEPARTMENT (Aircraft)

REPORT No. 7-2558-8

SHEET No. Introduction

AIRCRAFT:

C-105

PREPARED BY

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6-10-55

CHECKED BY

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Introduction:

On page 29 the result for δ was first obtained in error; i.e., the coefficient of T_1 was smaller than the coefficient of T_2 . In searching for the truth a time came when we felt that the error might be due to the method of reducing the strain energy.

We proved that the method was correct and discovered that the error was arithmetic. We include the proof because it is worth something and ought to appear somewhere; if only to preclude the possibility of its being sought again.

M



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Derivation
of Formulas for
 δ_1 ; δ_2 ; δ_3 ; & δ_4

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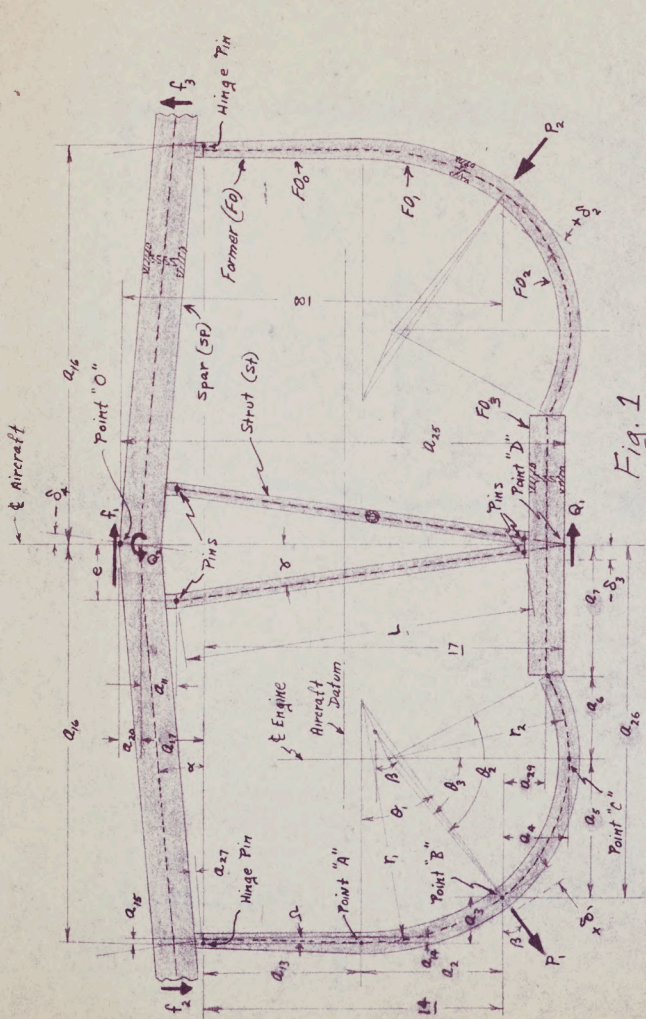


Fig. 1

Note: Forces are positive in the directions shown; P_1 & P_2 are produced by the longerons; Q_1 is produced by shear webs; Q_2 is produced by the fin and/or adjacent structure; f_1, f_2, f_3 are functions of the P 's and Q 's by statics. Second degree curves are approximated by radii r_1, r_2, r_3 through control points A, B, C, \dots . An underlined number is the name of a constant, e.g. $\underline{I} = S_1 = 19.09.89$, etc.



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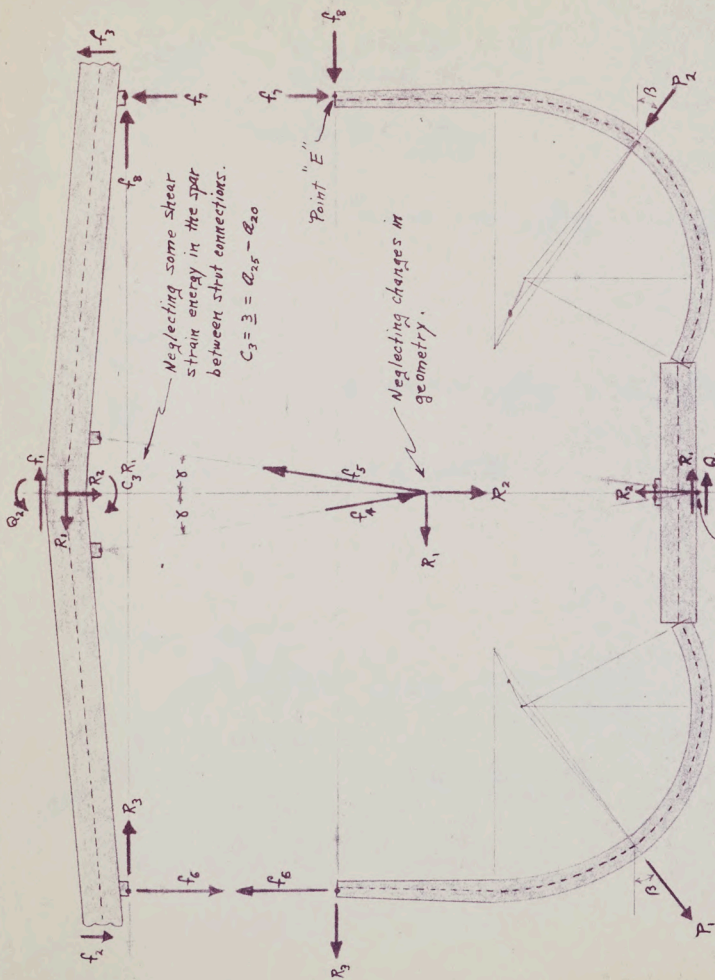


Fig. 2



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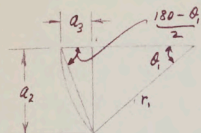
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Former Radii and Angles:

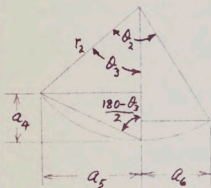


$$a_3 = r_1 (1 - \cos \theta_1)$$

$$a_2 = r_1 \sin \theta_1$$

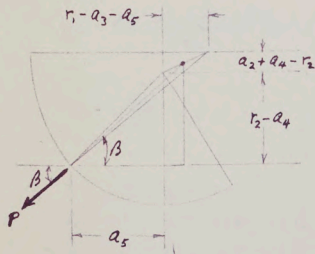
$$\sin \left(90 - \frac{\theta_1}{2} \right) = \frac{a_2}{\sqrt{a_2^2 + a_3^2}}$$

$$\theta_1 = 180 - 2 \sin^{-1} \frac{a_2}{\sqrt{a_2^2 + a_3^2}} ; r_1 = \frac{a_2}{\sin \theta_1}$$



$$\theta_2 = 180 - 2 \sin^{-1} \frac{a_4}{\sqrt{a_5^2 + a_6^2}} ; r_2 = \frac{a_4}{\sin \theta_2}$$

$$\theta_2 = \theta_3 + \sin^{-1} \frac{a_6}{r_2}$$



$$\beta = \tan^{-1} \frac{r_1 - a_4 + \frac{a_2 + a_4 - r_2}{2}}{a_5 + \frac{r_1 - a_3 - a_5}{2}}$$

$$= \tan^{-1} \frac{r_2 + a_2 - a_4}{r_1 + a_5 - a_3}$$



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Forces f_1 to f_8 by statics: P 's & Q 's are applied loads; R 's are redundants.

Referring to fig. 1,

$$F_1 = 0 = f_1 + Q_1 - P_1 \cos \beta - P_2 \cos \beta$$

$$F_2 = 0 = f_2 - f_3 - P_1 \sin \beta + P_2 \sin \beta$$

$$M_0 = 0 = a_{16} f_2 + Q_2 + a_{16} f_3 + a_{25} Q_1 - 8 P_1 \cos \beta - 8 P_2 \cos \beta + a_{26} P_1 \sin \beta + a_{26} P_2 \sin \beta$$

$$\epsilon = \cos \beta \quad ; \quad \eta = \sin \beta$$

$$f_1 = \epsilon P_1 + \epsilon P_2 - Q_1$$

$$f_2 - f_3 = -\eta P_1 + \eta P_2$$

$$f_2 + f_3 = \frac{68}{a_{16}} P_1 + \frac{68}{a_{16}} P_2 - \frac{7 a_{26}}{a_{16}} P_1 - \frac{7 a_{26}}{a_{16}} P_2 - \frac{a_{25}}{a_{16}} Q_1 - \frac{1}{a_{16}} Q_2$$

$$f_2 = \frac{1}{2 a_{16}} \left[P_1 (68 - 2 a_{26} - 7 a_{16}) + P_2 (68 - 2 a_{26} + 7 a_{16}) - a_{25} Q_1 - Q_2 \right]$$

$$f_3 = \frac{1}{2 a_{16}} \left[P_1 (68 - 2 a_{26} + 7 a_{16}) + P_2 (68 - 2 a_{26} - 7 a_{16}) - a_{25} Q_1 - Q_2 \right]$$

$$8 = a_{20} + a_{17} + a_2 + a_{13} \quad ; \quad 9 = a_3 - a_{14} + a_{15} \quad ; \quad a_{26} = a_{16} - 9$$

$$10 = \frac{68 - (a_{16} - 9) 7 - 7 a_{16}}{2 a_{16}} = \frac{68 + 79}{2 a_{16}} - 7 = 11 - 7$$

$$11 = \frac{68 - (a_{16} - 9) 7 + 7 a_{16}}{2 a_{16}} = \frac{68 + 79}{2 a_{16}}$$

$$12 = \frac{a_{25}}{2 a_{16}} \quad ; \quad 13 = \frac{1}{2 a_{16}} \quad ; \quad 14 = a_2 + a_{13} \quad ; \quad 15 = a_{26} + a_{16}$$

$$f_1 = \epsilon P_1 + \epsilon P_2 - Q_1$$

$$f_2 = 10 P_1 + 11 P_2 - 12 Q_1 - 13 Q_2$$

$$f_3 = 11 P_1 + 10 P_2 - 12 Q_1 - 13 Q_2$$

Referring to fig. 2,

$$\pm = \frac{1}{2 \cos \theta} \quad ; \quad \bar{\epsilon} = \frac{1}{2 \sin \theta}$$

$$f_5 - f_4 = \frac{R_2}{\cos \theta}$$

$$f_5 + f_4 = \frac{R_1}{\sin \theta}$$

$$f_4 = \bar{\epsilon} R_1 - \pm R_2$$

$$f_5 = \bar{\epsilon} R_1 + \pm R_2$$



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Referring to fig. 2,

$$F_H = 0 = -R_3 - f_8 - 6P_2 - 6P_1 + R_1 + Q_1$$

$$F_V = 0 = f_6 - f_7 + 7P_2 - 7P_1 + R_2$$

$$M_E = 0 = -20Q_1 f_6 - 146P_1 - 146P_2 + 157P_1 - 92P_2 - 416R_2 + 17R_1 + 17Q_1$$

$$2416f_6 = P_1(-614 + 715) + P_2(-614 - 79) + 17Q_1 + 17R_1 - 416R_2$$

$$\underline{18} = \frac{-614 + 715}{2416} ; \quad \underline{19} = -\frac{614 + 79}{2416} ; \quad \underline{20} = \frac{17}{2416}$$

$$\left. \begin{aligned} f_6 &= \underline{18}P_1 + \underline{19}P_2 + \underline{20}Q_1 + \underline{20}R_1 - .5R_2 \\ f_7 &= \underline{24}P_1 + \underline{23}P_2 + \underline{20}Q_1 + \underline{20}R_1 + .5R_2 \\ f_8 &= -6P_1 - 6P_2 + Q_1 + R_1 \end{aligned} \right\} -R_3$$

$$\underline{24} = \underline{18} - \underline{7} ; \quad \underline{23} = \underline{19} + \underline{7}$$

$$\left. \begin{aligned} f_9 &= f_2 + f_6 = \underline{49}P_1 + \underline{50}P_2 + \underline{51}Q_1 - \underline{13}Q_2 + \underline{20}R_1 - .5R_2 \\ f_{10} &= f_3 + f_7 = \underline{52}P_1 + \underline{53}P_2 + \underline{51}Q_1 - \underline{13}Q_2 + \underline{20}R_1 + .5R_2 \end{aligned} \right\}$$

$$\underline{49} = \underline{10} + \underline{18}$$

$$\underline{50} = \underline{11} + \underline{19}$$

$$\underline{51} = \underline{20} - \underline{12}$$

$$\underline{52} = \underline{11} + \underline{24}$$

$$\underline{53} = \underline{10} + \underline{23}$$



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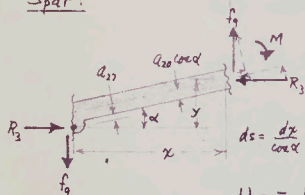
Strain Energy Function: "U"

M = Bending moment ; N = Axial force ; V = Shear force.

$$U = \int \frac{M^2}{2EI} ds + \int \frac{N^2}{2AE} ds + \int \frac{\alpha V^2}{2AG} ds ; \alpha = \frac{V_{max.}}{V_{ave.}}$$

Struts:M = 0 ; V = 0 ; N = f₄, f₅

$$\begin{aligned} U_{ST} &= \left[\frac{L}{2A_{ST} E_{ST}} \right] (f_4^2 + f_5^2) \\ &= 25 \left[(5R_1 - 4R_2)^2 + (5R_1 + 4R_2)^2 \right] \\ &= (2 \cdot 25 \cdot 5^2) R_1^2 + (2 \cdot 25 \cdot 4^2) R_2^2 \\ &= 26 R_1^2 + 27 R_2^2 \end{aligned}$$

Spar:

$$28 = \tan \alpha ; 29 = \sin \alpha ; 30 = \cos \alpha$$

$$y = \left(\frac{a_{27} + 30 a_{30}}{30} \right) + 28 x = 31 + 28 x$$

$$U_{SP} = U_{SP_{L.H.}} + U_{SP_{R.H.}}$$

Note: Symmetry is such that substitution of f₉ and f₉ for R₃ and f₉ respectively in U_{SP_{L.H.} yields U_{SP_{R.H.}}}

$$M = R_3 y + f_9 x = 31 R_3 + 28 R_3 x + f_9 x = 31 R_3 + (f_9 + 28 R_3) x$$

$$N = 30 R_3 - 29 f_9$$

$$V = 29 R_3 + 30 f_9$$

$$U_{SP_{L.H.}} = \int_0^{a_{16}} \left(\frac{1}{2 \cdot 30 E_{SP} I_{SP}} \right) \left(31^2 R_3^2 + 2 \cdot 31 R_3 (f_9 + 28 R_3) x + (f_9 + 28 R_3)^2 x^2 \right) dx + \left(\frac{a_{16}}{2 \cdot 30 A_{SP} E_{SP}} \right) N^2 + \left(\frac{a_{16} a_{16}}{2 \cdot 30 A_{SP} G_{SP}} \right) V^2$$



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$$U_{SP_{L.H.}} = (32 \overset{35}{3} a_1 R_3^2 + (32 \overset{36}{3} a_1^2 R_3 (f_0 + 28 R_3) + \frac{32 \overset{38}{3} a_1^3}{3} (f_0 + 28 R_3)^2 + 33 (20 R_3 - 29 f_0)^2 + 24 (29 R_3 + 20 f_0)^2) \\ = 35 R_3^2 + 36 R_3 f_0 + (32 \overset{37}{28} R_3^2 + 38 f_0^2 + (2 \overset{299}{38} 28) R_3 f_0 + (38 \overset{39}{28} 28) R_3^2 + (23 \overset{40}{38} 28) R_3^2 - (2 \overset{41}{33} 20 29) R_3 f_0 \\ + (33 \overset{42}{29} 29) f_0^2 + (34 \overset{43}{29} 29) R_3^2 + (2 \overset{44}{24} 29 \overset{45}{30}) R_3 f_0 + (24 \overset{45}{30} 29) f_0^2.$$

$$46 = 35 + 37 + 39 + 40 + 43$$

$$47 = 36 + 299 - 41 + 44$$

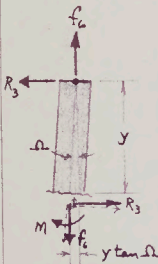
$$48 = 38 + 42 + 45$$

$$U_{SP} = 46 R_3^2 + 47 R_3 f_0 + 48 f_0^2 + 46 f_0^2 + 47 f_0 f_0 + 48 f_0^2$$

Former:

$$U_{FO} = U_{FO_{L.H.}} + U_{FO_{R.H.}}$$

Note: When $U_{FO_{L.H.}}$ is obtained as a function of R_3 ; f_0 ; and P ; symmetry allows $U_{FO_{R.H.}}$ to be written by substitution of f_2 ; f_1 ; & R_2 ; respectively.



Assume the pin is at the centroid of the section and that section properties do not vary along the length a_{13} .

$$\Omega = \tan^{-1} \frac{a_{13} - a_{15}}{a_{13}} = \tan^{-1} \frac{56}{57}; \quad 56 = \sin \Omega; \quad 57 = \cos \Omega$$

$$N = f_0 57 - R_3 56$$

$$V = R_3 57 + f_0 56$$

$$M = R_3 y + f_0 y \tan \Omega = (R_3 + 55 f_0) y; \quad ds = \frac{dy}{\cos \Omega}$$

$$U_{FO_{L.H.}} = \int_0^{a_{13}} \frac{(R_3 + 55 f_0)^2 y^2 dy}{2 E_{FO} I_{FO} 57} + \frac{(57 f_0 - 56 R_3)^2 a_{13}}{2 A_{FO} E_{FO} 57} + \frac{a_{FO} (57 R_3 + 56 f_0) a_{13}}{2 A_{FO} G_{FO} 57}$$

$$= \left(\frac{a_{13}^3}{6 E_{FO} I_{FO} 57} \right) (R_3^2 + 2 55 R_3 f_0 + 55^2 f_0^2) + \left(\frac{a_{13}}{2 A_{FO} E_{FO} 57} \right) (57^2 f_0^2 - 2 57 56 R_3 f_0 + 56^2 R_3^2) \\ + \left(\frac{a_{13} a_{FO}}{2 A_{FO} G_{FO} 57} \right) (57^2 R_3^2 + 2 57 56 R_3 f_0 + 56^2 f_0^2)$$

$$= R_3^2 (82 + 83 \overset{85}{56} 56^2 + 84 \overset{87}{57}) + R_3 f_0 (2 55 \overset{88}{82} - 2 56 \overset{89}{57} 83 + 2 56 \overset{89}{57} 84) + f_0^2 (82 \overset{89}{55} 55^2 + 83 \overset{89}{57} 57^2 + 84 \overset{89}{56} 56^2)$$



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$$U_{SP, L.H.} = \left(\overset{35}{32} \overset{31}{31} a_{13}^2 \right) R_3^2 + \left(\overset{36}{32} \overset{21}{21} a_{13}^2 \right) R_3 (f_9 + 28 R_3) + \left[\frac{\overset{38}{32} a_{13}^3}{3} \right] (f_9^2 + 28 R_3 f_9) + 33 (10 R_3 - 27 f_9^2) + 34 (27 R_3 + 30 f_9^2) \\ = 35 R_3^2 + 26 R_3 f_9 + \left(\overset{37}{34} 28 \right) R_3^2 + 38 f_9^2 + \left(\overset{39}{2} 38 \overset{28}{28} \right) R_3 f_9 + \left(\overset{39}{32} 28^2 \right) R_3^2 + \overset{40}{33} R_3^2 - \left(\overset{41}{2} 33 \overset{27}{27} \right) R_3 f_9 \\ + \left(\overset{42}{33} \overset{27}{27} \right) f_9^2 + \left(\overset{43}{34} 27^2 \right) R_3^2 + \left(\overset{44}{2} 34 \overset{29}{29} \overset{30}{30} \right) R_3 f_9 + \left(\overset{45}{34} 30^2 \right) f_9^2$$

$$46 = 35 + 37 + 38 + 40 + 43$$

$$47 = 36 + 299 - 41 + 44$$

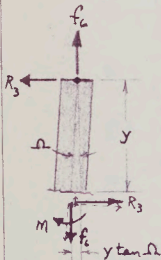
$$48 = 38 + 42 + 45$$

$$U_{SP} = 46 R_3^2 + 47 R_3 f_9 + 48 f_9^2 + 46 R_3^2 + 47 f_9^2 + 48 f_9^2$$

Former:

$$U_{FO} = U_{FO, L.H.} + U_{FO, R.H.}$$

Note: When $U_{FO, L.H.}$ is obtained as a function of R_3 ; f_6 ; and P_1 ; symmetry allows $U_{FO, R.H.}$ to be written by substitution of f_8 ; f_7 ; R_2 ; respectively.



Assume the pin is at the centroid of the section and that section properties do not vary along the length a_{13} .

$$\Omega = \tan^{-1} \frac{a_{13} - a_{13}}{a_{13}} = \tan^{-1} 55; \quad \underline{56} = \sin \Omega; \quad \underline{57} = \cos \Omega$$

$$N = f_6 \underline{57} - R_3 \underline{56}$$

$$V = R_3 \underline{57} + f_6 \underline{56}$$

$$M = R_3 y + f_6 y \tan \Omega = (R_3 + 55 f_6) y; \quad ds = \frac{dy}{\cos \Omega}$$

$$U_{FO, L.H.O} = \int_0^{a_{13}} \frac{(R_3 + 55 f_6)^2}{2 E_{FO} I_{FO} \underline{57}} dy + \frac{(57 f_6 - 56 R_3)^2 a_{13}}{2 A_{FO} E_{FO} \underline{57}} + \frac{a_{FO} (57 R_3 + 56 f_6)^2 a_{13}}{2 A_{FO} G_{FO} \underline{57}}$$

$$= \left(\frac{a_{13}^2}{6 E_{FO} I_{FO} \underline{57}} \right) (R_3^2 + 255 R_3 f_6 + 55^2 f_6^2) + \left(\frac{a_{13}}{2 A_{FO} E_{FO} \underline{57}} \right) (57^2 f_6^2 - 2 \underline{57} \underline{56} R_3 f_6 + 56^2 R_3^2)$$

$$+ \left(\frac{a_{13} a_{FO}}{2 A_{FO} G_{FO} \underline{57}} \right) (57^2 R_3^2 + 2 \underline{57} \underline{56} R_3 f_6 + 56^2 f_6^2)$$

$$= R_3^2 (82 + 83 \underline{56}^2 + 84 \underline{57}^2) + R_3 f_6 (2 \underline{55} \underline{82} - 2 \underline{56} \underline{57} \underline{83} + 2 \underline{56} \underline{57} \underline{84}) + f_6^2 (82 \underline{55}^2 + 83 \underline{57}^2 + 2 \underline{56} \underline{84})$$



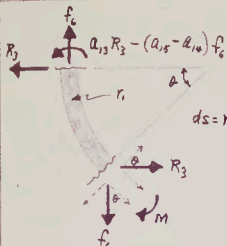
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$$N = R_3 \sin \alpha + f_c \cos \alpha$$

$$V = R_3 \cos \alpha - f_c \sin \alpha$$

$$M = a_{13} R_3 - (a_{15} - a_{14}) f_c + R_3 r_1 \sin \alpha - f_c r_1 (1 - \cos \alpha)$$

$$= u + R_3 r_1 \sin \alpha + f_c r_1 \cos \alpha$$

$$u = a_{13} R_3 - (r_1 + a_{15} - a_{14}) f_c$$

$$U_{F_0, M_1} = \int_0^{\theta_1} \left\{ \left(\frac{r_1^2}{2 E F_0} \right) \left(u^2 + R_3^2 r_1^2 \sin^2 \alpha + f_c^2 r_1^2 \cos^2 \alpha + 2u R_3 r_1 \sin \alpha + 2u f_c r_1 \cos \alpha + R_3 r_1^2 f_c \sin 2\alpha \right) \right. \\ \left. + \left(\frac{r_1^3}{2 A F_0 E F_0} \right) \left(R_3^2 \sin^2 \alpha + f_c^2 \cos^2 \alpha + R_3 f_c \sin 2\alpha \right) + \left(\frac{r_1^3}{2 A F_0 G I_0} \right) \left(f_c^2 \sin^2 \alpha + R_3^2 \cos^2 \alpha - R_3 f_c \sin 2\alpha \right) \right\} d\alpha$$

$$90 = \int_0^{\theta_1} \sin^2 \alpha d\alpha = \frac{\alpha}{2} - \frac{\sin 2\alpha}{4} \Big|_0^{\theta_1} = \frac{1}{2} \left(\theta_1 - \frac{\sin 2\theta_1}{2} \right)$$

$$91 = \int_0^{\theta_1} \cos^2 \alpha d\alpha = \frac{\alpha}{2} + \frac{\sin 2\alpha}{4} \Big|_0^{\theta_1} = \frac{1}{2} \left(\theta_1 + \frac{\sin 2\theta_1}{2} \right)$$

$$92 = \int_0^{\theta_1} \sin \alpha d\alpha = -\cos \alpha \Big|_0^{\theta_1} = 1 - \cos \theta_1$$

$$93 = \int_0^{\theta_1} \cos \alpha d\alpha = \sin \alpha \Big|_0^{\theta_1} = \sin \theta_1$$

$$94 = \int_0^{\theta_1} \sin 2\alpha d\alpha = -\frac{1}{2} \cos 2\alpha \Big|_0^{\theta_1} = \frac{1}{2} (1 - \cos 2\theta_1)$$

$$U_{F_0, M_1} = 87 \theta_1 u^2 + 90 r_1^2 R_3^2 + 91 r_1^2 f_c^2 + 2 92 r_1 R_3 u + 2 93 r_1 f_c u + 94 r_1^2 R_3 f_c \\ + 88 \left(90 R_3^2 + 91 f_c^2 + 94 R_3 f_c \right) + 89 \left(90 f_c^2 + 91 R_3^2 - 94 R_3 f_c \right)$$

$$= \left(87 \theta_1 a_{13}^2 \right) R_3^2 - \left(2 87 \theta_1 a_{13} a_{14} \right) R_3 f_c + \left(87 \theta_1 a_{14}^2 \right) f_c^2 + \left(87 90 r_1^2 \right) R_3^2 + \left(87 91 r_1^2 \right) f_c^2 \\ + \left(2 87 92 r_1 a_{13} \right) R_3^2 - \left(2 87 92 r_1 a_{14} \right) R_3 f_c + \left(2 87 93 r_1 a_{13} \right) R_3 f_c - \left(2 87 93 r_1 a_{14} \right) f_c^2 + \left(87 94 r_1^2 \right) R_3 f_c \\ + \left(88 90 \right) R_3^2 + \left(88 91 \right) f_c^2 + \left(88 94 \right) R_3 f_c + \left(89 90 \right) f_c^2 + \left(89 91 \right) R_3^2 - \left(89 94 \right) R_3 f_c$$



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$$U_{F_0, H_1} = \bar{r}_3^2 \left(\frac{95}{95} + \frac{98}{100} + \frac{105}{105} + \frac{109}{109} \right) + \bar{r}_3^2 \left(\frac{-96}{-96} - \frac{101}{101} + \frac{102}{102} + \frac{104}{104} + \frac{107}{107} - \frac{110}{110} \right) \\ + \bar{r}_6^2 \left(\frac{97}{97} + \frac{99}{99} - \frac{103}{103} + \frac{106}{106} + \frac{108}{108} \right).$$

$$M_0 = \bar{r}_3 (a_{13} + a_{14}) - \bar{r}_6 (a_{15} - a_{14} + a_{13})$$

$$\underline{6} = \cos \beta ; \quad \underline{7} = \sin \beta ; \quad \underline{114} = \cos \beta_2 ; \quad \underline{115} = \sin \beta_2$$



$$N_0 = (\bar{r}_6 - \bar{r}_1 \sin \beta) \sin \beta_2 + (\bar{r}_3 + \bar{r}_1 \cos \beta) \cos \beta_2$$

$$V_0 = (\bar{r}_6 - \bar{r}_1 \sin \beta) \cos \beta_2 - (\bar{r}_3 + \bar{r}_1 \cos \beta) \sin \beta_2$$

$$N_0 = \underline{115} (\bar{r}_6 - \underline{7} \bar{r}_1) + \underline{114} (\bar{r}_3 + \underline{6} \bar{r}_1)$$

$$V_0 = \underline{114} (\bar{r}_6 - \underline{7} \bar{r}_1) - \underline{115} (\bar{r}_3 + \underline{6} \bar{r}_1)$$

$$N_0 = \underline{115} \mu + \underline{114} \nu$$

$$V_0 = \underline{114} \mu - \underline{115} \nu$$

$$N = N_0 \cos \theta - V_0 \sin \theta$$

$$V = N_0 \sin \theta + V_0 \cos \theta$$

$$M = M_0 - V_0 \bar{r}_2 \sin \theta - N_0 \bar{r}_2 (1 - \cos \theta) = (M_0 - N_0 \bar{r}_2) - V_0 \bar{r}_2 \sin \theta + N_0 \bar{r}_2 \cos \theta$$

$$U_{F_0, H_2} = \int_0^{\beta_2} \left\{ \left(\frac{\bar{r}_3}{2 E_{F_0} I_{F_0}} \right) \left[\omega^2 + V_0^2 \bar{r}_2^2 \sin^2 \theta + N_0^2 \bar{r}_2^2 \cos^2 \theta - 2 \omega V_0 \bar{r}_2 \sin \theta + 2 \omega N_0 \bar{r}_2 \cos \theta - V_0 N_0 \bar{r}_2^2 \sin 2 \theta \right] \right. \\ \left. + \left(\frac{\bar{r}_1}{2 A_{F_0} E_{F_0}} \right) \left[V_0^2 \sin^2 \theta + N_0^2 \cos^2 \theta - N_0 V_0 \sin 2 \theta \right] \right. \\ \left. + \left(\frac{\bar{r}_2 A_{F_0}}{2 A_{F_0} I_{F_0}} \right) \left[N_0^2 \sin^2 \theta + V_0^2 \cos^2 \theta + N_0 V_0 \sin 2 \theta \right] \right\} d\theta$$

$$\underline{119} = \int_0^{\beta_2} \sin^2 \theta d\theta = \frac{1}{2} (\beta_2 - \frac{\sin 2\beta_2}{2})$$

$$\underline{120} = \int_0^{\beta_2} \cos^2 \theta d\theta = \frac{1}{2} (\beta_2 + \frac{\sin 2\beta_2}{2})$$

$$\underline{121} = \int_0^{\beta_2} \sin \theta d\theta = 1 - \cos \beta_2$$

$$\underline{122} = \int_0^{\beta_2} \cos \theta d\theta = \sin \beta_2$$

$$\underline{123} = \int_0^{\beta_2} \sin 2\theta d\theta = \frac{1}{2} (1 - \cos 2\beta_2)$$



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$$U_{FO, L_{H_2}} = \frac{116}{2} (4\alpha^2 B_2 + 119 V_0^2 f_2^2 + 120 N_0^2 f_2^2 - 2 \cdot 121 V_0 f_2 4\alpha + 2 \cdot 122 N_0 f_2 4\alpha - 123 V_0 N_0 f_2^2) \\ + \frac{117}{2} (119 V_0^2 + 120 N_0^2 - 123 N_0 V_0) + \frac{118}{2} (119 N_0^2 + 120 V_0^2 + 123 N_0 V_0).$$

$$= \frac{124}{2} M_0^2 - \frac{125}{2} M_0 N_0 + \frac{126}{2} N_0^2 + \frac{127}{2} V_0^2 + \frac{128}{2} N_0^2 - \frac{129}{2} M_0 V_0 \\ + (2 \cdot \frac{130}{116} 121 f_2^2) N_0 V_0 + (2 \cdot \frac{131}{116} 122 f_2) M_0 N_0 - (2 \cdot \frac{132}{116} 123 f_2^2) N_0^2 - (\frac{133}{116} 123 f_2^2) N_0 V_0 + (\frac{134}{117} 119) V_0^2 \\ + (\frac{135}{117} 120) N_0^2 - (\frac{136}{117} 123) N_0 V_0 + (\frac{137}{118} 119) N_0^2 + (\frac{138}{118} 120) V_0^2 + (\frac{139}{118} 123) N_0 V_0.$$

$$= M_0^2 (124) + N_0^2 (\frac{126}{2} + \frac{128}{2} - \frac{132}{2} + \frac{135}{2} + \frac{137}{2}) + V_0^2 (\frac{127}{2} + \frac{134}{2} + \frac{138}{2}) + M_0 N_0 (-\frac{125}{2} + \frac{131}{2}) \\ + M_0 V_0 (-\frac{129}{2}) + N_0 V_0 (\frac{130}{2} - \frac{133}{2} - \frac{136}{2} + \frac{139}{2}).$$

$$= 124 (14^2 R_3^2 + 9^2 f_6^2 - 2 \cdot 9 \cdot 14 R_3 f_6) + 140 (115^2 u^2 + 114^2 v^2 + 2 \cdot 114 \cdot 115 u v) \\ + 141 (114^2 u^2 + 115^2 v^2 - 2 \cdot 114 \cdot 115 u v) + 142 (14 \cdot 115 R_3 u + 14 \cdot 114 R_3 v - 9 \cdot 115 f_6 u - 9 \cdot 114 f_6 v) \\ - 129 (14 \cdot 114 R_3 u - 14 \cdot 115 R_3 v - 9 \cdot 114 f_6 u + 9 \cdot 115 f_6 v) + 143 (114 \cdot 115 u^2 + [114^2 - 115^2] uv - 114 \cdot 115 v^2).$$

$$= (124 \cdot 14^2) R_3^2 + (124 \cdot 9^2) f_6^2 - (2 \cdot 124 \cdot 9 \cdot 14) R_3 f_6 + (f_6^2 \cdot 2 \cdot 7 \cdot f_6^2 + 7^2 f_6^2) (140 \cdot 115^2 + 141 \cdot 114^2 + 143 \cdot 114 \cdot 115) \\ + (R_3^2 + 2 \cdot 6 R_3 f_6 + 6^2 f_6^2) (140 \cdot 114^2 + 141 \cdot 115^2 - 143 \cdot 114 \cdot 115)$$

$$+ (R_3 f_6 + 6 f_6^2 - 7 R_3 f_6 - 6 \cdot 7 f_6^2) (2 \cdot 140 \cdot 114 \cdot 115 - 2 \cdot 141 \cdot 114 \cdot 115 + 143 \cdot [114^2 - 115^2])$$

$$+ (R_3^2 u^2 - 2 R_3 f_6) (182 \cdot 14 \cdot 115 - 129 \cdot 14 \cdot 114) + (R_3^2 v^2 + 6 R_3 f_6) (182 \cdot 14 \cdot 114 + 129 \cdot 14 \cdot 115)$$

$$+ (f_6^2 u^2 - 7 f_6^2 v) (-142 \cdot 9 \cdot 115 + 129 \cdot 9 \cdot 114) + (R_3 f_6 + 6 f_6^2) (-182 \cdot 9 \cdot 114 - 129 \cdot 9 \cdot 115).$$

$$= R_3^2 (144 + 154 + 164) + f_6^2 (145 + 150 + 167) + R_3^2 (2^2 \cdot 150 + 6^2 \cdot 154 - 6 \cdot 7 \cdot 158)$$

$$+ R_3 f_6 (-146 + 158 + 161 + 170) + R_3^2 (2 \cdot 6 \cdot 154 - 7 \cdot 158 - 7 \cdot 161 + 6 \cdot 164)$$

$$+ f_6^2 (-2 \cdot 7 \cdot 150 + 6 \cdot 158 - 7 \cdot 167 + 6 \cdot 170).$$



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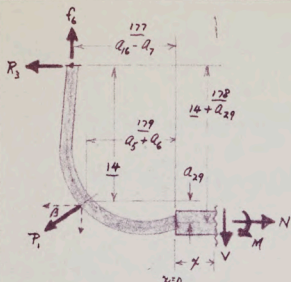
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$$N = R_3 + P_1 \cos \beta = R_3 + \epsilon P_1$$

$$V = f_6 - P_1 \sin \beta = f_6 - 2 P_1$$

$$M = 128 R_3 + a_{29} \epsilon P_1 - f_6 (127 + x) + 2 P_1 (129 + x)$$

$$= [128 R_3 + (a_{29} \epsilon + 129) P_1 - 127 f_6] - (f_6 - 2 P_1) x$$

$$= (128 R_3 + 180 P_1 - 127 f_6) - (f_6 - 2 P_1) x$$

$$U_{FOLM_2} = \int_0^{a_2} \left(\frac{1}{2 A_{F0} I_{F0}} \right) \left[u^2 - 2 u V x + V^2 x^2 \right] dx + \left(\frac{182}{2 A_{F0} E_{F0}} \right) N^2 + \left(\frac{183}{2 A_{F0} G_{F0}} \right) V^2$$

$$= (181 a_2) u^2 - (181 a_2^2) u V + \left(\frac{181 a_2^3}{3} + 183 \right) V^2 + 182 N^2$$

$$= 184 (128^2 R_3^2 + 180^2 P_1^2 + 127^2 f_6^2 + 2 \cdot 128 \cdot 180 R_3 P_1 - 2 \cdot 128 \cdot 127 R_3 f_6 - 2 \cdot 180 \cdot 127 f_6 P_1) \\ - 185 (128 R_3 f_6 + 180 f_6 P_1 - 127 f_6^2 - 2 \cdot 128 R_3 P_1 - 2 \cdot 180 P_1^2 + 2 \cdot 127 f_6 P_1) \\ + 186 (f_6^2 - 2 \cdot 2 f_6 P_1 + 2^2 P_1^2) + 183 (R_3^2 + 2 \epsilon R_3 P_1 + \epsilon^2 P_1^2)$$

$$= \left(\frac{184}{184} 128^2 \right) R_3^2 + \left(\frac{184}{184} 180^2 \right) P_1^2 + \left(\frac{184}{184} 127^2 \right) f_6^2 + \left(\frac{184}{184} 2 \cdot 128 \cdot 180 \right) R_3 P_1 - \left(\frac{184}{184} 2 \cdot 128 \cdot 127 \right) R_3 f_6$$

$$- \left(\frac{185}{185} 2 \cdot 128 \cdot 180 \right) f_6 P_1 - \left(\frac{185}{185} 128^2 \right) R_3 f_6 - \left(\frac{185}{185} 180^2 \right) f_6 P_1 + \left(\frac{185}{185} 127 \cdot 128 \right) R_3 P_1$$

$$+ \left(\frac{185}{185} 2 \cdot 127 \right) P_1^2 - \left(\frac{185}{185} 2 \cdot 127 \right) f_6 P_1 + \left(\frac{186}{186} f_6^2 \right) - \left(\frac{186}{186} 2 \right) f_6 P_1 + \left(\frac{186}{186} 2^2 \right) P_1^2 + \left(\frac{182}{182} \right) R_3^2$$

$$+ \left(\frac{182}{182} 2 \right) R_3 P_1 + \left(\frac{182}{182} \epsilon^2 \right) P_1^2$$

$$= R_3^2 (187 + 182) + P_1^2 (188 + 187 + 200 + 202) + f_6^2 (189 + 195 + 186)$$

$$+ R_3 P_1 (190 + 196 + 201) + R_3 f_6 (-191 - 192) + f_6 P_1 (-192 - 194 - 198 - 199)$$



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$$\begin{aligned}
 U_{F_{0LH}} &= U_{F_{0LH_0}} + U_{F_{0LH_1}} + U_{F_{0LH_2}} + U_{F_{0LH_3}} \\
 &= R_3^2 (85 + \overset{209}{111} + \overset{209}{171} + \overset{209}{202}) + P_1^2 (\overset{210}{173} + \overset{210}{204}) + f_6^2 (\overset{211}{59} + \overset{211}{113} + \overset{211}{172} + \overset{211}{205}) \\
 &\quad + R_3 P_1 (\overset{212}{125} + \overset{212}{206}) + R_3 f_6 (\overset{213}{112} + \overset{213}{58} + \overset{213}{174} + \overset{213}{207}) + f_6 P_1 (\overset{214}{176} + \overset{214}{208}).
 \end{aligned}$$

$$\begin{aligned}
 U_{F_0} &= U_{F_{0LH}} + U_{F_{0RH}} \\
 &= 209 R_3^2 + 209 f_8^2 + 210 P_1^2 + 210 P_2^2 + 211 f_6^2 + 211 f_7^2 + 212 R_3 P_1 + 212 f_6 P_2 \\
 &\quad + 212 R_3 f_6 + 213 f_7 f_8 + 214 f_6 P_1 + 214 f_7 P_2.
 \end{aligned}$$

$$U = U_{ST} + U_{SP} + U_{F_0} \quad 300 = 96 + 209$$

$$\begin{aligned}
 &= 210 P_1^2 + 210 P_2^2 + 25 R_1^2 + 27 R_2^2 + 300 R_3^2 + 211 f_6^2 + 211 f_7^2 + 300 f_8^2 + 48 f_9^2 + 48 f_{10}^2 \\
 &\quad + 212 P_1 R_3 + 214 P_1 f_6 + 214 R_2 f_7 + 212 f_7 f_8 + 213 R_3 f_6 + 47 R_3 f_9 + 213 f_7 f_8 + 47 f_8 f_{10}.
 \end{aligned}$$

$$\begin{aligned}
 \frac{\partial U}{\partial X} &= 2 \overset{301}{210} P_1 \frac{\partial P_1}{\partial X} + 2 \overset{301}{210} P_2 \frac{\partial P_2}{\partial X} + 2 \overset{302}{26} R_1 \frac{\partial R_1}{\partial X} + 2 \overset{302}{27} R_2 \frac{\partial R_2}{\partial X} + 2 \overset{304}{300} R_3 \frac{\partial R_3}{\partial X} + 2 \overset{305}{211} f_6 \frac{\partial f_6}{\partial X} \\
 &\quad + 2 \overset{305}{211} f_7 \frac{\partial f_7}{\partial X} + 2 \overset{304}{300} f_8 \frac{\partial f_8}{\partial X} + 2 \overset{306}{48} f_9 \frac{\partial f_9}{\partial X} + 2 \overset{306}{48} f_{10} \frac{\partial f_{10}}{\partial X} + 212 R_3 \frac{\partial P_1}{\partial X} + 212 P_2 \frac{\partial R_3}{\partial X} \\
 &\quad + 214 f_6 \frac{\partial P_1}{\partial X} + 214 P_1 \frac{\partial f_6}{\partial X} + 214 R_2 \frac{\partial f_7}{\partial X} + 214 P_2 \frac{\partial R_3}{\partial X} + 212 f_7 \frac{\partial f_8}{\partial X} + 212 P_2 \frac{\partial f_8}{\partial X} + 213 f_8 \frac{\partial R_3}{\partial X} \\
 &\quad + 213 R_3 \frac{\partial f_6}{\partial X} + 47 R_3 \frac{\partial f_9}{\partial X} + 47 R_3 \frac{\partial f_9}{\partial X} + 213 f_7 \frac{\partial f_8}{\partial X} + 213 f_7 \frac{\partial f_8}{\partial X} + 47 f_{10} \frac{\partial f_8}{\partial X} + 47 f_8 \frac{\partial f_{10}}{\partial X}.
 \end{aligned}$$

We treat the partial derivatives, $\frac{\partial P_i}{\partial X}$, $\frac{\partial Q_i}{\partial X}$, and $\frac{\partial X_i}{\partial X}$ as unity when X is P_i , Q_i , and R_i respectively; but as zero when X has other values. $\frac{\partial f_i}{\partial X}$ are constants; and hence $\frac{\partial}{\partial Y} \left(\frac{\partial f_i}{\partial X} \right) = 0$.



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$$\begin{aligned} \frac{\partial^2 U}{\partial X \partial Y} = & \frac{\partial P_1}{\partial Y} \left[301 \frac{\partial P_1}{\partial X} + 212 \frac{\partial P_2}{\partial X} + 214 \frac{\partial f_1}{\partial X} \right] + \frac{\partial P_2}{\partial Y} \left[301 \frac{\partial P_2}{\partial X} + 214 \frac{\partial f_1}{\partial X} + 212 \frac{\partial f_2}{\partial X} \right] \\ & + \frac{\partial P_3}{\partial Y} \left[302 \frac{\partial P_1}{\partial X} \right] + \frac{\partial P_2}{\partial Y} \left[303 \frac{\partial P_2}{\partial X} \right] + \frac{\partial P_3}{\partial Y} \left[304 \frac{\partial P_3}{\partial X} + 212 \frac{\partial P_1}{\partial X} + 213 \frac{\partial f_1}{\partial X} + 47 \frac{\partial f_2}{\partial X} \right] \\ & + \frac{\partial f_1}{\partial Y} \left[305 \frac{\partial f_1}{\partial X} + 214 \frac{\partial P_1}{\partial X} + 213 \frac{\partial P_2}{\partial X} \right] + \frac{\partial f_2}{\partial Y} \left[305 \frac{\partial f_2}{\partial X} + 214 \frac{\partial P_2}{\partial X} + 213 \frac{\partial f_1}{\partial X} \right] \\ & + \frac{\partial f_3}{\partial Y} \left[304 \frac{\partial f_1}{\partial X} + 212 \frac{\partial P_2}{\partial X} + 213 \frac{\partial f_1}{\partial X} + 47 \frac{\partial f_2}{\partial X} \right] + \frac{\partial f_4}{\partial Y} \left[306 \frac{\partial f_2}{\partial X} + 47 \frac{\partial P_3}{\partial X} \right] \\ & + \frac{\partial f_5}{\partial Y} \left[306 \frac{\partial f_3}{\partial X} + 47 \frac{\partial f_2}{\partial X} \right]. \end{aligned}$$

Referring to the formulas for the f 's,

$\frac{\partial f_1}{\partial P_1} = 18$	$\frac{\partial f_1}{\partial P_2} = 24$	$\frac{\partial f_1}{\partial P_3} = -6$	$\frac{\partial f_1}{\partial Q_1} = 49$	$\frac{\partial f_1}{\partial P_1} = 52$
$\frac{\partial f_2}{\partial P_2} = 19$	$\frac{\partial f_2}{\partial P_2} = 23$	$\frac{\partial f_2}{\partial P_2} = -6$	$\frac{\partial f_2}{\partial P_2} = 50$	$\frac{\partial f_2}{\partial P_2} = 53$
$\frac{\partial f_3}{\partial Q_1} = 20$	$\frac{\partial f_3}{\partial Q_1} = 20$	$\frac{\partial f_3}{\partial Q_1} = 1$	$\frac{\partial f_3}{\partial Q_1} = 51$	$\frac{\partial f_3}{\partial Q_1} = 51$
$\frac{\partial f_4}{\partial Q_2} = 0$	$\frac{\partial f_4}{\partial Q_2} = 0$	$\frac{\partial f_4}{\partial Q_2} = 0$	$\frac{\partial f_4}{\partial Q_2} = -13$	$\frac{\partial f_4}{\partial Q_2} = -13$
$\frac{\partial f_5}{\partial P_1} = 20$	$\frac{\partial f_5}{\partial P_1} = 20$	$\frac{\partial f_5}{\partial P_1} = 1$	$\frac{\partial f_5}{\partial P_1} = 20$	$\frac{\partial f_5}{\partial P_1} = 20$
$\frac{\partial f_6}{\partial P_2} = -\frac{1}{2}$	$\frac{\partial f_6}{\partial P_2} = \frac{1}{2}$	$\frac{\partial f_6}{\partial P_2} = 0$	$\frac{\partial f_6}{\partial P_2} = -\frac{1}{2}$	$\frac{\partial f_6}{\partial P_2} = \frac{1}{2}$
$\frac{\partial f_7}{\partial P_3} = 0$	$\frac{\partial f_7}{\partial P_3} = 0$	$\frac{\partial f_7}{\partial P_3} = -1$	$\frac{\partial f_7}{\partial P_3} = 0$	$\frac{\partial f_7}{\partial P_3} = 0$

Introducing the notations,

$$\underline{UXY} = \frac{\partial^2 U}{\partial X \partial Y} = \underline{UYX}$$

$$\underline{XY} = \frac{\partial X}{\partial Y}$$



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$$UP_1 = \delta_1 = UP_1^2 P_1 + UP_1 R_1 + UP_1 Q_1 + UP_1 Q_2 Q_2 + UP_1 R_1 R_1 + UP_1 R_2 R_2 + UP_1 R_3 R_3$$

$$UP_2 = \delta_2 = UP_2^2 P_2 + UP_2 Q_1 + UP_2 Q_2 Q_2 + UP_2 R_1 R_1 + UP_2 R_2 R_2 + UP_2 R_3 R_3$$

$$UP_3 = \delta_3 = UP_3^2 P_3 + UP_3 R_2 R_2 + UP_3 Q_1 Q_1 + UP_3 Q_2 Q_2 + UP_3 R_1 R_1 + UP_3 R_2 R_2 + UP_3 R_3 R_3$$

$$UP_4 = \delta_4 = UP_4^2 P_4 + UP_4 R_2 R_2 + UP_4 Q_1 Q_1 + UP_4 Q_2 Q_2 + UP_4 R_1 R_1 + UP_4 R_2 R_2 + UP_4 R_3 R_3$$

$$UR_1 = 0 = UR_1^2 P_1 + UR_1 R_2 R_2 + UR_1 Q_1 Q_1 + UR_1 Q_2 Q_2 + UR_1^2 R_1 + UR_1 R_2 R_2 + UR_1 R_3 R_3$$

$$UR_2 = 0 = UR_2^2 P_2 + UR_2 R_2 R_2 + UR_2 Q_1 Q_1 + UR_2 Q_2 Q_2 + UR_2^2 R_1 + UR_2^2 R_2 + UR_2 R_3 R_3$$

$$UR_3 = 0 = UR_3^2 P_3 + UR_3 R_2 R_2 + UR_3 Q_1 Q_1 + UR_3 Q_2 Q_2 + UR_3 R_1 R_1 + UR_3 R_2 R_2 + UR_3^2 R_3$$

$$UP_X = 301 P_X + 212 R_3 X + 214 f_6 X + 18 305 f_6 X + 18 214 P_X + 18 213 R_3 X + 24 305 f_6 X + 24 214 P_X \\ + 24 213 f_6 X - 6 304 f_6 X - 6 212 P_X - 6 213 R_X - 6 213 f_6 X - 6 47 f_6 X + 49 306 f_6 X + 49 47 P_X + 52 306 f_6 X + 52 47 f_6 X$$

$$UR_X = 301 P_X + 214 f_6 X + 212 f_6 X + 19 305 f_6 X + 19 214 P_X + 19 213 R_3 X + 23 305 f_6 X + 23 214 P_X + 23 213 f_6 X \\ - 6 304 f_6 X - 6 212 P_X - 6 213 f_6 X - 6 47 f_6 X + 50 306 f_6 X + 50 47 P_X + 53 306 f_6 X + 53 47 f_6 X$$

$$UQ_X = 20 305 f_6 X + 20 214 P_X + 20 213 R_3 X + 20 305 f_6 X + 20 214 P_X + 20 213 f_6 X + 304 f_6 X + 212 P_X \\ + 213 f_6 X + 47 f_6 X + 51 306 f_6 X + 51 47 P_X + 51 306 f_6 X + 51 47 f_6 X$$

$$UQ_2 X = -13 306 f_6 X - 13 47 P_X - 13 306 f_6 X - 13 47 f_6 X$$

$$UR_1 X = 302 R_1 X + 20 305 f_6 X + 20 214 P_X + 20 213 R_3 X + 20 305 f_6 X + 20 214 P_X + 20 213 f_6 X + 304 f_6 X + 212 P_X \\ + 213 f_6 X + 47 f_6 X + 20 306 f_6 X + 20 47 P_X + 20 306 f_6 X + 20 47 f_6 X$$

$$UR_2 X = 303 R_2 X - 5 305 f_6 X - 5 214 P_X - 5 213 R_3 X + 5 305 f_6 X + 5 214 P_X + 5 213 f_6 X - 5 306 f_6 X \\ - 5 47 P_X + 5 306 f_6 X + 5 47 f_6 X$$

$$UR_3 X = 304 R_3 X + 212 P_X + 213 f_6 X + 47 f_6 X - 304 f_6 X - 212 P_X - 213 f_6 X - 47 f_6 X$$

$$UP_1^2 = 301 + 18 214 + 18^2 305 + 18 214 + 24^2 305 - 6 24 213 + 6^2 304 - 6 24 213 - 6 47 52 + 49^2 306 \\ + 52^2 306 - 6 47 52 \\ = 301 + 2 18 214 + 305 (18^2 + 24^2) - 2 6 24 213 + 6^2 304 - 2 6 47 52 + 306 (49^2 + 52^2)$$

$$UP_R = 19 214 + 18 19 305 + 23 24 305 + 24 214 - 6 24 213 + 6^2 304 - 6 212 - 6 27 213 - 6 47 52 + 49 50 306 \\ + 52 53 306 - 6 47 52 \\ = 214 (19 + 24) + 305 (18 19 + 23 24) - 6 213 (23 + 24) + 6^2 304 - 6 212 - 6 47 (52 + 53) + 306 (49 50 + 52 53)$$



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TECHNICAL DEPARTMENT (Aircraft)

REPORT NO. 7-0558-8

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$$\begin{aligned} \underline{UP_1 Q_1} &= 20 \underline{214} + 18 \underline{20 \ 305} + 20 \underline{24 \ 305} + 24 \underline{213} - 6 \underline{304} - 6 \underline{20 \ 213} - 6 \underline{47 \ 51} + 49 \underline{51 \ 306} + 51 \underline{52 \ 306} + 47 \underline{52} \\ &= 20 \underline{214} + 20 \underline{305} (18+24) + 213 (24-6 \ 20) - 6 \underline{304} - 47 (6 \ 51 - 52) + 306 \underline{51} (49 + \underline{52}). \end{aligned}$$

$$\begin{aligned} \underline{UP_1 Q_2} &= 6 \underline{13 \ 47} - 13 \underline{49 \ 306} - 13 \underline{52 \ 306} \\ &= 6 \underline{13 \ 47} - 13 \underline{306} (49 + \underline{52}) \end{aligned}$$

$$\begin{aligned} \underline{UP_1 R_1} &= 20 \underline{214} + 18 \underline{20 \ 305} + 20 \underline{24 \ 305} + 24 \underline{213} - 6 \underline{304} - 6 \underline{20 \ 213} - 6 \underline{20 \ 47} + 20 \underline{49 \ 306} + 20 \underline{52 \ 306} + 47 \underline{52} \\ &= 20 \underline{214} + 20 \underline{305} (18+24) + 213 (24-6 \ 20) - 6 \underline{304} - 47 (6 \ 20 - 52) + 20 \underline{306} (49 + \underline{52}). \end{aligned}$$

$$\begin{aligned} \underline{UP_1 R_2} &= -5 \underline{214} - 5 \underline{18 \ 305} + 5 \underline{24 \ 305} - 5 \underline{6 \ 213} - 5 \underline{6 \ 47} - 5 \underline{49 \ 306} + 5 \underline{52 \ 306} \\ &= -5 \underline{214} - 5 \underline{305} (18-24) - 5 \underline{6 \ 213} - 5 \underline{6 \ 47} - 5 \underline{306} (49 - \underline{52}). \end{aligned}$$

$$\begin{aligned} \underline{UP_1 R_3} &= 212 + 18 \underline{213} - 24 \underline{213} + 6 \underline{304} + 47 \underline{49} - 47 \underline{52} \\ &= 212 + 213 (18-24) + 6 \underline{304} + 47 (49 - \underline{52}). \end{aligned}$$

$$\begin{aligned} \underline{UP_2^2} &= 301 + 23 \underline{214} - 6 \underline{212} + 19^2 \underline{305} + 23^2 \underline{305} + 23 \underline{214} - 6 \underline{23 \ 213} + 6^2 \underline{304} - 6 \underline{212} - 6 \underline{23 \ 213} \\ &\quad - 6 \underline{47 \ 53} + 50^2 \underline{306} + 53^2 \underline{306} - 6 \underline{47 \ 53} \\ &= 301 + 2 \underline{23 \ 214} - 2 \underline{6 \ 212} + 305 (19^2 + 23^2) - 2 \underline{6 \ 23 \ 213} + 6^2 \underline{304} - 2 \underline{6 \ 47 \ 53} + 306 (50^2 + 53^2). \end{aligned}$$

$$\begin{aligned} \underline{UP_2 Q_1} &= 20 \underline{214} + 212 + 19 \underline{20 \ 305} + 20 \underline{23 \ 305} + 23 \underline{213} - 6 \underline{304} - 6 \underline{20 \ 213} - 6 \underline{47 \ 51} + 50 \underline{51 \ 306} \\ &\quad + 51 \underline{52 \ 306} + 47 \underline{53} \\ &= 20 \underline{214} + 212 + 20 \underline{305} (19+23) + 213 (23-6 \ 20) - 6 \underline{304} - 47 (6 \ 51 - 53) + 51 \underline{306} (50 + \underline{53}). \end{aligned}$$

$$\begin{aligned} \underline{UP_2 Q_2} &= 6 \underline{13 \ 47} - 13 \underline{50 \ 306} - 13 \underline{53 \ 306} \\ &= 6 \underline{13 \ 47} - 13 \underline{306} (50 + \underline{53}). \end{aligned}$$

$$\begin{aligned} \underline{UP_2 R_1} &= 20 \underline{214} + 212 + 19 \underline{20 \ 305} + 20 \underline{23 \ 305} + 23 \underline{213} - 6 \underline{304} - 6 \underline{20 \ 213} - 6 \underline{20 \ 47} + 20 \underline{50 \ 306} \\ &\quad + 20 \underline{53 \ 306} + 47 \underline{53} \end{aligned}$$

$$\begin{aligned} \underline{UP_2 R_2} &= -5 \underline{214} - 5 \underline{19 \ 305} + 5 \underline{23 \ 305} - 5 \underline{6 \ 213} - 5 \underline{6 \ 47} - 5 \underline{50 \ 306} + 5 \underline{53 \ 306} \\ &= -5 \underline{214} - 5 \underline{305} (19-23) - 5 \underline{6 \ 213} - 5 \underline{6 \ 47} - 5 \underline{306} (50 - \underline{53}). \end{aligned}$$

$$\begin{aligned} \underline{UP_2 R_3} &= -212 + 19 \underline{213} + 23 \underline{213} + 6 \underline{304} + 47 \underline{50} - 47 \underline{53} \\ &= -212 + 213 (19+23) + 6 \underline{304} + 47 (50 - \underline{53}). \end{aligned}$$



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$$\begin{aligned} UQ_1^2 &= 20^2 305 + 20^2 305 + 20 213 + 304 + 20 213 + 47 51 + 51^2 306 + 51^2 306 + 47 51 \\ &= 2 20^2 305 + 2 20 213 + 304 + 2 47 51 + 2 51^2 306. \end{aligned}$$

$$UQ_1 R_2 = -13 47 - 13 51 306 - 13 51 306 = -13 47 - 2 13 51 306.$$

$$UQ_1 R_3 = 2 20^2 305 + 2 20 213 + 304 + 47 (20 + 51) + 2 20 51 306.$$

$$UQ_1 R_2 = .5 213 + .5 47.$$

$$UQ_1 R_3 = -304$$

$$UQ_2^2 = 2 13^2 306.$$

$$UQ_2 R_1 = -2 13 20 306 - 13 47.$$

$$UQ_2 R_2 = 0.$$

$$UQ_2 R_3 = 0.$$

$$UR_1^2 = 302 + 2 20^2 305 + 2 20 213 + 304 + 2 20 47 + 2 20^2 306.$$

$$UR_1 R_2 = .5 213 + .5 47.$$

$$UR_1 R_3 = -304$$

$$UR_2^2 = 303 + .5 305 + .5 306.$$

$$UR_2 R_3 = -213 - 47$$

$$UR_3^2 = 21 304.$$



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Sample Calculation

Using the geometry as shown on sheet no. of the check analysis,

$$e = 8.00000 ; \quad d = 3.20000 ; \quad a = 27.00000 ; \quad L = 54.5894$$

$$a_2 = .707107 \times 27 = 19.09189$$

$$a_{16} = 54.00000$$

$$a_3 = .292893 \times 27 = 7.90811$$

$$a_{17} = 3.20000$$

$$a_4 = 7.90811$$

$$a_{20} = 3.20000$$

$$a_5 = 19.09189$$

$$a_{25} = 60.40000$$

$$a_6 = 0$$

$$a_{26} = 46.09189$$

$$a_7 = 27.00000$$

$$a_{27} = 0$$

$$a_{11} = 3.20000$$

$$a_{29} = 7.90811$$

$$a_{13} = 27.00000$$

$$a_{14} = 0$$

$$a_{15} = 0$$

For another geometry these numbers would be measurements. Although a measurement might be 16.85", it must be carried as 16.85000" in order to insure accuracy when we obtain numbers consisting of few digits by subtraction of two numbers consisting of many digits.

$$\theta_1 = 180 - 2 \sin^{-1} \frac{a_2}{\sqrt{a_2^2 + a_3^2}} = 45^\circ$$

$$r_1 = \frac{a_2}{\sin \theta_1} = 27.00000$$

$$\theta_3 = 180 - 2 \sin^{-1} \frac{a_5}{\sqrt{a_5^2 + a_4^2}} = 45^\circ$$

$$r_2 = \frac{a_5}{\sin \theta_3} = 27.00000$$

$$\theta_2 = \theta_3 + \sin^{-1} \frac{a_6}{r_2}$$

$$\beta = \tan^{-1} \frac{r_2 + a_2 - a_4}{r_1 + a_5 - a_3}$$

$$A_{ST} = .20000 \text{ in}^2 ; \quad E_{ST} = 2.90000 \times 10^3 \text{ p.s.i.}$$

$$A_{SP} = 4.47100 ; \quad E_{SP} = 1.00000 \times 10^3 ; \quad G_{SP} = .40000 \times 10^3 ; \quad I_{SP} = 33.72000 \text{ in}^4 ; \quad \alpha_{SP} = 4.14660$$

$$A_{FO} = 1.21040 ; \quad E_{FO} = 1.00000 \times 10^3 ; \quad G_{FO} = .40000 \times 10^3 ; \quad I_{FO} = 1.73320 ; \quad \alpha_{FO} = 5.93100$$



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$$3 = a_{25} - a_{20} = 60.4 - 3.2 = 57.20000$$

$$4 = \frac{1}{2 \cos \delta} = \frac{54.5894}{108} = .505457$$

$$5 = \frac{1}{2 \sin \delta} = \frac{54.5894}{14} = 3.411838$$

$$6 = \cos \beta = .707107$$

$$7 = \sin \beta = .707107$$

$$8 = a_{20} + a_{17} + a_2 + a_{13} = 3.2 + 3.2 + 19.09189 + 27 = 52.49189$$

$$9 = a_3 - a_{14} + a_{15} = 7.90811 - 0 + 0 = 7.90811$$

$$10 = 11 - 7 = .395456 - .707107 = -.311651$$

$$11 = \frac{6.8 + 7.9}{2 a_{16}} = \frac{.707107(52.49189 + 7.90811)}{108} = .395456$$

$$12 = \frac{a_{25}}{2 a_{16}} = \frac{60.4}{108} = .559259$$

$$13 = \frac{1}{2 a_{16}} = \frac{1}{108} = .00925925$$

$$14 = a_2 + a_{13} = 19.09189 + 27 = 46.09189$$

$$15 = 2 a_{16} - 9 = 108 - 7.90811 = 100.09189$$

$$16 = 9 = 7.90811$$

$$17 = a_{25} - a_{20} - a_{17} = 60.4 - 3.2 - 3.2 = 54.00000$$

$$18 = \frac{7.15 - 6.14}{2 a_{16}} = \frac{.707107(100.09189 - 46.09189)}{108} = .353553$$

$$19 = -\frac{6.14 + 7.16}{2 a_{16}} = -\frac{.707107(46.09189 + 7.90811)}{108} = -.353553$$

$$20 = \frac{17}{2 a_{16}} = \frac{54}{108} = .50000$$

$$21 = 11 = .395456$$

$$22 = 10 = -.311651$$

$$23 = 19 + 7 = -.353553 + .707107 = .353553$$

$$24 = 18 - 7 = .353553 - .707107 = -.353553$$

$$25 = \frac{L}{2 A_{ST} E_{ST}} = \frac{54.5894}{2 \times 2 \times 2.9 \times 10^7} = 47.05982 \times 10^{-7}$$

$$10^{-7} = g$$

$$26 = 2 \cdot 25^2 = 2 \times 47.05982 g \cdot 3.411838^2 = 1095.613 g$$

$$27 = 2 \cdot 25^2 = 2 \times 47.05982 g \cdot .505457^2 = 24.04632 g$$



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$$28 = \tan \alpha = 0$$

$$29 = \sin \alpha = 0$$

$$30 = \cos \alpha = 1.00000$$

$$31 = \frac{a_{32}}{30} + a_{30} = 0 + 3.2 = 3.20000$$

$$32 = \frac{1}{2 \cdot 30 E_{sp} I_{sp}} = \frac{1}{2 \times 1 \times 10^7 \times 33.72} = .014828 \text{ g}$$

$$33 = \frac{a_{16}}{2 \cdot 30 A_{sp} E_{sp}} = \frac{54}{2 \times 1 \times 4471 \times 10^7} = 6.03890 \text{ g}$$

$$34 = \frac{a_{16} \alpha_{sp}}{2 \cdot 30 A_{sp} b_{sp}} = \frac{54 \times 4.146}{2 \times 1 \times 4471 \times 4 \times 10^7} = 62.5932 \text{ g}$$

$$35 = 32 \cdot 31^2 \cdot a_{16} = .014828 \text{ g} \cdot 3.2^2 \times 54 = 8.19929 \text{ g}$$

$$36 = 32 \cdot 31 \cdot a_{16}^2 = .014828 \text{ g} \cdot 3.2 \times 54^2 = 138.363 \text{ g}$$

$$37 = 32 \cdot 31 \cdot a_{16}^2 \cdot 28 = 0$$

$$38 = \frac{32 \cdot a_{16}}{3} = .014828 \times 54^2 \times 18 = 772.292 \text{ g}$$

$$39 = 38 \cdot 28^2 = 0$$

$$40 = 33 \cdot 30^2 = 6.03890 \text{ g}$$

$$299 = 2 \cdot 38 \cdot 28 = 0$$

$$41 = 2 \cdot 33 \cdot 30 \cdot 29 = 0$$

$$42 = 33 \cdot 29^2 = 0$$

$$43 = 34 \cdot 29^2 = 0$$

$$44 = 2 \cdot 34 \cdot 29 \cdot 30 = 0$$

$$45 = 34 \cdot 30^2 = 62.5932 \text{ g}$$

$$46 = 35 + 37 + 39 + 40 + 43 = \text{g} (8.19929 + 0 + 0 + 6.03890 + 0) = 14.2382 \text{ g}$$

$$47 = 36 + 299 - 41 + 44 = \text{g} (138.363 + 0 - 0 + 0) = 138.363 \text{ g}$$

$$48 = 38 + 42 + 45 = \text{g} (772.292 + 0 + 62.5932) = 840.885 \text{ g}$$

$$49 = 10 + 18 = -.311651 + .353553 = .041902 \text{) } .0419025$$

$$50 = 11 + 19 = .395456 - .353553 = .041903$$

$$51 = 20 - 12 = .500000 - .559259 = -.059259$$

$$52 = 21 + 24 = .395456 - .353553 = .041903 \text{) } .0419025$$

$$53 = 22 + 23 = -.311651 + .353553 = .041902$$

$$55 = \tan \Omega = \frac{a_{14} - a_{15}}{a_{13}} = 0 ; \quad \Omega = 0$$

$$56 = \sin \Omega = 0$$

$$57 = \cos \Omega = 1.00000$$



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$$82 = \frac{a_3^3}{6 E_{F0} I_{F0} 57} = \frac{27^3}{6 \times 10^7 \times 1.7382 \times 1} = 1287.3 \text{ g}$$

$$83 = \frac{a_2}{2 A_{F0} E_{F0} 87} = \frac{27}{2 \times 1.2104 \times 10^4 \times 10^3} = 11.1533 \text{ g}$$

$$84 = \frac{a_3 \times F_0}{2 A_{F0} G_{F0} 57} = \frac{27 \times 5.931}{2 \times 1.2104 \times 4 \times 10^3} = 165.376 \text{ g}$$

$$85 = 82 + 83 56^2 + 84 57^2 = \text{g} (1287.3 + 0 + 165.376) = 2052.68 \text{ g}$$

$$88 = 2 55 82 + 2 56 87 (84 - 83) = 0$$

$$59 = 82 55^2 + 83 57^2 + 84 56^2 = \text{g} (0 + 11.1533 \times 1 + 0) = 11.1533 \text{ g}$$

$$86 = r_1 + a_{15} - a_{14} = 27 + 0 - 0 = 27.00000$$

$$87 = \frac{r_1}{2 E_{F0} I_{F0}} = \frac{27}{2 \times 10^7 \times 1.7382} = 7.76665 \text{ g}$$

$$88 = \frac{r_1}{2 A_{F0} E_{F0}} = 11.1533 \text{ g}$$

$$89 = \frac{r_1 \times F_0}{2 A_{F0} G_{F0}} = 165.376 \text{ g}$$

$$90 = .5 (8_1 - .5 \sin 28_1) = .5 (.7854 - .5) = .14270$$

$$91 = .5 (8_1 + .5 \sin 28_1) = .64270$$

$$92 = 1 - \cos 8_1 = 1 - .707107 = .292893$$

$$93 = \sin 8_1 = .707107$$

$$94 = .5 (1 - \cos 28_1) = .5 (1 - 0) = .50000$$

$$95 = 87 8_1 a_{13}^2 = 7.76665 \text{ g} .7854 \times 27^2 = 4446.85 \text{ g}$$

$$96 = 2 87 8_1 a_{13} 86 = 2 \times 4446.85 = 8893.70 \text{ g}$$

$$97 = 87 8_1 86^2 = 4446.85 \text{ g}$$

$$98 = 87 90 r_1^2 = 7.76665 \text{ g} .1427 \times 27^2 = 807.949 \text{ g}$$

$$99 = 87 91 r_1^2 = 7.76665 \text{ g} .6427 \times 27^2 = 3638.893 \text{ g}$$

$$100 = 2 87 92 r_1 a_{13} = 2 \times 7.76665 \text{ g} .292893 \times 27^2 = 3316.655 \text{ g}$$

$$101 = 2 87 92 r_1 86 = 3316.655 \text{ g}$$

$$102 = 2 87 93 r_1 a_{13} = 2 \times 7.76665 \text{ g} .707107 \times 27^2 = 8007.121 \text{ g}$$

$$103 = 2 87 93 r_1 86 = 8007.121 \text{ g}$$

$$104 = 87 94 r_1^2 = 7.76665 \text{ g} .5 \times 27^2 = 2830.944 \text{ g}$$

$$105 = 88 90 = 11.1533 \text{ g} .1427 = 1.59158 \text{ g}$$

$$106 = 88 91 = 11.1533 \text{ g} .6427 = 7.16823 \text{ g}$$

$$107 = 88 94 = 11.1533 \text{ g} .50000 = 5.57665 \text{ g}$$



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$$108 = 89 \ 90 = 165.376 \ g \cdot 1427 = 23.5892 \ g$$

$$109 = 89 \ 91 = 165.376 \ g \cdot 6427 = 106.2872 \ g$$

$$110 = 89 \ 94 = 165.376 \ g \cdot 5000 = 82.6880 \ g$$

$$111 = 95 + 98 + 100 + 105 + 109 = g(4446.85 + 807.95 + 3316.66 + 1.59 + 106.29) = 8679.34 \ g$$

$$112 = -96 - 101 + 102 + 104 + 107 - 110 = g(-8893.70 - 3316.66 + 8007.12 + 2820.94 + 6.58 - 82.69) = -14449.41 \ g$$

$$113 = 97 + 99 - 103 + 106 + 108 = g(4446.85 + 3638.89 - 8007.12 + 7.17 + 23.60) = 109.39 \ g$$

$$114 = \cos \beta_3 = .707107$$

$$115 = \sin \beta_3 = .707107$$

$$116 = \frac{r_2}{2 E_{F0} I_{F02}} = 7.76665 \ g$$

$$117 = \frac{r_2}{2 A_{F02} E_{F0}} = 11.1533 \ g$$

$$118 = \frac{r_2 \cos \alpha_0}{2 A_{F02} G_{F0}} = 165.376 \ g$$

$$119 = .5(\beta_2 - .5 \sin 2\beta_2) = .1427$$

$$120 = .5(\beta_2 + .5 \sin 2\beta_2) = .6427$$

$$121 = 1 - \cos \beta_2 = .292893$$

$$122 = \sin \beta_2 = .707107$$

$$123 = .5(1 - \cos 2\beta_2) = .50000$$

$$124 = 116 \ \beta_2 = 7.76665 \ g \cdot 7854 = 6.09993 \ g$$

$$125 = 2 \ 116 \ \beta_2^2 = 2 \times 6.09993 \times 27 = 329.396 \ g$$

$$126 = 116 \ \beta_2^3 = 4446.85 \ g$$

$$127 = 116 \ 119 \ \beta_2^4 = 807.949 \ g$$

$$128 = 116 \ 120 \ \beta_2^5 = 3638.893 \ g$$

$$129 = 2 \ 116 \ 121 \ \beta_2^6 = 122.839 \ g$$

$$130 = 2 \ 116 \ 121 \ \beta_2^7 = 3316.655 \ g$$

$$131 = 2 \ 116 \ 122 \ \beta_2^8 = 296.560 \ g$$

$$132 = 2 \ 116 \ 122 \ \beta_2^9 = 8007.121 \ g$$

$$133 = 116 \ 123 \ \beta_2^{10} = 2820.944 \ g$$

$$134 = 117 \ 119 = 1.5958 \ g$$

$$135 = 117 \ 120 = 7.16823 \ g$$

$$136 = 117 \ 123 = 5.57665 \ g$$

$$137 = 118 \ 119 = 23.5792 \ g$$

$$138 = 118 \ 120 = 106.2872 \ g$$

$$139 = 118 \ 123 = 82.6880 \ g$$



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$$140 = 126 + 128 - 132 + 135 + 137 = g(4446.85 + 2638.89 - 8007.12 + 7.17 + 23.60) = 109.39 g$$

$$141 = 127 + 134 + 138 = g(807.95 + 1.59 + 106.28) = 915.83 g$$

$$142 = -125 + 131 = g(-329.40 + 296.56) = -32.84 g$$

$$143 = 130 - 123 - 136 + 129 = g(3316.66 - 2820.94 - 5.58 + 82.69) = 562.83 g$$

$$144 = 124 14^2 = 6.09993 g 46.09189^2 = 12,959.06 g$$

$$145 = 124 g^2 = 6.09993 g 7.90811^2 = 381.48 g$$

$$146 = 2 124 g 14 = 2 \times 6.09993 g 7.90811 \times 46.09189 = 4446.85 g$$

$$147 = 140 115^2 = 109.39 g 5000 = 54,695 g$$

$$148 = 141 114^2 = 915.83 g 5000 = 457,915 g$$

$$149 = 143 114 115 = 562.83 g 5000 = 281,415 g$$

$$150 = 147 + 148 + 149 = 794.025 g$$

$$151 = 140 114^2 = 54,695 g$$

$$152 = 141 115^2 = 457,915 g$$

$$153 = 143 114 115 = 281,415 g$$

$$154 = 151 + 152 - 163 = 231,195 g$$

$$155 = 2 140 114 115 = 109.39 g$$

$$156 = 2 141 114 115 = 915.83 g$$

$$157 = 143 (114^2 - 115^2) = 0$$

$$158 = 155 - 156 + 157 = -806.44 g$$

$$159 = 142 14 115 = -32.84 g 46.09189 \times 7.0707 = -1070.318 g$$

$$160 = 129 14 114 = 122.839 g 46.09189 \times 7.07107 = 4083.56 g$$

$$161 = 159 - 160 = -5073.878 g$$

$$162 = 142 14 114 = -1070.318 g$$

$$163 = 129 14 115 = 4083.56 g$$

$$164 = 162 + 163 = 2933.242 g$$

$$165 = 142 g 115 = -32.84 g 7.90811 \times 7.07107 = -183.637 g$$

$$166 = 129 g 114 = 122.839 g 7.90811 \times 7.07107 = 686.901 g$$

$$167 = -165 + 166 = 870.538 g$$

$$168 = 142 g 114 = -183.637 g$$

$$169 = 129 g 115 = 686.901 g$$

$$170 = -168 - 169 = -503.264 g$$

$$171 = 144 + 154 + 164 = g(12,959.06 + 231,195 + 2933.242) = 16,123.497 g$$

$$172 = 145 + 150 + 167 = g(381.48 + 794.025 + 870.538) = 2046.043 g$$

$$173 = 7^2 150 + 6^2 154 - 6^2 158 = -5 g(794.025 + 231,195 + 806.44) = 915.83 g$$

$$174 = -144 - 158 + 161 + 170 = g(-4446.84 - 806.44 - 5073.878 - 503.264) = -10,830.422 g$$

$$175 = 2 \times 154 - 2(158 + 161) + 6 164 = 707107 g(462.39 + 806.44 + 5073.878 + 2930.242) = 6559,089 g$$

$$176 = -7(2 150 + 167) + 6(158 + 170) = 707107(-1838.05 - 870.538 - 806.44 - 503.264) = -2664.586 g$$



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$$177 = a_{16} - a_7 = 54 - 27 = 27$$

$$178 = 14 + a_{29} = 46.09189 + 7.90811 = 54$$

$$179 = a_5 + a_6 = 19.09189 + 0 = 19.09189$$

$$180 = 6a_{24} + 7179 = .707107 (7.90811 + 19.09189) = 19.09189$$

$$181 = \frac{1}{2E_{FO} I_{FO_3}} = .287654 \text{ g}$$

$$182 = \frac{a_2}{2A_{FO_2} E_{FO}} = 11.1533 \text{ g}$$

$$183 = \frac{a_2 \times F_{O_2}}{2A_{FO_2} G_{FO}} = 165.376 \text{ g}$$

$$184 = 181 a_7 = 7.76665 \text{ g}$$

$$185 = 184 a_7 = 209.700$$

$$186 = 185 \frac{a_7}{2} + 183 = 1887.300 + 165.376 = 2052.676 \text{ g}$$

$$187 = 184 178^2 = 7.76665 \text{ g } 54^2 = 22,647.551 \text{ g}$$

$$188 = 184 180^2 = 2830.94 \text{ g}$$

$$189 = 184 177^2 = 5661.888 \text{ g}$$

$$190 = 2 184 178 180 = 2 \times 7.76665 \text{ g } 54 \times 19.09189 = 16,014.24 \text{ g}$$

$$191 = 2 184 178 177 = 2 \times 7.76665 \text{ g } 54 \times 27 = 22,647.55 \text{ g}$$

$$192 = 2 184 180 177 = 2 \times 7.76665 \text{ g } 19.09189 \times 27 = 8007.12 \text{ g}$$

$$193 = 185 178 = 209.7 \text{ g } 54 = 11,323.800 \text{ g}$$

$$194 = 185 180 = 209.7 \text{ g } 19.09189 = 4003.56 \text{ g}$$

$$195 = 185 177 = 209.7 \text{ g } 27 = 5661.9 \text{ g}$$

$$196 = 185 7 178 = 209.7 \text{ g } .707107 \times 54 = 8007.12 \text{ g}$$

$$197 = 185 7 180 = 209.7 \text{ g } .707107 \times 19.09189 = 2830.95 \text{ g}$$

$$198 = 185 7 177 = 209.7 \text{ g } .707107 \times 27 = 4003.56 \text{ g}$$

$$199 = 2 186 7 = 1.414214 \times 2052.676 = 2902.923 \text{ g}$$

$$200 = 186 7^2 = 1026.338 \text{ g}$$

$$201 = 2 182 6 = 15.7732 \text{ g}$$

$$202 = 182 6^2 = 5.5767 \text{ g}$$

$$203 = 187 + 182 = \text{g} (22,647.55 + 11.15) = 22,658.70 \text{ g}$$

$$204 = 188 + 197 + 200 + 202 = \text{g} (2830.94 + 2830.94 + 1026.34 + 5.58) = 6693.80 \text{ g}$$

$$205 = 189 + 195 + 186 = \text{g} (5661.89 + 5661.89 + 2052.68) = 13,376.46 \text{ g}$$

$$206 = 190 + 196 + 201 = \text{g} (16,014.24 + 8007.12 + 15.77) = 24,037.13 \text{ g}$$

$$207 = -191 - 193 = \text{g} (-22,647.55 - 11,323.80) = -33,971.35 \text{ g}$$

$$208 = -192 - 194 - 198 - 199 = \text{g} (-8007.12 - 4003.56 - 4003.56 - 2902.92) = -18,917.16 \text{ g}$$

$$209 = 85 + 111 + 171 + 203 = \text{g} (2052.68 + 8679.34 + 16,123.497 + 22,658.70) = 49,514.217 \text{ g}$$



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$$\begin{aligned}
 210 &= 173 + 204 = q(915.83 + 6673.80) = 7609.63 q \\
 211 &= 59 + 113 + 172 + 205 = q(11.15 + 107.39 + 2046.043 + 13,376.46) = 15,543.043 q \\
 212 &= 175 + 206 = q(6559.09 + 24,037.13) = 30,596.22 q \\
 213 &= 58 + 112 + 178 + 207 = q(-1449.41 - 10,830.422 - 33,971.35) = -46,251.182 q \\
 214 &= 176 + 208 = -2664.59 q - 18,917.16 q = -21,581.75 q \\
 300 &= 209 + 46 = 49,528.455 q \\
 301 &= 2210 = 15,219.26 q \\
 302 &= 226 = 2,191.226 q \\
 303 &= 227 = 48,09264 q \\
 304 &= 2300 = 99,056.91 q \\
 305 &= 2211 = 31,086.086 q \\
 306 &= 248 = 1681.770 q
 \end{aligned}$$

$$\begin{aligned}
 UP_1^2 &= +301 = 15,219.260 \\
 &+ 215 \ 214 = 2 \times .353553(-21,581.75) = -15,260.606 \\
 &+ 305(18^2 + 24^2) = 31,086.086(.125 + .125) = 7,771.522 \\
 &- 26 \ 24 \ 213 = -2 \times .707107(-353553)(-46,251.182) = -23,125.591 \\
 &+ 6^2 \ 304 = .5 \times 99,056.91 = 49,528.455 \\
 &- 6 \ 47 \ 52 = -1.414214 \times 138.363 \times .0419025 = -8.199 \\
 &+ 306(49^2 + 52^2) = 1681.770 \times 2(.0419025)^2 = 5.906 \\
 &= 34,130.747 q
 \end{aligned}$$

$$\begin{aligned}
 UP_2 &= +214(19 + 24) = -21,581.75(-2)(.353553) = 15,260.606 \\
 &+ 305(1819 + 2324) = 31,086.086(-.125 - .125) = -7,771.522 \\
 &- 6 \ 213(23 + 24) = -7.07107(-46,251.182) = 0.0 \\
 &+ 6^2 \ 304 = 49,528.455 \\
 &- 6 \ 212 = -7.07107 \times 30,596.22 = -21,434.801 \\
 &- 6 \ 47(52 + 53) = -26 \ 47 \ 52 = -8.199 \\
 &+ 306(49 \ 50 + 52 \ 53) = 306(49^2 + 52^2) = 5.906 \\
 &= 35,380.445 q
 \end{aligned}$$

$$\begin{aligned}
 UP_3 &= +20 \ 214 = .5(-21,581.75) = -10,790.875 \\
 &+ 20 \ 205(18 + 24) = .5(31,086.086)(0) = 0.0 \\
 &+ 213(24 - 6 \ 20) = (-46,251.182)(-.707107) = 32,704.535 \\
 &- 6 \ 304 = -.707107 \times 99,056.91 = -70,043.834 \\
 &- 47(6 \ 51 - 52) = -138.363[.707107(-.059259) - .0419025] = 11.596 \\
 &+ 306 \ 51(49 + 52) = 1681.770(-.059259)(2)(.0419025) = -8.352 \\
 &= -48,136.930
 \end{aligned}$$



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Note: We wish to compare
 $\frac{\partial U_{Avro}}{\partial R_i}$ with $\frac{\partial U}{\partial R_i}$.

Considering only the strain energy of the former (Infinitely stiff spar and struts)

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Referring to page 12, $46 = 0 \quad \therefore 300 = 209 = 49,514.217 \text{ g}$

Also $0 = 26, 27, 48, 47, 302, 303, 306 \quad 304 = 99,028.834 \text{ g}$

$$\underline{U P_1^2} = 34,130.747 \text{ g} - 5.906 \text{ g} + 8.199 \text{ g} - 49,528.455 \text{ g} + 49,514.217 \text{ g} = 34,118.802 \text{ g}$$

$$\underline{U P_1 R_2} = 35,380.445 - 5.906 + 8.199 - 49,528.455 + 49,514.217 = 35,368.500 \text{ g}$$

$$\underline{U P_1 R_3} = -48,126.930 + 8.352 - 11.596 + 70,043.834 - 70,023.699 = -48,110.05 \text{ g}$$

$$\underline{U P_1 R_4} = 0$$

$$\underline{U P_2 R_1} = -48,102.825 - 70.47 + 43.12 + 70,043.834 - 70,023.699 = -48,110.05 \text{ g}$$

$$\underline{U P_2 R_2} = 16,103.62 + 48.919 = 16,152.54 \text{ g}$$

$$\underline{U P_2 R_3} = 67,935.519 - 70,043.834 + 70,023.699 = 67,915.39 \text{ g}$$

$$\underline{U P_2 R_4} = 37,112.326 - 5.906 + 8.199 - 49,528.455 + 49,514.217 = 37,100.38 \text{ g}$$

$$\underline{U R_2 R_1} = -50,235.245 + 8.352 - 11.596 + 70,043.834 - 70,023.699 = -50,218.37 \text{ g}$$

$$\underline{U R_2 R_2} = 0$$

$$\underline{U R_2 R_3} = -50,211.14 - 70.47 + 43.12 + 70,043.83 - 70,023.70 = -50,218.26 \text{ g}$$

$$\underline{U R_2 R_4} = 16,503.029 + 48.919 = 16,551.95 \text{ g}$$

$$\underline{U R_3 R_1} = 72,152.15 - 70,043.83 + 70,023.70 = 72,132.02 \text{ g}$$

$$\underline{U Q_1^2} = 68,244.18 - 11.81 + 16.40 - 99,056.91 + 99,028.43 = 68,320.29 \text{ g}$$

$$\underline{U Q_1 R_2} = 0$$

$$\underline{U Q_1 R_3} = 68,310.09 + 99.66 - 60.98 - 99,056.91 + 99,028.43 = 68,320.29 \text{ g}$$

$$\underline{U Q_1 R_4} = -23,056.411 - 69.18 = -23,125.59 \text{ g}$$



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$$\begin{aligned} \underline{UPQ_2} &= +6 \underline{13} \underline{42} &= .707107(.00925925)(138.363) &= .9059014 \\ &- \underline{13} \underline{306} (\underline{49} + \underline{52}) &= -.00925925(1621.770)(083805) &= \underline{-1.3050052} \\ & & &= \underline{-.399104 g} \end{aligned}$$

$$\begin{aligned} \underline{UPR_1} &= \underline{20} \underline{214} &= &= -10,790.875 \\ &+ \underline{20} \underline{305} (\underline{18} + \underline{24}) &= &= 0.0 \\ &+ \underline{213} (\underline{24} - \underline{6} \underline{20}) &= &= 32,704.535 \\ &- \underline{6} \underline{304} &= &= -70,043.834 \\ &- \underline{47} (\underline{6} \underline{20} - \underline{52}) &= -138.263(.353553 - .0419025) &= -43.121 \\ &+ \underline{20} \underline{206} (\underline{49} + \underline{52}) &= .5(1681.770)(.083805) &= \underline{70.470} \\ & & &= \underline{-48,102.825 g} \end{aligned}$$

$$\begin{aligned} \underline{UPR_2} &= -\underline{5} \underline{214} &= &= 10,790.875 \\ &- \underline{5} \underline{305} (\underline{18} - \underline{24}) &= -5(3,086.086)(.707107) &= -10,990.579 \\ &- \underline{5} \underline{6} \underline{213} &= -353553(-44,251.182) &= 16,352.244 \\ &- \underline{5} \underline{6} \underline{42} &= -353553(138.363) &= -48,919 \\ &- \underline{5} \underline{306} (\underline{49} - \underline{52}) &= &= \underline{0.0} \\ & & &= \underline{16,103.621 g} \end{aligned}$$

$$\begin{aligned} \underline{UPR_3} &= +\underline{212} &= &= 30,596.22 \\ &+ \underline{213} (\underline{18} - \underline{24}) &= -46,251.182(.707107) &= -32,704.535 \\ &+ \underline{6} \underline{304} &= &= 70,043.834 \\ &+ \underline{47} (\underline{49} - \underline{52}) &= &= \underline{0.0} \\ & & &= \underline{67,935.519 g} \end{aligned}$$

$$\begin{aligned} \underline{UP_2^2} &= +\underline{301} &= &= 15,219.26 \\ &+ \underline{2} \underline{33} \underline{214} &= .707107 \underline{214} &= -15,260.606 \\ &- \underline{2} \underline{6} \underline{212} &= -1.414214 \underline{212} &= -43,269.603 \\ &+ \underline{305} (\underline{19}^2 + \underline{23}^2) &= &= 7,771.522 \\ &- \underline{2} \underline{6} \underline{23} \underline{213} &= -\underline{5} \underline{213} &= 23,125.591 \\ &+ \underline{6}^2 \underline{308} &= \underline{5} \underline{304} &= 49,528.455 \\ &- \underline{2} \underline{6} \underline{47} \underline{53} &= &= -8.199 \\ &+ \underline{304} (\underline{50}^2 + \underline{53}^2) &= &= \underline{5,906} \\ & & &= \underline{37,112.326 g} \end{aligned}$$



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$$U_{R_1 R_3} = -99,028.43 \text{ g}$$

$$0 = U_{R_2^2}; U_{R_2 R_1}; U_{R_2 R_2}; U_{R_2 R_3}$$

$$U_{R_1^2} = 71,519.25 - 840.89 - 138.26 - 99,056.91 + 99,028.43 - 2,191.23 = 68,720.29 \text{ g}$$

$$U_{R_1 R_2} = -23,125.59 \text{ g}$$

$$U_{R_1 R_3} = -99,028.43 \text{ g}$$

$$U_{R_2^2} = 16,432.02 - 840.89 = 15,591.13 \text{ g}$$

$$U_{R_2 R_3} = 46,251.18 \text{ g}$$

$$U_{R_3^2} = 198,056.87 \text{ g}$$

$$0 = U_1 + 0.006832029 R_1 - 0.002312559 R_2 - 0.009902843 R_3$$

$$0 = U_2 - 0.002312559 R_1 + 0.001559113 R_2 + 0.004625118 R_3$$

$$0 = U_3 - 0.009902843 R_1 + 0.004625118 R_2 + 0.019805687 R_3$$

$$0 = 144.8740 U_1 + R_1 - 335.0297 R_2 - 1.434664 R_3$$

$$0 = 432.4213 U_2 - R_1 + 1.6741937 R_2 + 2.000000 R_3$$

$$0 = -100.9811 U_3 + R_1 - 467.0495 R_2 - 2.000000 R_3$$

$$R_2 = -2071.442 U_2 = -432.4213 U_2 + 100.9811 U_3$$

$$R_2 = -2087.538 U_2 + 487.4918 U_3$$

$$.565336 R_3 + 1.339164 R_2 = -144.8740 U_1 - 432.4213 U_2$$

$$.565336 R_3 = -144.8740 U_1 - 432.4213 U_2 + 708.0177 U_2 - 165.3397 U_3$$

$$= -144.8740 U_1 + 275.5964 U_2 - 165.3397 U_3$$

$$R_3 = -256.2617 U_1 + 487.4912 U_2 - 292.4626 U_3$$

$$R_1 = 100.9811 U_3 - 974.9831 U_2 + 227.6828 U_3 - 512.5234 U_1 + 974.9824 U_2$$

$$= -584.9252 U_3$$

$$= -512.5234 U_1 - 256.2613 U_3$$



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$$\begin{array}{rcl}
 \underline{UP_2Q_1} & = & + 20 \ 214 \\
 & + & 212 \\
 & + & 20 \ 305 (19 + 23) \\
 & + & 213 (23 - 6 \ 20) \\
 & - & 6 \ 304 \\
 & - & 47 (6 \ 51 - 53) \\
 & + & 51 \ 306 (50 + 53) \\
 & = & -10,790.875 \\
 & = & 30,596.22 \\
 & = & 0.0 \\
 & = & 0.0 \\
 & = & -70,043.834 \\
 & = & 11.596 \\
 & = & \underline{-8.352} \\
 & & -50,235.245 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UP_2Q_2} & = & + 6 \ 13 \ 47 \\
 & - & 13 \ 306 (50 + 53) \\
 & = & .9059014 \\
 & = & \underline{-1.3050052} \\
 & & - .399104 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UP_2R_1} & = & 20 \ 214 \\
 & + & 212 \\
 & + & 20 \ 305 (19 + 23) \\
 & + & 213 (23 - 6 \ 20) \\
 & - & 6 \ 304 \\
 & - & 47 (6 \ 20 - 53) \\
 & + & 20 \ 306 (50 + 53) \\
 & = & -10,790.875 \\
 & = & 30,596.22 \\
 & = & 0.0 \\
 & = & 0.0 \\
 & = & -70,043.834 \\
 & = & -43.121 \\
 & = & \underline{70.470} \\
 & & -50,211.14 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UP_2R_2} & = & .5 \ 214 \\
 & - & 5 \ 305 (19 - 23) \\
 & - & 5 \ 6 \ 213 \\
 & - & 5 \ 6 \ 47 \\
 & - & 5 \ 306 (50 - 53) \\
 & = & -10,790.875 \\
 & = & .353553 \ 305 \\
 & = & 10,990.579 \\
 & = & 16,352.244 \\
 & = & -48.919 \\
 & = & \underline{0.0} \\
 & & 16,503.029 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UP_2R_3} & = & -212 \\
 & + & 213 (19 - 23) \\
 & + & 6 \ 304 \\
 & + & 47 (50 - 53) \\
 & = & -30,596.22 \\
 & = & -707107 \ 213 \\
 & = & 32,704.535 \\
 & = & 70,043.834 \\
 & = & \underline{0.0} \\
 & & 72,152.149 \ g
 \end{array}$$

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$$\begin{aligned} \frac{\partial U_{F0}}{\partial R_2} = \delta_{3V} &= u_2 - .002312559 R_1 + .001559113 R_2 + .004625118 R_3 \\ &= .001610362 P_1 + .001650303 P_2 - .002305641 Q_1 \\ &\quad - .001486448 P_1 - .001486448 P_2 + .001975085 Q_1 - .0000017222 Q_2 \\ &\quad - .000079616 P_1 + .000079624 P_2 \\ &\quad - .000044582 P_1 - .0000252971 P_2 + .000337472 Q_1 + .000017722 Q_2 \\ &= -.00000284 P_1 - .000009492 P_2 - .000060661 Q_1 \\ &\quad \text{(Cannot be correct).} \end{aligned}$$

$$\begin{aligned} \frac{\partial U_{F0}}{\partial R_1} = \delta_3 &= u_1 + .006832029 R_1 - .002312559 R_2 - .009902843 R_3 \\ &= -.004811005 P_1 - .00502836 P_2 + .006832029 Q_1 \\ &\quad + .004391437 P_1 + .004391437 P_2 - .005835024 Q_1 + .0000052257 Q_2 \\ &\quad + .000118091 P_1 - .000118102 P_2 \\ &\quad + .0000954535 P_1 + .000541676 P_2 - .000722561 Q_1 - .00000379453 Q_2 \\ &= -.00020602 P_1 - .00020686 P_2 + .00027444 Q_1 + .0000014412 Q_2 \end{aligned}$$

Note: We obtained δ_3 previously from $\frac{\partial U}{\partial Q_1} = \delta_3$, which was

$$\delta_3 = -.00020869 P_1 - .00020869 P_2 + .00027750 Q_1 + .0000014958 Q_2$$

The different results are attributed to the approximations we made in writing the spar energy. The deflection δ_{3V} is extremely small, and cannot be computed in the manner shown above.



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$$\begin{aligned}
 UQ_1^2 &= 2 \underline{20^2 305} &= .5 \underline{305} &= 15,543.043 \\
 &+ 2 \underline{20 213} &= \underline{213} &= -46,251.182 \\
 &+ \underline{304} &= &= 99,056.91 \\
 &+ 2 \underline{47 51} &= 2(138.363)(-.059259) &= -16.399 \\
 &+ 2 \underline{51^2 306} &= 2(.003511629)(1681.770) &= \underline{11.812} \\
 &&&= 68,344.184 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 UQ_1 R_2 &= -13 \underline{47} &= -.00925925 \times 138.363 &= -1.281138 \\
 &- 2 \underline{13 51 306} &= -2(.00925925)(-.059259)(1681.77) &= \underline{1.845554} \\
 &&&= .56441 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 UQ_1 R_1 &= 2 \underline{20^2 305} &= &= 15,543.043 \\
 &+ 2 \underline{20 213} &= &= -46,251.182 \\
 &+ \underline{304} &= &= 99,056.91 \\
 &+ \underline{47(20+51)} &= 138.363(.5000-.059259) &= \underline{60.98} \\
 &+ 2 \underline{20 51 306} &= -.059259(1681.77) &= \underline{-99.66} \\
 &&&= 68,310.091 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 UQ_1 R_3 &= .5 \underline{213} &= &= -23,125.591 \\
 &+ .5 \underline{47} &= .5 \times 138.363 &= \underline{69.18} \\
 &&&= -23,056.411 \text{ g}
 \end{aligned}$$

$$UQ_1 R_3 = -304 = -99,056.91 \text{ g}$$

$$UQ_2^2 = 2 \underline{13^2 306} = 2 \times .000857337 \times 1681.77 = .28837 \text{ g}$$

$$UQ_2 R_1 = -2 \underline{13 20 306} - \underline{13 47} = -.00925925 \times 1681.77 - 1.28114 = -16.85307 \text{ g}$$

$$UQ_2 R_2 = 0 ; UQ_2 R_3 = 0 ;$$

$$\begin{aligned}
 UR_1^2 &= \underline{302} &= &= 2,191.226 \\
 &+ 2 \underline{20^2 305} &= &= 15,543.043 \\
 &+ 2 \underline{20 213} &= &= -46,251.182 \\
 &+ \underline{304} &= &= 99,056.91 \\
 &+ 2 \underline{20 47} &= &= 138.363 \\
 &+ 2 \underline{20^2 306} &= .5 \times 1681.770 &= \underline{840.885} \\
 &&&= 71,519.245 \text{ g}
 \end{aligned}$$



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$$UR_2 R_2 = .5213 + .547 = -23,125.591_f + 69.18_g = -23,056.411_f$$

$$UR_1 R_3 = -304 = -99,056.91_f$$

$$UR_2^2 = 303 + .5305 + .5206 = 48.093 + 15,543.043 + 840.885 = 16,432.021_g$$

$$UR_2 R_3 = -213 - 47 = 46,251.182 - 138.363 = 46,112.819_f$$

$$UR_3^2 = 2304 = 198,113.820_g$$

Referring to the set of equations at the top of sheet No. , using the last 3 equations, let u_1 represent the first 4 terms in $UR_1 = 0$; u_2 in $UR_2 = 0$; u_3 in $UR_3 = 0$.

$$\left. \begin{aligned} 0 &= u_1 + .0071519245 R_1 - .002305641 R_2 - .009905691 R_3 \\ 0 &= u_2 - .602305641 R_1 + .0016432021 R_2 + .0046112819 R_3 \\ 0 &= u_3 - .009905691 R_1 + .0046112819 R_2 + .0198113820 R_3 \end{aligned} \right\}$$

$$\left. \begin{aligned} 0 &= 139.8225 u_1 + R_1 - .3223805 R_2 - 1.385038 R_3 \\ 0 &= 433.7188 u_2 - R_1 + .7126876 R_2 + 2.000000 R_3 \\ 0 &= -100.9520 u_3 + R_1 - .4655181 R_2 - 2.000000 R_3 \end{aligned} \right\}$$

$$\left. \begin{aligned} -139.8225 u_1 - 433.7188 u_2 &= .3903071 R_2 + .614962 R_3 \\ 100.9520 u_3 - 433.7188 u_2 &= .2471695 R_2 + \quad \quad \end{aligned} \right\}$$

$$R_2 = 408.4322 u_3 - 1754.7423 u_2$$

$$.614962 R_3 = -139.8225 u_1 - 433.7188 u_2 - 159.4140 u_3 + 694.8884 u_2$$

$$R_3 = -227.3676 u_1 + 408.4309 u_2 - 259.2257 u_3$$

$$R_1 = 100.9520 u_3 + .4655181 (408.4322 u_3 - 1754.7423 u_2) + 2 R_3$$

$$= 100.9520 u_3$$

$$+ 190.1326 u_3 - 816.8643 u_2$$

$$- 518.4514 u_3 + 816.8618 u_2 - 454.7352 u_1$$

$$= -454.7352 u_1 - .0025 u_2 - 227.3668 u_3$$



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$$R_1 = 2.187404 P_1 + 2.283277 P_2 - 3.106300 Q_1 + .00766351 Q_2 \\ - .000004 P_1 - .000004 P_2 + .000006 Q_1 \\ - 1.544628 P_1 - 1.640500 P_2 + 2.252225 Q_1$$

$$= .642772 P_1 + .642773 P_2 - .854069 Q_1 + .00766351 Q_2$$

$$R_2 = 2.774705 P_1 + 2.946926 P_2 - 4.045803 Q_1 \\ - 2.825770 P_1 - 2.895856 P_2 + 4.045806 Q_1$$

$$= -.051065 P_1 + .051070 P_2$$

$$R_3 = 1.093702 P_1 + 1.141639 P_2 - 1.553150 Q_1 + .000383176 Q_2 \\ + .657722 P_1 + .674035 P_2 - .941695 Q_1 \\ - 1.761063 P_1 - 1.870369 P_2 + 2.567810 Q_1$$

$$= -.009639 P_1 - .054695 P_2 + .072965 Q_1 + .000383176 Q_2$$

Substituting for R 's, the first 4 equations on sheet No 14 became;

$$\delta_1 = .0034130747 P_1 + .0035380445 P_2 - .004812693 Q_1 - .000000039910 Q_2 \\ - .003091915 P_1 - .003091920 P_2 + .004108313 Q_1 - .000003866365 Q_2 \\ - .0000892233 P_1 + .000082241 P_2 \\ - .000065483 P_1 - .000371573 P_2 + .000495692 Q_1 + .00000260313 Q_2$$

$$= .00017344 P_1 + .00015679 P_2 - .00020869 Q_1 - .0000011232 Q_2$$

$$\delta_2 = .0035380445 P_1 + .0037112326 P_2 - .0050235245 Q_1 - .00000003991 Q_2 \\ - .003227431 P_1 - .003227436 P_2 + .004288378 Q_1 - .00000384794 Q_2 \\ - .000084273 P_1 + .000084281 P_2 \\ - .000069547 P_1 - .000394636 P_2 + .00052646 Q_1 + .00000276470 Q_2$$

$$= .00015679 P_1 + .00017344 P_2 - .00020869 Q_1 - .0000011232 Q_2$$

When $P_2 = P_1 = P$, & $Q = Q_1 = Q_2$; then $\delta_1 = \delta_2 = .00033023 P$ (3028. #/IN.)
 $P_2 = -P_1 = -P$, & $Q = Q_1 = Q_2$; then $\delta_1 = -\delta_2 = .00001665 P$ (66,700. #/IN.)



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$$\delta_3 = -.0048126930 P_1 - .0050235245 P_2 + .0068344184 Q_1 + .0000005644 Q_2$$

$$+ .0042907814 P_1 + .00439078 P_2 - .0058341531 Q_1 + .000005234951 Q_2$$

$$+ .000117738 P_1 - .0001177491 P_2$$

$$+ .000095481 P_1 + .000541792 P_2 - .000222769 Q_1 - .000003795623 Q_2$$

$$= \underline{-.00020869 P_1 - .00020869 P_2 + .00027750 Q_1 + .0000014958 Q_2}$$

$$\delta_4 = -.00000039910 P_1 - .00000039910 P_2 + .0000005644 Q_1 + .00000028836 Q_2$$

$$- .0000108324 P_1 - .0000108325 P_2 + .00000143934 Q_1 - .00000001292 Q_2$$

$$= \underline{-.000011232 P_1 - .000011232 P_2 + .0000014958 Q_1 + .000000027544 Q_2}$$

Note: If $P_1 = P_2$ & $\delta_3 = 0$; & $Q_2 = 0$,

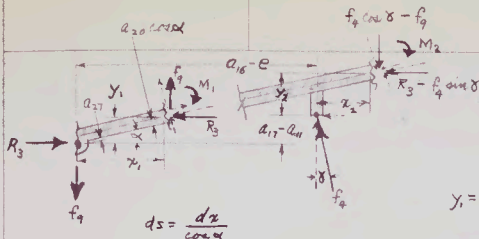
$$Q_1 = \frac{2 \times .00020869}{.00027750} P = 1.50407 P$$

$$\text{Then } \delta_1 = .00033023 P - .00020869 \times 1.50407 P = .00001635 P$$

This checks with δ_1 when $P_1 = -P_2$ and $Q_1 = 0$, & $Q_2 = 0$.

(Difference is due to strut deflection when $P_1 = -P_2$).

→



$$\underline{28} = \tan \alpha$$

$$\underline{29} = \sin \alpha$$

$$\underline{30} = \cos \alpha$$

$$y_1 = \left[\frac{a_{21}}{30} + \frac{30 a_{20}}{30} \right] + \underline{28} x_1 = \underline{31} + \underline{28} x_1$$

$$a_{17} - a_{11} + y_2 = \underline{31} + \underline{28} (a_{16} - c + x_2)$$

$$y_2 = \left[\underline{31} + \frac{61}{\underline{28}} (a_{16} - c) \right] + \underline{28} x_2 - (a_{17} - a_{11})$$

Alternate Spar Energy Function:

$$N_1 = 30 R_3 - 29 f_9$$

$$V_1 = 29 R_3 + 30 f_9$$

$$M_1 = R_3 y_1 + f_9 x_1 = \underline{31} R_3 + \underline{28} R_3 x_1 + f_9 x_1 = \underline{31} R_3 + (f_9 + \underline{28} R_3) x_1$$

$$U_{SP, M_1} = \int_0^{a_{16}-c} \left(\frac{1}{2 E_{SP} I_{SP}} \right) M_1^2 dx_1 + \left(\frac{62}{230 E_{SP} A_{SP}} \right) N^2 + \left(\frac{60 V_{SP}}{230 A_{SP} S_{SP}} \right) V^2$$

$$= \int_0^{60} \left[\underline{31}^2 R_3^2 + 2 \underline{31} R_3 (f_9 + \underline{28} R_3) x_1 + (f_9 + \underline{28} R_3)^2 x_1^2 \right] dx_1 + 62 N^2 + 63 V^2$$

$$= \left(\underline{32} \underline{31}^2 \underline{60} \right) R_3^2 + \left(\underline{32} \underline{31} \underline{60} \right) R_3 (f_9 + \underline{28} R_3) + \left(\frac{\underline{66}}{3} \right) (f_9 + \underline{28} R_3)^2 + 62 N^2 + 63 V^2$$

$$= \underline{64} R_3^2 + \underline{65} R_3 f_9 + \underline{65} \underline{28}^2 R_3^2 + \underline{66} f_9^2 + \underline{66} \underline{28} \underline{60} R_3 f_9 + \underline{66} \underline{28}^2 R_3^2 + \underline{62} \underline{30}^2 R_3^2 - \underline{29} \underline{29} \underline{60} \underline{62} R_3 f_9$$

$$+ \underline{62} \underline{29}^2 f_9^2 + \underline{63} \underline{29}^2 R_3^2 + \underline{2} \underline{29} \underline{30} \underline{63} R_3 f_9 + \left(\frac{\underline{63}}{\underline{30}} \underline{30} \right) f_9^2$$

$$= R_3^2 (\underline{64} + \underline{67} + \underline{69} + \underline{70} + \underline{73}) + R_3 f_9 (\underline{65} + \underline{68} - \underline{21} + \underline{74}) + f_9^2 (\underline{66} + \underline{72} + \underline{25})$$

$$N_2 = (f_9 \cos \alpha - f_9) \sin \alpha + (R_3 - f_9 \sin \alpha) \cos \alpha$$

$$\underline{307} = \sin \alpha ; \underline{308} = \cos \alpha$$

$$V_2 = (f_9 \cos \alpha - f_9) \cos \alpha - (R_3 - f_9 \sin \alpha) \sin \alpha$$

$$M_2 = R_3 (\underline{c1} + \underline{28} x_2) + f_9 (a_{16} - c + x_2) - f_9 \cos \alpha x_2 - f_9 \sin \alpha (\underline{c1} + \underline{28} x_2 - (a_{17} - a_{11}))$$

$$N_2 = \left(\underline{308} f_4 - f_9 \right) \underline{29} + \left(R_3 - \underline{307} f_9 \right) \underline{30} = \underline{29} u + \underline{30} w$$

$$V_2 = \left(\underline{308} f_4 - f_9 \right) \underline{30} - \left(R_3 - \underline{307} f_9 \right) \underline{29} = \underline{30} u - \underline{29} w$$

$$M_2 = x_2 \left[\underline{28} R_3 + f_9 - f_9 \underline{308} - f_9 \underline{307} \underline{28} \right] + \underline{c1} (R_3 - \underline{307} f_9) + (a_{16} - c) (f_9) + (a_{17} - a_{11}) \underline{307} f_9$$

$$= (\underline{28} w - u) x_2 + \left[\underline{c1} w + f_9 (a_{16} - c) + \underline{307} f_9 (a_{17} - a_{11}) \right]$$



$$\begin{aligned}
 U_{SP_{LH_2}} &= \int_0^{\theta} 32 \left[(28N - 4)X_2^2 + 2w(28N - 4)X_2 + w^2 \right] dX_2 + \left[\frac{310}{60} 6L \right] N^2 + \left[\frac{311}{60} 6L \right] V^2 \\
 &= \left(\frac{32}{3} \frac{e^3}{3} \right) (28^2 N^2 - 2 \cdot 28 \cdot 4N + 4^2) + \left(\frac{32}{3} e^3 \cdot 28 \right) Nw - \frac{314}{(32 e^3)} w^2 + \frac{315}{(32 e)} w^2 \\
 &\quad + \frac{316}{310} 29^2 U^2 + 2 \frac{317}{310} 29 \cdot 30 UN + \frac{318}{310} 30^2 N^2 + \frac{319}{311} 30^2 U^2 - 2 \frac{320}{311} 29 \cdot 30 UN + \frac{321}{311} 29^2 N^2 \\
 &= \frac{322}{312} 28^2 N^2 - 2 \frac{323}{312} 28 \cdot 4N + \frac{312}{312} 4^2 + \frac{313}{312} Nw - \frac{314}{312} w^2 + \frac{315}{312} w^2 + \frac{316}{312} U^2 \\
 &\quad + \frac{317}{312} UN + \frac{318}{312} N^2 + \frac{319}{312} U^2 - \frac{320}{312} UN + \frac{321}{312} N^2 \\
 &= N^2 \left(\frac{322}{312} + \frac{324}{312} + \frac{321}{312} \right) + U^2 \left(\frac{325}{312} + \frac{326}{312} + \frac{319}{312} \right) + w^2 \left(\frac{315}{312} \right) + UN \left(-\frac{323}{312} + \frac{317}{312} - \frac{320}{312} \right) \\
 &\quad + Nw \left(\frac{313}{312} \right) + w^2 \left(-\frac{314}{312} \right) \\
 &= \frac{324}{312} N^2 + \frac{325}{312} U^2 + \frac{326}{312} UN + \frac{327}{315} 61^2 N^2 + \frac{328}{315} 60^2 f_9^2 + \frac{329}{315} 307^2 307^2 f_4^2 \\
 &\quad + 2 \frac{330}{315} 61 \cdot 60 f_9 N + 2 \frac{331}{315} 61 \cdot 307 \cdot 307 f_4 N + 2 \frac{332}{315} 60 \cdot 307 \cdot 307 f_4 f_9 + \frac{333}{313} 61 N^2 \\
 &\quad + \frac{334}{313} 60 f_9 N + \frac{335}{313} 307 \cdot 307 f_4 N - \frac{336}{314} 61 UN - \frac{337}{314} 60 f_9 U - \frac{338}{314} 307 \cdot 307 f_4 U \\
 &= N^2 \left(\frac{339}{324} + \frac{327}{322} + \frac{333}{323} \right) + U^2 \left(\frac{325}{325} \right) + UN \left(\frac{326}{322} - \frac{336}{326} \right) + f_9^2 \left(\frac{328}{328} \right) + f_4^2 \left(\frac{329}{329} \right) \\
 &\quad + f_4 f_9 \left(\frac{332}{322} \right) + f_9 N \left(\frac{330}{320} + \frac{334}{324} \right) + f_4 N \left(\frac{331}{321} + \frac{335}{325} \right) + f_9 U \left(-\frac{337}{327} \right) + f_4 U \left(-\frac{338}{328} \right) \\
 &= \frac{339}{324} (R_3^2 - 2 \cdot 307 R_3 f_4 + 307^2 f_4^2) + \frac{325}{325} (308^2 f_4^2 - 2 \cdot 308 \cdot 60 f_9 + 60^2) \\
 &\quad + \frac{340}{328} (308 R_3 f_9 - 307 \cdot 308 f_4^2 - R_3 f_9 + 307 f_4 f_9) + \frac{328}{328} f_9^2 + \frac{329}{329} f_4^2 + \frac{332}{322} f_4 f_9 \\
 &\quad + \frac{341}{324} R_3 f_9 - \frac{341}{324} 307 f_4 f_9 + \frac{342}{324} R_3 f_4 - \frac{342}{324} 307 f_4^2 - \frac{337}{327} 308 f_4 f_9 + \frac{337}{327} f_9^2 - \frac{338}{328} 308 f_4^2 + \frac{336}{326} f_4 f_9 \\
 &= R_3^2 \left(\frac{339}{324} \right) + R_3 f_4 \left(-\frac{343}{324} 307 \cdot \frac{339}{324} + \frac{340}{324} 308 + \frac{342}{324} \right) + f_4^2 \left(\frac{329}{329} 307^2 + \frac{325}{325} 308^2 \right) \\
 &\quad - \frac{344}{327} 307 \cdot \frac{340}{324} + \frac{349}{329} - \frac{350}{324} 307 - \frac{348}{324} 308 + f_4 f_9 \left(-\frac{352}{327} 308 \cdot \frac{353}{329} + \frac{347}{327} 340 + \frac{354}{326} \right) \\
 &\quad - \frac{354}{324} 307 - \frac{355}{327} 308 + \frac{338}{328} + f_9^2 \left(\frac{325}{325} + \frac{328}{328} + \frac{337}{327} \right) + R_3 f_9 \left(-\frac{340}{340} + \frac{341}{341} \right) \\
 U_{SP_{LH_2}} &= R_3^2 \left(\frac{359}{26} + \frac{339}{26} \right) + R_3 f_9 \left(\frac{360}{28} + \frac{357}{22} \right) + f_9^2 \left(\frac{361}{28} + \frac{352}{22} \right) + R_3 f_4 \left(\frac{345}{24} \right) + f_4^2 \left(\frac{351}{24} \right) + f_4 f_9 \left(\frac{350}{24} \right) \\
 U_{SP} &= \frac{359}{26} R_3^2 + \frac{359}{26} f_9^2 + \frac{360}{26} R_3 f_9 + \frac{360}{26} f_9^2 + \frac{361}{26} f_9^2 + \frac{361}{26} f_9^2 + \frac{345}{24} R_3 f_4 + \frac{345}{24} f_4 f_9 + \frac{351}{24} f_4^2 + \frac{351}{24} f_4^2 \\
 &\quad + \frac{356}{24} f_4 f_9 + \frac{356}{24} f_4 f_9
 \end{aligned}$$



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4-5-55

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Proof of Strain Energy Reduction Technique:

The strain energy for a system with applied loads P_1, \dots, P_n ; and internal redundants R_1, \dots, R_m ; is of the form

$$\begin{aligned}
 U = & \underline{11} P_1^2 + \underline{12} P_1 P_2 + \dots + \underline{1n} P_1 P_n; + \underline{11} P_1 R_1 + \underline{12} P_1 R_2 + \dots + \underline{1m} P_1 R_m \\
 & + \underline{22} P_2^2 + \dots + \underline{2n} P_2 P_n; + \underline{21} P_2 R_1 + \underline{22} P_2 R_2 + \dots + \underline{2m} P_2 R_m \\
 & \dots \dots \dots \\
 & + \underline{nn} P_n^2; + \underline{n1} P_n R_1 + \underline{n2} P_n R_2 + \dots + \underline{nm} P_n R_m; \\
 & + \underline{11} R_1^2 + \underline{12} R_1 R_2 + \dots + \underline{1m} R_1 R_m \\
 & \dots \dots \dots \\
 & + \underline{22} R_2^2 + \dots + \underline{2m} R_2 R_m \\
 & \dots \dots \dots \\
 & + \underline{mm} R_m^2.
 \end{aligned}$$

where underlined numbers are constants, some of which may be zero.

If we consider only those terms which contain P_i when we take $\left[\frac{\partial U}{\partial P_i}\right]_0$; i.e., we treat $\frac{\partial R_j}{\partial P_i}$ as zero even though we know that

$$R_i = \frac{\partial R_i}{\partial P_1} P_1 + \frac{\partial R_i}{\partial P_2} P_2 + \dots + \frac{\partial R_i}{\partial P_n} P_n, \text{ then the displacement}$$

$$\delta_i = \left[\frac{\partial U}{\partial P_i}\right]_0 = \underline{1i} P_1 + \underline{2i} P_2 + \dots + \underline{ni} P_n; + \underline{1i} R_1 + \underline{2i} R_2 + \dots + \underline{mi} R_m.$$

$$0 = \left[\frac{\partial U}{\partial R_i}\right]_0 = \underline{1i} P_1 + \underline{2i} P_2 + \dots + \underline{ni} P_n; + \underline{1i} R_1 + \underline{2i} R_2 + \dots + \underline{2ii} R_i + \dots + \underline{mi} R_m.$$

Thus we obtain $(n+m)$ equations involving n P 's and m R 's. The last m equations can be transformed into the m equations

$$R_i = \left[\frac{\partial R_i}{\partial P_1}\right]_0 P_1 + \left[\frac{\partial R_i}{\partial P_2}\right]_0 P_2 + \dots + \left[\frac{\partial R_i}{\partial P_n}\right]_0 P_n.$$

Now substituting for the R 's in the first n equations,

$$\delta_{i,00} = P_i \left\{ \underline{1i} + \underline{1i} \left[\frac{\partial R_1}{\partial P_i}\right]_0 + \dots + \underline{im} \left[\frac{\partial R_m}{\partial P_i}\right]_0 \right\} + \dots + P_n \left\{ \underline{in} + \underline{1i} \left[\frac{\partial R_1}{\partial P_n}\right]_0 + \dots + \underline{im} \left[\frac{\partial R_m}{\partial P_n}\right]_0 \right\}.$$

We shall now differentiate U with respect to P_i recognizing that every R is a function of P_i . After subtraction of the terms shown above in δ_i , the remainder will be shown to be zero.



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$$\begin{aligned} \frac{\partial U}{\partial P_i} &= \frac{11}{1} P_1 + \frac{21}{2} P_2 + \dots + \frac{211}{n} P_n; + \frac{11}{1} R_1 + \frac{12}{2} R_2 + \dots + \frac{1m}{m} R_m; \leftarrow \text{This} = \delta_i \\ &+ \frac{\partial R_1}{\partial P_i} \left(\frac{11}{1} P_1 + \frac{21}{2} P_2 + \dots + \frac{111}{n} P_n; + \frac{21}{2} R_1 + \frac{12}{2} R_2 + \dots + \frac{1m}{m} R_m \right); \leftarrow \text{This} = \frac{\partial R_1}{\partial R_i} \left[\frac{\partial U}{\partial R_i} \right] \\ &+ \frac{\partial R_2}{\partial P_i} \left(\frac{12}{2} P_1 + \frac{22}{2} P_2 + \dots + \frac{122}{n} P_n; + \frac{12}{2} R_1 + \frac{222}{2} R_2 + \dots + \frac{21m}{m} R_m \right); \leftarrow \text{This} = \frac{\partial R_2}{\partial R_i} \left[\frac{\partial U}{\partial R_i} \right] \\ &\vdots \\ &+ \frac{\partial R_m}{\partial P_i} \left(\frac{1m}{m} P_1 + \frac{2m}{m} P_2 + \dots + \frac{1mm}{n} P_n; + \frac{1m}{m} R_1 + \frac{2m}{m} R_2 + \dots + \frac{2mm}{m} R_m \right); \leftarrow \text{This} = \frac{\partial R_m}{\partial R_i} \left[\frac{\partial U}{\partial R_i} \right] \end{aligned}$$

Inasmuch as the deflection of one redundant member depends upon the deflections of other redundant members, the corresponding forces are dependent. When we consider $\frac{\partial U}{\partial R_i}$ we ought to consider $\frac{\partial R_i}{\partial R_i}$. Following the pattern of $\frac{\partial U}{\partial P_i}$,

$$0 = \frac{\partial U}{\partial R_i} = \frac{\partial R_1}{\partial R_i} \left[\frac{\partial U}{\partial R_1} \right] + \frac{\partial R_2}{\partial R_i} \left[\frac{\partial U}{\partial R_2} \right] + \dots + \frac{\partial R_m}{\partial R_i} \left[\frac{\partial U}{\partial R_m} \right].$$

Now, if we multiply this expression by $\frac{\partial R_i}{\partial P_i}$, we have

$$0 = \frac{\partial U}{\partial P_i} - \left[\frac{\partial U}{\partial R_i} \right]_0, \text{ which was to be proved}$$

To see that every term in $\frac{\partial U}{\partial R_i}$ is zero, consider

$$\left. \begin{aligned} 0 &= \frac{\partial U}{\partial R_1} = \left[\frac{\partial U}{\partial R_1} \right]_0 + \frac{\partial R_2}{\partial R_1} \left[\frac{\partial U}{\partial R_2} \right]_0 + \frac{\partial R_3}{\partial R_1} \left[\frac{\partial U}{\partial R_3} \right]_0 \\ 0 &= \frac{\partial U}{\partial R_2} = \frac{\partial R_1}{\partial R_2} \left[\frac{\partial U}{\partial R_1} \right]_0 + \left[\frac{\partial U}{\partial R_2} \right]_0 + \frac{\partial R_3}{\partial R_2} \left[\frac{\partial U}{\partial R_3} \right]_0 \\ 0 &= \frac{\partial U}{\partial R_3} = \frac{\partial R_1}{\partial R_3} \left[\frac{\partial U}{\partial R_1} \right]_0 + \frac{\partial R_2}{\partial R_3} \left[\frac{\partial U}{\partial R_2} \right]_0 + \left[\frac{\partial U}{\partial R_3} \right]_0 \end{aligned} \right\} \begin{aligned} 0 &= x + ay + bz \\ 0 &= \frac{1}{2}x + y + cz \\ 0 &= \frac{1}{8}x + \frac{1}{2}y + z \end{aligned}$$

$$0 = x = y = z$$

Nevertheless, one final doubt remains: If the coefficients of P_i in $\frac{\partial U}{\partial P_i}$ and $\frac{\partial U}{\partial R_i}$ are not the same as the coefficients of P_i in $\left[\frac{\partial U}{\partial R_i} \right]_0$ and $\left[\frac{\partial U}{\partial R_i} \right]_0$, it may be that $\left[\frac{\partial R_i}{\partial P_i} \right]_0 \neq \frac{\partial R_i}{\partial P_i}$.



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The coefficient of P_i in $\frac{\partial U}{\partial R_i}$ is $\left\{ \underline{j}_1 + \underline{j}_1 \frac{\partial R_1}{\partial R_i} + \underline{j}_2 \frac{\partial R_2}{\partial R_i} + \dots + \underline{j}_m \frac{\partial R_m}{\partial R_i} \right\}$; and in $\left[\frac{\partial U}{\partial R_i} \right]_0$ it is $\left\{ \underline{j}_i \right\}$.

The coefficient of P_j in $\frac{\partial U}{\partial R_j}$ is $\left\{ \underline{j}_1 + \underline{j}_2 \frac{\partial R_2}{\partial R_j} + \underline{j}_3 \frac{\partial R_3}{\partial R_j} + \dots + \underline{j}_m \frac{\partial R_m}{\partial R_j} \right\}$; and in $\left[\frac{\partial U}{\partial R_j} \right]_0$ it is $\left\{ \underline{j}_j \right\}$.

Now subtracting the respective coefficients, we see that the differences cannot both be zero. If the first difference is zero, then

$$0 = \underline{j}_1 \alpha_1 + \underline{j}_2 \alpha_2 + \dots + \underline{j}_m \alpha_m; \text{ but if the second difference is zero,}$$

$$0 = \underline{j}_2 \alpha_2 + \dots + \underline{j}_m \alpha_m. \text{ Therefore, the } (m+1)\text{th order matrix}$$

obtained by reduction of the strain energy function by the simplified method cannot be identical (Coefficient for Coefficient) to the matrix one ought to obtain by the long method. Moreover, we may conclude that none of the coefficients can correspond between matrices because $\frac{\partial^2 U}{\partial R_j \partial R_i} = \frac{\partial^2 U}{\partial R_i \partial R_j}$; i.e., in each matrix there are complementary

coefficients which must both correspond if one corresponds.

$$\begin{aligned} \frac{\partial U}{\partial R_i} &= P_i \left\{ \underline{i}_1 + \underline{i}_1 \frac{\partial R_1}{\partial R_i} + \underline{i}_2 \frac{\partial R_2}{\partial R_i} + \dots + \underline{i}_m \frac{\partial R_m}{\partial R_i} \right\} \\ &+ P_j \left\{ \underline{2}_1 + \underline{2}_1 \frac{\partial R_1}{\partial R_i} + \underline{2}_2 \frac{\partial R_2}{\partial R_i} + \dots + \underline{2}_m \frac{\partial R_m}{\partial R_i} \right\} \\ &\vdots \\ &+ P_n \left\{ \underline{i}_n + \underline{n}_1 \alpha_1 + \underline{n}_2 \alpha_2 + \dots + \underline{n}_m \alpha_m \right\} \\ &+ R_1 \left\{ \underline{i}_1 + \underline{2}_1 \alpha_1 + \underline{i}_2 \alpha_2 + \dots + \underline{i}_m \alpha_m \right\} \\ &+ R_2 \left\{ \underline{i}_2 + \underline{i}_2 \alpha_1 + \underline{2}_2 \alpha_2 + \dots + \underline{2}_m \alpha_m \right\} \\ &\vdots \\ &+ R_m \left\{ \underline{i}_m + \underline{i}_m \alpha_1 + \underline{2}_m \alpha_2 + \dots + \underline{2}_m \alpha_m \right\}. \end{aligned} \quad \left. \begin{array}{l} \alpha_j = \frac{\partial R_j}{\partial R_i} \\ \\ \\ \\ \\ \\ \\ \end{array} \right\} \text{Zero.}$$

On the other hand, if the co-factor of every R_j above is zero, it may be seen that the two methods are absolutely identical. The general expression for the cofactors is $\frac{\partial^2 U}{\partial R_i^2}$ of $\left[\frac{\partial U}{\partial R_i} \right]_0$; which certainly is zero.

$$0 = \left[\frac{\partial^2 U}{\partial R_j \partial R_i} \right]_0 + \left[\frac{\partial^2 U}{\partial R_i \partial R_j} \right]_0 \frac{\partial R_j}{\partial R_i} + \left[\frac{\partial^2 U}{\partial R_i \partial R_j} \right]_0 \frac{\partial R_2}{\partial R_i} + \dots + \left[\frac{\partial^2 U}{\partial R_m \partial R_j} \right]_0 \frac{\partial R_m}{\partial R_i}.$$

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$$a_{11}x_1 + a_{12}x_2 + C_1 = 0 = \frac{\partial U}{\partial R_1} = 0$$

$$a_{21}x_1 + a_{22}x_2 + C_2 = 0 = \frac{\partial U}{\partial R_2} = 0$$

$$\frac{\partial^2 U}{\partial R_1 \partial R_2} = \frac{\partial^2 U}{\partial R_2 \partial R_1}$$

$$\therefore a_{12} = a_{21}$$

$$a_{22}a_{11}x_1 + a_{22}a_{12}x_2 + a_{22}C_1 = 0$$

$$-a_{12}a_{21}x_1 - a_{12}a_{22}x_2 - a_{12}C_2 = 0$$

$$x_1 (a_{22}a_{11} - a_{12}a_{21}) = a_{12}C_2 - a_{22}C_1$$

$$x_1 = \frac{a_{12}C_2 - a_{22}C_1}{a_{22}a_{11} - a_{12}a_{21}}$$

$$x_2 = \frac{a_{11}C_2 - a_{21}C_1}{a_{21}a_{12} - a_{11}a_{22}}$$

Let

$$\beta = a_{11}a_{22} - a_{12}a_{21}$$

$$\therefore x_1 = \frac{a_{12}}{a_{11}a_{22} - a_{12}a_{21}} C_2 - \frac{a_{22}}{a_{11}a_{22} - a_{12}a_{21}} C_1$$

$$x_2 = \frac{a_{21}}{a_{11}a_{22} - a_{12}a_{21}} C_1 - \frac{a_{11}}{a_{11}a_{22} - a_{12}a_{21}} C_2$$

$$x_1 = \frac{a_{12}}{\beta} C_2 - \frac{a_{22}}{\beta} C_1$$

$$x_2 = -\frac{a_{21}}{\beta} C_2 + \frac{a_{11}}{\beta} C_1$$

Let $C_1 = b_{11}P_1 + b_{12}P_2 + \dots + b_{1n}P_n$

$C_2 = b_{21}P_1 + b_{22}P_2 + \dots + b_{2n}P_n$

$$x_1 = -\frac{a_{22}}{\beta} b_{11}P_1 - \frac{a_{22}}{\beta} b_{12}P_2 - \dots - \frac{a_{22}}{\beta} b_{1n}P_n; + \frac{a_{12}}{\beta} b_{21}P_1 + \frac{a_{12}}{\beta} b_{22}P_2 + \dots + \frac{a_{12}}{\beta} b_{2n}P_n$$

$$x_2 = \frac{a_{21}}{\beta} b_{11}P_1 + \frac{a_{21}}{\beta} b_{12}P_2 + \dots + \frac{a_{21}}{\beta} b_{1n}P_n; - \frac{a_{11}}{\beta} b_{21}P_1 - \frac{a_{11}}{\beta} b_{22}P_2 - \dots - \frac{a_{11}}{\beta} b_{2n}P_n$$

Theorem:

$$\sum_{k=1}^{K \times X} \frac{\partial^2 U}{\partial R_k \partial P_k} \frac{\partial P_k}{\partial P_j} = \sum_{k=1}^{K \times X} \frac{\partial^2 U}{\partial R_k \partial P_j} \frac{\partial P_k}{\partial P_k}$$

$$b_{11} \left[\frac{a_{12}}{\beta} b_{22} - \frac{a_{22}}{\beta} b_{12} \right] + b_{21} \left[\frac{a_{21}}{\beta} b_{12} - \frac{a_{11}}{\beta} b_{22} \right] = b_{12} \left[\frac{a_{12}}{\beta} b_{21} - \frac{a_{22}}{\beta} b_{11} \right] + b_{22} \left[\frac{a_{21}}{\beta} b_{11} - \frac{a_{11}}{\beta} b_{21} \right]$$

$$a_{12}b_{11}b_{22} - a_{22}b_{11}b_{12} + a_{21}b_{12}b_{21} - a_{11}b_{21}b_{22} = a_{12}b_{12}b_{21} - a_{22}b_{11}b_{12} + a_{21}b_{11}b_{22} - a_{11}b_{21}b_{22}$$

Inasmuch as $a_{12} = a_{21}$, we see that the check theorem is trivial. W.



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Deflection Formulas
Check Analysis

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V.J. Maci

4-4-55

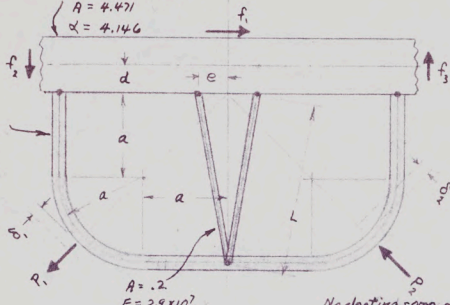
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$a = 27''$
 $d = 3.2''$
 $e = 8''$

$I = 33.72$,
 $E = 10^7$; $\nu = .4 \times 10^7$
 $A = 4.471$
 $\alpha = 4.146$

$I = 1.7382$
 $E = 10^7$
 $\nu = .4 \times 10^7$
 $A = 1.2104$
 $\alpha = 5.931$



Note: $P_2 = P_1$

By statics, $f_2 = f_3$

$$f_1 = \frac{P_1}{\sqrt{2}} + \frac{P_2}{\sqrt{2}}$$

$$f_2 + f_3 = \frac{2d \frac{P_1}{\sqrt{2}} + 2d \frac{P_2}{\sqrt{2}}}{2a}$$

$$\therefore f_2 = f_3 = \frac{Pd}{2\sqrt{2}a} + \frac{P_2 d}{2\sqrt{2}d}$$

$$= \frac{d}{2a} f_1$$

Neglecting some energy in the spar between the strut connections.

$$P_2 = 0$$

$$f_2 = f_3 = \frac{P_1 L}{2e}$$

$$f_2 = f_3 = \frac{P_1}{2} \quad (M_0 = 0)$$

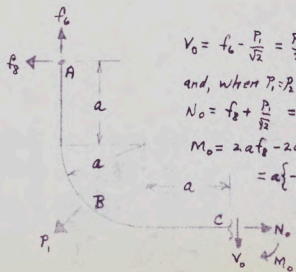
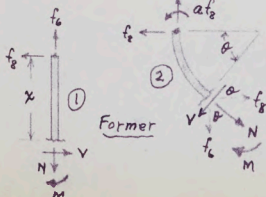
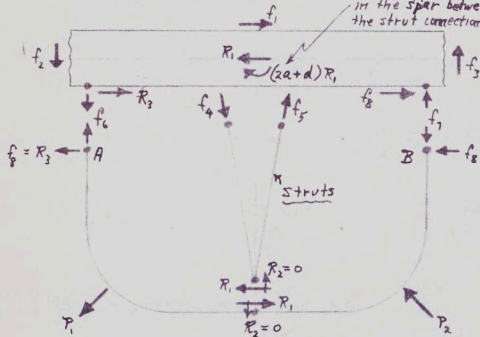
$$(M_B = 0)$$

$$f_2 = P_3 = \frac{P_1}{2} - \frac{P_1}{2\sqrt{2}} - \frac{P_2}{2\sqrt{2}}$$

$$= \frac{P_1}{2} - \frac{f_1}{2}$$

$$f_2 + f_3 = \frac{d}{2a} f_1 + \frac{P_1}{2} = f_1$$

$$f_3 + f_3 = \frac{d}{2a} f_1 + \frac{P_1}{2} = f_1$$



$$V_0 = f_2 - \frac{P_1}{\sqrt{2}} = \frac{P_1}{2} - \frac{P_1}{\sqrt{2}}$$

and, when $P_1 = P_2$, $V_0 = f_2$

$$N_0 = f_2 + \frac{P_1}{\sqrt{2}} = f_2$$

$$M_0 = 2af_2 - 2af_2 + a\sqrt{2}P_1$$

$$= a\sqrt{2}f_1 + \sqrt{2}P_1 = 0$$

$$\therefore U_{AB} = U_{BC}$$



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Strut Energy:

$$U_{ST} = 2 \frac{f_4^2 L}{2A_{ST} E_{ST}} = \frac{R_1^2 L^3}{4e^2 A_{ST} E_{ST}} = \frac{1}{4} R_1^2$$

$$\frac{1}{4} = \frac{L^3}{4e^2 A_{ST} E_{ST}}$$

Spar Energy:

$$U_{SP} = 2 \int_0^{2a} \left\{ \frac{M^2}{2E_{SP} I_{SP}} + \frac{N^2}{2A_{SP} E_{SP}} + \frac{\alpha_{SP} V^2}{2A_{SP} G_{SP}} \right\} dx$$

$$M = f_9 x + d f_8 ; \quad N = f_8 ; \quad V = f_9$$

$$U_{SP} = 2 \int_0^{2a} \left\{ \frac{f_9^2 x^2 + 2d f_8 f_9 x + d^2 f_8^2}{2E_{SP} I_{SP}} + \frac{f_8^2}{2A_{SP} E_{SP}} + \frac{\alpha_{SP} f_9^2}{2A_{SP} G_{SP}} \right\} dx$$

$$= \frac{1}{E_{SP} I_{SP}} \left(\frac{8a^3 f_9^2}{3} + 4ad f_8 f_9 + 2ad^2 f_8^2 \right) + \frac{2a}{A_{SP} E_{SP}} f_8^2 + \frac{2a \alpha_{SP}}{A_{SP} G_{SP}} f_9^2$$

$$\frac{2}{3} = \frac{2a}{E_{SP} I_{SP}} ; \quad \frac{3}{4} = \frac{2a}{A_{SP} E_{SP}} ; \quad \frac{4}{5} = \frac{2a \alpha_{SP}}{A_{SP} G_{SP}}$$

$$U_{SP} = \frac{2}{3} \frac{4a^3}{f_9} f_9^2 + \frac{2}{4} 2ad f_8 f_9 + \frac{2}{5} d^2 f_8^2 + \frac{3}{4} f_8^2 + \frac{4}{5} f_9^2$$

$$\frac{5}{6} = \frac{2}{3} \frac{4a^3}{f_9} + \frac{4}{5} ; \quad \frac{6}{7} = \frac{2}{4} d^2 + \frac{3}{4} ; \quad \frac{7}{8} = \frac{2}{5} 2ad$$

$$U_{SP} = \frac{6}{7} f_8^2 + \frac{7}{8} f_8 f_9 + \frac{5}{8} f_9^2$$

Former Energy:U_{F0AB}

$$V = f_8 ; \quad N = f_c ; \quad M = f_8 x$$

$$U_{F0AB} = \int_0^a \left(\frac{f_8^2 x^2}{6E_{F0} I_{F0}} + \frac{f_c^2 a}{2A_{F0} F_{F0}} + \frac{\alpha_{F0} f_8^2 a}{2A_{F0} G_{F0}} \right) dx$$

$$\frac{8}{9} = \frac{a^3}{6E_{F0} I_{F0}} ; \quad \frac{9}{10} = \frac{a}{2A_{F0} F_{F0}} ; \quad \frac{10}{11} = \frac{\alpha_{F0} a}{2A_{F0} G_{F0}} ; \quad \frac{11}{12} = \frac{8}{10}$$

$$U_{F0AB} = \frac{9}{12} f_c^2 + \frac{11}{12} f_8^2$$

$U_{FO_{AB_2}}$

$$N = f_2 \sin \theta + f_2 \cos \theta$$

$$V = f_2 \sin \theta - f_2 \cos \theta$$

$$M = a f_2 + f_2 a \sin \theta - f_2 a (1 - \cos \theta) = a (f_2 + f_2) + f_2 a \sin \theta + f_2 a \cos \theta$$

$$f_8 - f_2 = -\frac{f_1}{2}$$

$$ds = a d\theta$$

$$U_{FO_{AB_2}} = \int_0^{\frac{\pi}{4}} \left\{ 3 \left[\frac{f_1^2}{4} + f_2^2 \sin^2 \theta + f_2^2 \cos^2 \theta - f_2 f_2 \sin \theta - f_2 f_2 \cos \theta + f_2 f_2 \sin 2\theta \right] \right. \\ \left. + 2 (f_2^2 \sin \theta + f_2^2 \cos \theta + f_2 f_2 \sin 2\theta) + 10 (f_2^2 \sin^2 \theta + f_2^2 \cos^2 \theta - f_2 f_2 \sin 2\theta) \right\} da$$

$$12 = \int_0^{\frac{\pi}{4}} \sin^2 \theta d\theta = \left[\frac{\theta}{2} - \frac{\sin 2\theta}{4} \right]_0^{\frac{\pi}{4}} = .14270$$

$$13 = \int_0^{\frac{\pi}{4}} \cos^2 \theta d\theta = \left[\frac{\theta}{2} + \frac{\sin 2\theta}{4} \right]_0^{\frac{\pi}{4}} = .64270$$

$$14 = \int_0^{\frac{\pi}{4}} \sin \theta d\theta = [-\cos \theta]_0^{\frac{\pi}{4}} = .292893$$

$$15 = \int_0^{\frac{\pi}{4}} \cos \theta d\theta = [\sin \theta]_0^{\frac{\pi}{4}} = .707107$$

$$16 = \int_0^{\frac{\pi}{4}} \sin 2\theta d\theta = \left[-\frac{1}{2} \cos 2\theta \right]_0^{\frac{\pi}{4}} = .50000$$

$$U_{FO_{AB_2}} = .75 \times .7854 \times 8 f_1^2 + 3 \times 8 \times 12 f_2^2 + 3 \times 8 \times 13 f_2^2 - 3 \times 8 \times 14 f_1 f_2 - 3 \times 8 \times 15 f_1 f_2 + 3 \times 8 \times 16 f_1 f_2 \\ + 9 \times 12 f_2^2 + 9 \times 13 f_2^2 + 9 \times 16 f_1 f_2 + 10 \times 12 f_2^2 + 10 \times 13 f_2^2 - 10 \times 16 f_1 f_2$$

$$29 = 19 + 24 + 26 \qquad 32 = 9 + 29$$

$$30 = 18 + 23 + 27 \qquad 33 = 11 + 30$$

$$31 = 22 + 25 - 28$$

$$U_{FO_{AB_2}} = 17 f_1^2 + 29 f_2^2 + 30 f_2^2 - 21 f_1 f_2 - 20 f_1 f_2 + 31 f_1 f_2$$

$$U_{FO_{AB}} = 17 f_1^2 + 32 f_2^2 + 33 f_2^2 - 21 f_1 f_2 - 20 f_1 f_2 + 31 f_1 f_2$$

Inasmuch as $f_2 = f_1 = \frac{R_2}{2}$; $R_2 = f_2 = \frac{R_1 - f_1}{2}$; and $\gamma_1 = \gamma_2$; the strain energy of the right half equals the energy of the left half. Therefore $U_{20} = 4 U_{FO_{AB}}$.

$$34 = 6 + 433$$

$$U = 1 R_1^2 + 4 17 f_1^2 + 4 32 f_2^2 + 4 33 f_2^2 + 5 f_1^2 - 4 21 f_1 f_2 - 4 20 f_1 f_2 + 4 31 f_1 f_2 + 2 f_1 f_2$$



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$$\frac{\partial U}{\partial R_1} = 0 \quad ; \quad \frac{\partial}{\partial P} \left(\frac{\partial U}{\partial R_1} \right) = 0 \quad ; \quad \frac{\partial}{\partial F_2} \left(\frac{\partial U}{\partial R_1} \right) = 0 \quad . \quad \text{These yield } R_1 = \frac{\partial U}{\partial P} r_1 + \frac{\partial U}{\partial F_2} r_2$$

$$\frac{\partial U}{\partial P} = \delta_1 \quad ; \quad \frac{\partial \delta_1}{\partial P} = \frac{\partial^2 U}{\partial P^2} + \frac{\partial^2 U}{\partial P \partial R_1} \frac{\partial R_1}{\partial P} \quad ; \quad \frac{\partial \delta_1}{\partial F_2} = \frac{\partial^2 U}{\partial P \partial F_2} + \frac{\partial^2 U}{\partial P \partial R_1} \frac{\partial R_1}{\partial F_2}$$

$$\text{These yield } \delta_1 = \frac{\partial \delta_1}{\partial P} r_1 + \frac{\partial \delta_1}{\partial F_2} r_2$$

$$\frac{\partial U}{\partial F_2} = \delta_2 \quad ; \quad \frac{\partial \delta_2}{\partial P} = \frac{\partial \delta_1}{\partial F_2} \quad ; \quad \frac{\partial \delta_2}{\partial F_2} = \frac{\partial^2 U}{\partial F_2^2} + \frac{\partial^2 U}{\partial F_2 \partial R_1} \frac{\partial R_1}{\partial F_2}$$

$$\text{and, } \delta_2 = \frac{\partial \delta_2}{\partial P} r_1 + \frac{\partial \delta_2}{\partial F_2} r_2$$

When we differentiate the strain energy function,

$$\text{for } \frac{\partial U}{\partial P}, \text{ then } \frac{\partial P}{\partial P} = 1, \text{ and } \frac{\partial F}{\partial R} = \frac{\partial F}{\partial R} \quad ; \quad \text{This also for}$$

$$\text{for } \frac{\partial U}{\partial P}, \text{ then } \frac{\partial P}{\partial P} = 1, \text{ and } \frac{\partial F}{\partial P} = \frac{\partial F}{\partial P} \quad ; \quad \left. \begin{array}{l} \frac{\partial U}{\partial R_1}, \frac{\partial U}{\partial F_2}, \text{ and } \frac{\partial U}{\partial R_2} \end{array} \right\}$$

$$\text{for } \frac{\partial}{\partial P} \left(\frac{\partial U}{\partial R} \right), \text{ then } \frac{\partial R}{\partial P} = \frac{\partial R}{\partial P}$$

Note: for only two applied forces and one redundant this method is not the shortest that we might use. We use it now to check a more complicated problem.

$$\frac{\partial U}{\partial X} = 21 r_1 \frac{\partial R_1}{\partial X} + 817 f_1 \frac{\partial f_1}{\partial X} + 832 f_2 \frac{\partial f_2}{\partial X} + 234 f_3 \frac{\partial f_3}{\partial X} + 25 f_4 \frac{\partial f_4}{\partial X} - 421 f_5 \frac{\partial f_5}{\partial X} - 421 f_6 \frac{\partial f_6}{\partial X}$$

$$- 420 f_7 \frac{\partial f_7}{\partial X} - 420 f_8 \frac{\partial f_8}{\partial X} + 431 f_9 \frac{\partial f_9}{\partial X} + 431 f_{10} \frac{\partial f_{10}}{\partial X} + 1 f_{11} \frac{\partial f_{11}}{\partial X} + 1 f_{12} \frac{\partial f_{12}}{\partial X}$$

$$\text{Note: } \frac{\partial f_i}{\partial X} \text{ are constants. Therefore } \frac{\partial}{\partial Y} \left(\frac{\partial f_i}{\partial X} \right) = 0. \text{ Also } \frac{\partial}{\partial Y} \left(\frac{\partial R_1}{\partial X} \right) = 0.$$

$$\frac{\partial^2 U}{\partial X \partial Y} = 21 \frac{\partial R_1}{\partial Y} \frac{\partial R_1}{\partial X} + 817 \frac{\partial f_1}{\partial Y} \frac{\partial f_1}{\partial X} + 832 \frac{\partial f_2}{\partial Y} \frac{\partial f_2}{\partial X} + 234 \frac{\partial f_3}{\partial Y} \frac{\partial f_3}{\partial X} + 25 \frac{\partial f_4}{\partial Y} \frac{\partial f_4}{\partial X} - 421 \frac{\partial f_5}{\partial Y} \frac{\partial f_5}{\partial X}$$

$$- 421 \frac{\partial f_6}{\partial Y} \frac{\partial f_6}{\partial X} - 420 \frac{\partial f_7}{\partial Y} \frac{\partial f_7}{\partial X} - 420 \frac{\partial f_8}{\partial Y} \frac{\partial f_8}{\partial X} + 431 \frac{\partial f_9}{\partial Y} \frac{\partial f_9}{\partial X} + 431 \frac{\partial f_{10}}{\partial Y} \frac{\partial f_{10}}{\partial X} + 1 \frac{\partial f_{11}}{\partial Y} \frac{\partial f_{11}}{\partial X} + 1 \frac{\partial f_{12}}{\partial Y} \frac{\partial f_{12}}{\partial X}$$

$$\frac{\partial}{\partial P} \left(\frac{\partial U}{\partial R_1} \right) = \frac{\partial^2 U}{\partial P \partial R_1} + \frac{\partial^2 U}{\partial R_1^2} \frac{\partial R_1}{\partial P} = 0$$

$$\frac{\partial}{\partial F_2} \left(\frac{\partial U}{\partial R_1} \right) = \frac{\partial^2 U}{\partial F_2 \partial R_1} + \frac{\partial^2 U}{\partial R_1^2} \frac{\partial R_1}{\partial F_2} = 0$$



AVRO AIRCRAFT LIMITED

TECHNICAL DEPARTMENT (Aircraft)

REPORT NO. 7-0558-8

SHEET NO. 05

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4-14-55

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For substitution in $\frac{\partial^2 U}{\partial x \partial y}$,

$$\frac{\partial f_1}{\partial x_1} = 1$$

$$\frac{\partial f_1}{\partial x_2} = 0$$

$$\frac{\partial f_1}{\partial x_3} = 0$$

$$\frac{\partial f_1}{\partial x_4} = 0$$

$$\frac{\partial f_1}{\partial x_5} = \frac{1}{\sqrt{2}}$$

$$\frac{\partial f_1}{\partial x_6} = \frac{1}{\sqrt{2}}$$

$$\frac{\partial f_2}{\partial x_1} = \frac{1}{2}$$

$$\frac{\partial f_2}{\partial x_2} = 0$$

$$\frac{\partial f_2}{\partial x_3} = 0$$

$$\frac{\partial f_2}{\partial x_4} = \frac{1}{2}$$

$$\frac{\partial f_2}{\partial x_5} = -\frac{1}{2\sqrt{2}}$$

$$\frac{\partial f_2}{\partial x_6} = -\frac{1}{2\sqrt{2}}$$

$$\frac{\partial f_3}{\partial x_1} = \frac{1}{2}$$

$$\frac{\partial f_3}{\partial x_2} = \frac{d}{2} \frac{1}{2\sqrt{2}}$$

$$\frac{\partial f_3}{\partial x_3} = \frac{d}{2} \frac{1}{2\sqrt{2}}$$

$$\frac{\partial^2 U}{\partial x_1^2} = 8 \frac{1}{2} + 2 \frac{34}{8} + 2 \frac{5}{2} \frac{d^2}{8} + 4 \frac{20}{4} + 4 \frac{20}{4} - 7 \frac{d}{8} - 7 \frac{d}{8}$$

$$= 4 \frac{1}{2} + .25 \frac{34}{2} + .25 \frac{d^2}{2} 5 + 2 \frac{20}{2} - .25 \frac{d}{2} 7$$

$$\frac{\partial^2 U}{\partial x_1 \partial x_2} = \frac{\partial^2 U}{\partial x_2^2} \text{ because all } \frac{\partial f}{\partial x_1} = \frac{\partial f}{\partial x_2}$$

$$\frac{\partial^2 U}{\partial x_1 \partial x_2} = -2 \frac{34}{4\sqrt{2}} + 2 \frac{5}{2} \frac{d}{2} \frac{1}{4\sqrt{2}} - 4 \frac{21}{2\sqrt{2}} - 4 \frac{20}{2\sqrt{2}} - 4 \frac{31}{4\sqrt{2}} - 7 \frac{1}{4\sqrt{2}} + 7 \frac{d}{2} \frac{1}{4\sqrt{2}}$$

$$= .707107 \left(-5 \frac{34}{2} + .5 \frac{d}{2} 5 - 2 \frac{21}{2} - 2 \frac{20}{2} - 3 \frac{1}{2} - .25 \frac{d}{2} 7 + .25 \frac{d}{2} 7 \right)$$

$$\frac{\partial^2 U}{\partial x_2^2} = \frac{\partial^2 U}{\partial x_1^2} \text{, because all } \frac{\partial f}{\partial x_1} = \frac{\partial f}{\partial x_2}$$

$$\frac{\partial^2 U}{\partial x_2 \partial x_3} = \frac{\partial^2 U}{\partial x_3 \partial x_2} \text{, a must.}$$

$$\frac{\partial^2 U}{\partial x_2 \partial x_1} = \frac{\partial^2 U}{\partial x_1 \partial x_2} \text{, because all } \frac{\partial f}{\partial x_1} = \frac{\partial f}{\partial x_2}$$

$$\frac{\partial^2 U}{\partial x_1^2} = 2 \frac{1}{2} + 8 \frac{32}{4} + 2 \frac{34}{4} + 2 \frac{5}{4} + 4 \frac{21}{4} + 4 \frac{21}{4} + 7 \frac{1}{4} + 7 \frac{1}{4}$$

$$= 2 \frac{1}{2} + 2 \frac{32}{2} + .5 \frac{34}{2} + .5 \frac{5}{2} + 2 \frac{21}{2} + .5 \frac{7}{2}$$

$$\frac{\partial R_1}{\partial x_1} = - \frac{\frac{\partial^2 U}{\partial x_1 \partial x_1}}{\frac{\partial^2 U}{\partial x_1^2}}$$

$$R_1 = 2 \frac{\partial R_1}{\partial x_1} \pi$$

$$\frac{\partial R_1}{\partial x_2} = - \frac{\frac{\partial^2 U}{\partial x_1 \partial x_2}}{\frac{\partial^2 U}{\partial x_1^2}} = \frac{\partial R_1}{\partial x_1}$$

$$\delta_1 = \delta_2 = 2 \frac{\partial \delta_1}{\partial x_1} \pi = 2 \pi \left[\frac{\partial^2 U}{\partial x_1^2} + \frac{\partial^2 U}{\partial x_1 \partial x_2} \frac{\partial R_1}{\partial x_1} \right]$$



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Computations:

$g = 10^{-7}$

1 = $\frac{(54.5894)^3}{4 \times 64 \times 2.2 \times 10^3 \times 2.9} = \underline{1095.61 g}$

2 = $\frac{54}{10^7 \times 93072} = \underline{1.60142 g}$

3 = $\frac{54}{4.411 \times 10^7} = \underline{12.0778 g}$

4 = $\frac{54 \times 4.146}{4.471 \times 4 \times 10^7} = \underline{125.1865 g}$

5 = $1.60142 g \times 1.33333 \times 27^2 + 125.1865 g = \underline{1681.767 g}$

6 = $1.60142 \times 3.2 \frac{2}{18.1985} + 12.0778 g = \underline{28.4763 g}$

7 = $1.60142 g \times 54 \times 3.2 = \underline{276.722 g}$

8 = $\frac{27^3}{6 \times 10^7 \times 1.7382} = \underline{1887.2972 g}$

9 = $\frac{27}{2 \times 1.2104 \times 10^7} = \underline{11.1533 g}$

10 = $\frac{5.931 \times 27}{2 \times 1.2104 \times 4 \times 10^7} = \underline{165.3755 g}$

11 = $1887.2972 g + 165.3755 g = \underline{2052.673 g}$

12 = .14270

13 = .64270

14 = .29289

15 = .70711

16 = .50000

17 = $.75 \times .78540 \times 1887.297 g = \underline{1111.712 g}$

18 = $3 \times 1887.297 g \times .14270 = \underline{807.9519 g}$

19 = $3 \times 1887.297 g \times .64270 = \underline{3638.897 g}$

20 = $3 \times 1887.297 g \times .29289 = \underline{1658.311 g}$

21 = $3 \times 1887.297 g \times .70711 = \underline{4003.580 g}$

22 = $3 \times 1887.297 g \times .50000 = \underline{2830.9455 g}$



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$$23 = 11.1533 \text{ g} \times .14270 = \underline{1.5916 \text{ g}}$$

$$24 = 11.1533 \text{ g} \times .64270 = \underline{7.1682 \text{ g}}$$

$$25 = 11.1533 \text{ g} \times .50000 = \underline{5.5766 \text{ g}}$$

$$26 = 165.3755 \text{ g} \times .14270 = \underline{23.5991 \text{ g}}$$

$$27 = 165.3755 \text{ g} \times .64270 = \underline{106.2868 \text{ g}}$$

$$28 = 165.3755 \text{ g} \times .50000 = \underline{82.6878 \text{ g}}$$

$$29 = 3638.897 \text{ g} + 7.1682 \text{ g} + 23.5991 \text{ g} = \underline{3669.6643 \text{ g}}$$

$$30 = 807.9519 \text{ g} + 1.5916 \text{ g} + 106.2868 \text{ g} = \underline{915.8303 \text{ g}}$$

$$31 = 2830.9455 \text{ g} + 5.5766 \text{ g} - 82.6878 \text{ g} = \underline{2753.8343 \text{ g}}$$

$$32 = 11.1533 \text{ g} + 3669.6643 \text{ g} = \underline{3680.8176 \text{ g}}$$

$$33 = 2052.673 \text{ g} + 915.8303 \text{ g} = \underline{2968.5033 \text{ g}}$$

$$34 = 28.4763 \text{ g} + 4 \times 2968.5033 \text{ g} = \underline{11,902.4895 \text{ g}}$$

$$\frac{\partial^2 U}{\partial r^2} = 4 \times 1111.7128^2 + .25 \times 11,902.4895 \text{ g} + .25 \times \left(\frac{3.2}{27}\right)^2 \times 1681.767 \text{ g} + 2 \times 1458.311 \text{ g} - .25 \times \frac{3.2}{27} \times 276.722 \text{ g}$$

$$= \text{g} (4446.848 + 2975.622 + 8.904 + 3316.622 - 8.199)$$

$$= \underline{.001073680}$$

$$\frac{\partial^2 U}{\partial r^2} = \text{g} (2 \times 1095.61 + 2 \times 3680.8176 + .5 \times 11,902.4895 + .5 \times 1681.767 + 2 \times 2753.8343 + .5 \times 276.722)$$

$$= \text{g} (2191.220 + 7361.635 + 5951.245 + 840.884 + 5507.669 + 138.361)$$

$$= \underline{.002199105}$$

$$\frac{\partial^2 U}{\partial r^2 \partial r_1} = .70711 \text{ g} (-.5 \times 11,902.4895 + .5 \frac{3.2}{27} \times 1681.767 - 2 \times 4003.580 - 2 \times 1458.311 - 2753.8343 - .25 \times 276.722 + .25 \times \frac{3.2}{27} \times 276.722)$$

$$= .70711 \text{ g} (-5951.245 + 99.660 - 8007.160 - 3316.622 - 2753.8343 - 69.181 + 8.199)$$

$$= .70711 \text{ g} (-19,990.189)$$

$$= \underline{-.0014135258}$$



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$$\frac{\partial R_1}{\partial P} = -\frac{-0.001413526}{0.002199105} = +.642773$$

$$\delta_1 = 2P \left[.001073680 + \frac{-0.0009085762}{+.000165104} (-.001413526)(+.642773) \right]$$

$$\delta_1 = \frac{.00033021}{P}$$

$$\delta_2 = \frac{.00033021}{P}$$

} Note: Inherently $R_1 = R_2 = P$; i.e., no other condition can be considered in this set-up in which $R_2 = 0$, and $R_3 = f_8$.

$$\frac{P}{\delta} = \underline{\underline{3028. \#/\text{inch.}}}$$

Now, if we return to the strain energy function and ^{substitute} P for R_1 and for R_2 before we differentiate, and then say that $\frac{\partial U}{\partial P} = 2\delta$, because $\partial U = \delta \partial P + \delta_2 \partial R_2$

$$\frac{\partial f_1}{\partial P} = \sqrt{2} ; \quad \frac{\partial f_6}{\partial P} = 0 ; \quad \frac{\partial f_7}{\partial P} = -\frac{1}{\sqrt{2}} ; \quad \frac{\partial f_8}{\partial P} = \frac{d}{2a} \sqrt{2} = \frac{d}{a\sqrt{2}}$$

$$\frac{\partial f_1}{\partial R_1} = 0 ; \quad \frac{1}{2} = \frac{\partial f_1}{\partial R} = \frac{\partial f_6}{\partial R} = \frac{\partial f_7}{\partial R} \text{ as before.}$$

$$\frac{\partial R}{\partial P} = -\frac{\frac{\partial^2 U}{\partial P^2}}{\frac{\partial^2 U}{\partial R^2}} \text{ as before, but}$$

$$\delta = \frac{P}{2} \left(\frac{\partial^2 U}{\partial P^2} + \frac{\partial^2 U}{\partial R^2} \frac{\partial R}{\partial P} \right)$$

$$\begin{aligned} \frac{\partial^2 U}{\partial P^2} &= 8 \cdot 172 + 2 \cdot 34 \cdot \frac{1}{2} + 2 \cdot 5 \cdot \frac{d}{2a^2} + 4 \cdot 20 + 4 \cdot 20 - 7 \cdot \frac{d}{2a} - 7 \cdot \frac{d}{2a} \\ &= 16 \times 1111.712 \text{ g} + 11,902.4895 \text{ g} + \left(\frac{3.2}{27} \right) \times 1681.767 \text{ g} + P \times 1658.311 \text{ g} - 276.722 \text{ g} \cdot \frac{3.2}{27} \\ &= \text{g} (17,787.392 + 11,902.490 + 23.623 + 13,266.488 - 32.797) \\ &= \underline{\underline{.004294719}} \end{aligned}$$

$$\frac{\partial^2 U}{\partial R^2} = \underline{\underline{.002199105}} \text{ as before.}$$

$$\begin{aligned} \frac{\partial^2 U}{\partial P \partial R} &= 2 \cdot 34 \left(\frac{1}{2\sqrt{2}} \right) + 2 \cdot 5 \cdot \frac{d}{a} \cdot \frac{1}{2\sqrt{2}} - 4 \cdot 21 \cdot \frac{1}{\sqrt{2}} - 4 \cdot 20 \cdot \frac{1}{\sqrt{2}} - 4 \cdot 31 \cdot \frac{1}{2\sqrt{2}} - 7 \cdot \frac{1}{2\sqrt{2}} + 7 \cdot \frac{d}{a} \cdot \frac{1}{2\sqrt{2}} \\ &= .70711 \text{ g} (-11,902.4895 + 199.320 - 16,014.320 - 6633.244 - 5507.669 - 138.361 + 16.398) \\ &= \underline{\underline{-.0028270517}} \end{aligned}$$



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$$\frac{\partial R}{\partial P} = - \frac{-0.0028270517}{.002199105} = \underline{1.285546}$$

$$\delta = \frac{P}{2} \left(.004294719 - \frac{-0.003634205}{.000660414} \times 1.285546 \right)$$

$$\delta = \underline{.00033021} P$$

$$\frac{P}{\delta} = \underline{3028} \text{ \#/in. as before.}$$

Note:

The result obtained in "Stiffness of Former-Spar-Strut Assembly" of 2-4-55 was $\frac{P}{\delta} = 2947 \text{ \#/in.}$ The difference is attributed to the difference in type of Spar reactions.



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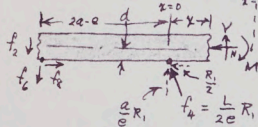
If we had not forces f_4 and f_5 as a pure moment acting at the midspan of the spar, would our answer have been much different?

$$U_{SP} = 2 \int_0^{2a-e} + 2 \int_{2a-e}^{2a} = U_{SP_1} + U_{SP_2} \quad \text{Say } \frac{2a-e}{2a} = \frac{35}{37}$$

$$U_{SP} = \frac{35^3}{37} (\frac{5}{2} - 4) f_9^2 + \frac{35^2}{37} 7 f_8 f_9 + \frac{35}{37} (4-3) f_8^2 + \frac{35}{37} 3 f_8^2 + \frac{35}{37} 4 f_9^2$$

$$\underline{36} = \frac{35}{37} \underline{6} ; \quad \underline{37} = \frac{35^2}{37} \underline{7} ; \quad \underline{38} = \frac{35}{37} (\frac{5}{2} - 4) + \frac{35}{37} \underline{4}$$

$$U_{SP} = \underline{36} f_8^2 + \underline{37} f_8 f_9 + \underline{38} f_9^2$$



$$N = f_8 - R_1 = -\frac{f_1}{2}$$

$$V = f_9 - \frac{a}{2e} R_1 = \frac{d}{2a} f_9 - R_1 (\frac{a}{2e} - \frac{1}{2})$$

$$M = f_9 (2a-e) + f_8 d + f_9 x - R_1 d - \frac{a}{e} R_1 x$$

$$= \frac{39}{2a} f_9 + \frac{39}{2} R_1 + d \frac{f_1}{2} - d \frac{f_1}{2} + (f_9 - \frac{a}{e} R_1) x$$

$$= f_9 (\frac{39}{2a} - \frac{a}{2}) + \frac{39}{2} R_1 + Vx$$

$$U_{SP_2} = \int_0^e \left\{ \frac{u^2 + 2uVx + V^2 x^2}{2AE_{SP} I_{SP}} + \frac{N^2}{2AE_{SP}} + \frac{M^2}{2A_{SP} G_{SP}} \right\} dx$$

$$\underline{40} = \frac{e}{2a} ; \quad \underline{41} = \underline{40} \underline{2} ; \quad \underline{42} = \underline{40} \underline{3} ; \quad \underline{43} = \underline{40} \underline{4} ; \quad \underline{44} = \frac{d}{2} (\frac{39}{2} - 1)$$

$$U_{SP_2} = \underline{41} u^2 + \underline{41} euV + \underline{41} \frac{e^2}{3} V^2 + \underline{42} N^2 + \underline{43} V^2 \quad \underline{45} = \frac{e^2}{3} + \underline{43}$$

$$= \underline{41} (44^2 f_9^2 + 39 \underline{44} f_9 R_1 + \frac{39^2}{4} R_1^2) + \underline{41} e (44 f_9 V + \frac{39}{2} R_1 V) + \underline{45} V^2 + \underline{42} N^2$$

$$= \underline{41} \underline{44}^2 f_9^2 + \underline{41} \underline{44} \underline{44} f_9 R_1 + \underline{41} \frac{39^2}{4} R_1^2 + e \underline{41} \underline{44} \frac{d}{2a} f_9^2 - e \underline{41} \underline{44} \frac{35}{2} f_9 R_1$$

$$+ e \underline{41} \frac{39}{4} \frac{d}{a} f_9 R_1 - e \underline{41} \frac{39}{2} \frac{35}{2} R_1^2 + \underline{45} \frac{d^2}{4a^2} f_9^2 - \underline{45} \frac{d}{a} \frac{35}{2} f_9 R_1 + \underline{45} \frac{35^2}{e^2} R_1^2 + \underline{42} \frac{f_1^2}{4}$$

$$= f_9^2 (\underline{46} + \underline{49} + \underline{53} + \underline{56}) + f_9 R_1 (\underline{47} - \underline{50} + \underline{51} - \underline{54}) + R_1^2 (\underline{48} - \underline{52} + \underline{55})$$



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The strain energy of the spar in the other case was $5f_8^2 + 2f_8f_9 + 5f_9^2$.
Then the difference in strain energy of the two cases is

$$U_1 - U_2 = (6 - 36)f_8^2 + (2 - 37)f_8f_9 + (5 - 39)f_9^2 - 57f_1^2 - 58f_1R_1 - 57R_1^2$$

$$U_2 = (1 + 59)R_1^2 + (41 + 57)f_1^2 + 432f_6^2 + (433 + 36)f_8^2 + 28f_9^2 - 421f_1f_6 - 420f_1f_9 \\ + 421f_6f_8 + 37f_8f_9 + 58f_1R_1$$

$$\frac{\partial U}{\partial X} = 260R_1 \frac{\partial R_1}{\partial X} + 261f_1 \frac{\partial f_1}{\partial X} + 832f_6 \frac{\partial f_6}{\partial X} + 262f_8 \frac{\partial f_8}{\partial X} + 239f_9 \frac{\partial f_9}{\partial X} - 421f_1 \frac{\partial f_6}{\partial X} - 421f_1 \frac{\partial f_9}{\partial X} \\ - 420f_1 \frac{\partial f_8}{\partial X} - 420f_6 \frac{\partial f_1}{\partial X} + 421f_6 \frac{\partial f_8}{\partial X} + 421f_8 \frac{\partial f_6}{\partial X} + 37f_8 \frac{\partial f_9}{\partial X} + 37f_9 \frac{\partial f_8}{\partial X} + 58f_1 \frac{\partial R_1}{\partial X} \\ + 58R_1 \frac{\partial f_1}{\partial X}$$

$$\frac{\partial^2 U}{\partial X^2} = 260 \frac{\partial R_1}{\partial Y} \frac{\partial R_1}{\partial X} + 261 \frac{\partial f_1}{\partial Y} \frac{\partial f_1}{\partial X} + 832 \frac{\partial f_6}{\partial Y} \frac{\partial f_6}{\partial X} + 262 \frac{\partial f_8}{\partial Y} \frac{\partial f_8}{\partial X} + 239 \frac{\partial f_9}{\partial Y} \frac{\partial f_9}{\partial X} - 421 \frac{\partial f_1}{\partial Y} \frac{\partial f_6}{\partial X} \\ - 421 \frac{\partial f_6}{\partial Y} \frac{\partial f_1}{\partial X} - 420 \frac{\partial f_1}{\partial Y} \frac{\partial f_8}{\partial X} - 420 \frac{\partial f_8}{\partial Y} \frac{\partial f_1}{\partial X} + 421 \frac{\partial f_6}{\partial Y} \frac{\partial f_8}{\partial X} + 421 \frac{\partial f_8}{\partial Y} \frac{\partial f_6}{\partial X} + 37 \frac{\partial f_8}{\partial Y} \frac{\partial f_9}{\partial X} \\ + 37 \frac{\partial f_9}{\partial Y} \frac{\partial f_8}{\partial X} + 58 \frac{\partial f_1}{\partial Y} \frac{\partial R_1}{\partial X} + 58 \frac{\partial R_1}{\partial Y} \frac{\partial f_1}{\partial X}$$

$$\frac{\partial^2 U}{\partial R_1^2} = 61 + 2562 + 2538 \frac{d}{a^2} + 20 + 20 - 2537 \frac{d}{a}$$

$$\frac{\partial^2 U}{\partial R_1 \partial R_1} = -353553 \frac{d}{a} + 353553 \frac{d}{a} - 1.44421421 - 1.44421420 - 707107 \frac{31}{a} - \frac{353553}{2} \frac{37}{a} + \frac{353553}{2} \frac{37}{a} + 707107 \frac{58}{a}$$

$$\frac{\partial^2 U}{\partial R_1^2} = 260 + 232 + 562 + 538 + 31 + 31 + 2537 + 2527$$

$$35 = \frac{54-8}{54} = .851851; \quad 35^2 = .725651; \quad 35^3 = .618148$$

$$36 = .851851 \times 28.4763 g = 24.2576 g$$

$$37 = .725651 \times 276.722 g = 200.804 g$$

$$38 = .618148 (1681.767 - 125.187) g + .851851 \times 125.1865 g = 962.1968 g + 106.6402 g = 1068.837 g$$

$$39 = 29 - 6 = 23$$

$$40 = \frac{8}{34} = .48148$$

$$41 = .148148 \times 160142 g = 23725 g$$



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$$42 = .148148 \times 12.0775 \text{ g} = 1.78920 \text{ g}$$

$$43 = .148148 \times 125.1865 \text{ g} = 18.5461 \text{ g}$$

$$44 = \frac{3.2}{2} \left(\frac{46}{27} - 1 \right) = 1.12593$$

$$45 = 2.3725 \text{ g} \times \frac{64}{3} + 18.5461 \text{ g} = 5.0613 \text{ g} + 18.5461 \text{ g} = 23.6064 \text{ g}$$

$$46 = 2.3725 \text{ g} \times 1.12593^2 = 30.0766 \text{ g}$$

$$47 = 46 \times 2.3725 \text{ g} \times 1.12593 = 12.2878 \text{ g}$$

$$48 = 46^2 \times 2.3725 \text{ g} \times .25 = 125.5053 \text{ g}$$

$$49 = 8 \times 2.3725 \text{ g} \times 1.12593 \times \frac{3.2}{54} = 1.26638 \text{ g}$$

$$50 = 2.3725 \text{ g} \times 1.12593 \times .851851 \times 27 = 6.1439 \text{ g}$$

$$51 = 8 \times 2.3725 \text{ g} \times 46 \times .25 \times \frac{3.2}{27} = 2.5869 \text{ g}$$

$$52 = 2.3725 \text{ g} \times 46 \times 5 \times .851851 \times 27 = 125.5052 \text{ g}$$

$$53 = 23.6064 \text{ g} \times \left(\frac{3.2}{54} \right) = .0828926 \text{ g}$$

$$54 = 23.6064 \text{ g} \times 3.2 \times .851851 \times 12.5 = 8.04352 \text{ g}$$

$$55 = 23.6064 \text{ g} \times .725651 \times \frac{3.2}{64} = 195.1211 \text{ g}$$

$$56 = 1.78920 \times .25 = .447325 \text{ g}$$

$$57 = (30.0766 + 1.266 + .0829 + 1.4473) \text{ g} = 30.7334 \text{ g}$$

$$58 = (12.2878 - 6.1439 + 2.5869 - 8.04352) \text{ g} = .6873 \text{ g}$$

$$59 = (125.5053 - 125.5052 + 195.1211) \text{ g} = 195.1211 \text{ g}$$

$$60 = 1095.61 \text{ g} + 195.12 \text{ g} = 1290.73 \text{ g}$$

$$61 = 4 \times 1111.712 + 30.7334 = 4477.581 \text{ g}$$

$$62 = 4 \times 2968.5033 + 24.2576 = 11,898.27 \text{ g}$$

$$\frac{\partial^2 U}{\partial R_1^2} = \text{g} (4477.581 + 2974.568 + 3.753 + 3316.622) = .00107725$$

$$\frac{\partial^2 U}{\partial R_1 \partial R_2} = \text{g} (353553) \sqrt{-11,898.27 + 126.677 - 16,014.32 - 6633.24 - 5507.67 - 100,402 + 11.90 + 1.33} - 40,013.955$$

$$= -.001414705$$

$$\frac{\partial^2 U}{\partial R_2^2} = \text{g} (2581.46 + 7361.635 + 5949.135 + 534.419 + 5507.668 + 100,402) = .002203472$$

$$\delta_1 = 2P_1 \left[\begin{array}{c} .000189961 \\ .642034 \\ .00107725 + (-.001414705) \left[\begin{array}{c} -.001414705 \\ .002203472 \end{array} \right] \end{array} \right] = .000337922 \text{ P}_1$$

$$\frac{\partial U}{\partial R_1} = 2959 \text{ #/IN} \quad (\text{Does make a difference. @ } 2\frac{1}{4} \%)$$



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The j^{th} cofactor is identical to the column of coefficients of of the j^{th} X in the expression for $\frac{\partial U}{\partial P}$ at the top of page "b". Observing that the coefficients of the $\frac{\partial U}{\partial P}$ X 's are mirrored about a diagonal through the right-hand portions of $\left[\frac{\partial U}{\partial X_i} \right]$, it must follow that $\underline{U}_i X_j$ of the first line can be replaced by $\frac{\partial X_j}{\partial P_i} (\underline{U}_1 P_1 + \underline{U}_2 P_2 + \dots + \underline{U}_n P_n)$.

Accordingly it may be seen that the expression for $\frac{\partial U}{\partial P}$ on page "c" is equivalent to the expression for δ_{i0} at the bottom of page "a". This peculiarity offers a desirable way of checking the results of a particular computation; e.g.,

$$f_i = \frac{\partial \delta_i}{\partial P_j} = \frac{\partial^2 U}{\partial P_i \partial P_j} = \frac{\partial^2 U}{\partial P_i \partial P_j} \frac{\partial X_j}{\partial P_i} + \dots + \frac{\partial^2 U}{\partial P_i \partial P_m} \frac{\partial X_m}{\partial P_j} = \frac{\partial^2 U}{\partial P_j \partial P_i} \frac{\partial X_j}{\partial P_i} + \dots + \frac{\partial^2 U}{\partial P_j \partial P_m} \frac{\partial X_m}{\partial P_i}$$

Rewriting this equality using the notation introduced on page 13,

$$\underline{U}_P X_j + \dots + \underline{U}_P X_m = \underline{U}_P X_j + \dots + \underline{U}_P X_m$$

Example! Let $i = 2$; and $j = 4$;

Using results on pages 24 to 30,

$$\underline{U}_P X_1 + \underline{U}_P X_2 + \underline{U}_P X_3 = \underline{U}_P X_1 + \underline{U}_P X_2 + \underline{U}_P X_3$$

$$\begin{array}{ll} (-50,211.14) \times (.000766351) & (-16,85307) \times (.642773) \\ (16,503.03) \times (.000000000) & (0.0) \times (.051070) \\ (72,152.15) \times (.000383176) & (0.0) \times (-.054695) \end{array}$$

$$-38.47936 + 0 + 27.64697 = -10.83270 + 0 + 0$$

$$-10.8324 = -10.8327 \quad \text{OK.}$$





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$$\theta_1 = 180 - 2 \sin^{-1} \frac{a_2}{\sqrt{a_2^2 + a_3^2}} =$$

$$r_1 = \frac{a_2}{\sin \theta_1} =$$

$$\theta_3 = 180 - 2 \sin^{-1} \frac{a_5}{\sqrt{a_5^2 + a_4^2}} =$$

$$r_2 = \frac{a_5}{\sin \theta_3} =$$

$$\theta_2 = \theta_3 + \sin^{-1} \frac{a_6}{r_2} =$$

$$\beta = \tan^{-1} \frac{r_2 + a_2 - a_4}{r_1 + a_5 - a_3} =$$

$$3 = a_{25} - a_{20} =$$

$$4 = \frac{1}{2 \cos \theta} =$$

$$5 = \frac{1}{2 \sin \theta} =$$

$$6 = \cos \beta =$$

$$7 = \sin \beta =$$

$$8 = a_{20} + a_{17} + a_2 + a_{13} =$$

$$9 = a_3 - a_{14} + a_{15} =$$

$$10 = 11 - 7 =$$

$$11 = \frac{6 \cdot 8 + 2 \cdot 9}{2 a_{16}} =$$

$$12 = \frac{a_{25}}{2 a_{16}} =$$

$$13 = \frac{1}{2 a_{16}} =$$

$$14 = a_2 + a_{13} =$$

$$15 = 2 a_{16} - 9 =$$

$$16 = 9 =$$

$$17 = a_{25} - a_{20} - a_{17} =$$

$$18 = \frac{7 \cdot 15 - 6 \cdot 14}{2 a_{16}} =$$

$$19 = - \frac{6 \cdot 14 + 7 \cdot 16}{2 a_{16}} =$$

$$20 = \frac{17}{2 a_{16}} =$$



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$$21 = 11 =$$

$$22 = 10 =$$

$$23 = 19 + 7 =$$

$$24 = 18 - 7 =$$

$$25 = \frac{L}{2 A_{ST} E_{ST}} =$$

$$26 = 2 \cdot 25 \cdot 5^2 =$$

$$27 = 2 \cdot 25 \cdot 4^2 =$$

$$28 = \tan \alpha =$$

$$29 = \sin \alpha =$$

$$30 = \cos \alpha =$$

$$31 = \frac{a_{27}}{30} + a_{20} =$$

$$32 = \frac{1}{2 \cdot 30 \cdot E_{SP} \cdot I_{SP}} =$$

$$33 = \frac{a_{16}}{2 \cdot 30 \cdot A_{SP} \cdot E_{SP}} =$$

$$34 = \frac{a_{16} \cdot \alpha_{SP}}{2 \cdot 30 \cdot A_{SP} \cdot G_{SP}} =$$

$$35 = 32 \cdot 31^2 \cdot a_{16} =$$

$$36 = 32 \cdot 31 \cdot a_{16}^2 =$$

$$27 = 36 \cdot 28 =$$

$$38 = 32 \cdot \frac{a_{16}^3}{3} =$$

$$39 = 38 \cdot 28^2 =$$

$$40 = 33 \cdot 30^2 =$$

$$299 = 2 \cdot 38 \cdot 28 =$$

$$41 = 2 \cdot 33 \cdot 30 \cdot 29 =$$

$$42 = 33 \cdot 29^2 =$$

$$43 = 34 \cdot 29^2 =$$

$$44 = 2 \cdot 34 \cdot 29 \cdot 30 =$$

$$45 = 34 \cdot 30^2 =$$

$$46 = 35 + 37 + 39 + 40 + 43 =$$

$$47 = 36 + 299 - 41 + 44 =$$

$$48 = 38 + 42 + 45 =$$

$$49 = 10 + 18 =$$

$$50 = 11 + 19 =$$



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$$\begin{aligned} 51 &= 20 - 12 = \\ 52 &= 21 + 24 = \\ 53 &= 22 + 23 = \\ 55 &= \tan \Omega = \frac{a_{14} - a_{15}}{a_{13}} = \end{aligned}$$

$$\begin{aligned} 56 &= \sin \Omega = \\ 57 &= \cos \Omega = \end{aligned}$$

$$82 = \frac{a_{13}^2}{6 E_{FO} I_{FO} 57} =$$

$$83 = \frac{a_{13}}{2 A_{FO} E_{FO} 57} =$$

$$84 = \frac{a_{13} \alpha_{FO}}{2 A_{FO} G_{FO} 57} =$$

$$85 = 82 + 83 56^2 + 84 57^2 =$$

$$88 = 2 55 82 + 2 56 57 (84 - 83) =$$

$$59 = 82 55^2 + 83 57^2 + 84 56^2 =$$

$$86 = r_1 + a_{15} - a_{14} =$$

$$87 = \frac{r_1}{2 E_{FO} I_{FO}} =$$

$$88 = \frac{r_1}{2 A_{FO} E_{FO}} =$$

$$89 = \frac{r_1 \alpha_{FO}}{2 A_{FO} G_{FO}} =$$

$$90 = .5 (8_1 - .5 \sin 2\theta) =$$

$$91 = .5 (8_1 + .5 \sin 2\theta) =$$

$$92 = 1 - \cos \theta_1 =$$

$$93 = \sin \theta_1 =$$

$$94 = .5 (1 - \cos 2\theta) =$$

$$95 = 87 8_1 a_3^2 =$$

$$96 = 2 87 8_1 a_{13} 56 =$$

$$97 = 87 8_1 86^2 =$$

$$98 = 87 90 r_1^2 =$$

$$99 = 87 91 r_1^2 =$$

$$100 = 2 87 92 r_1 a_{13} =$$

$$101 = 2 87 92 r_1 86 =$$

$$102 = 2 87 93 r_1 a_{13} =$$

$$103 = 2 87 93 r_1 86 =$$



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$$104 = 87 \ 94 \ r_2^2 =$$

$$105 = 88 \ 90 =$$

$$106 = 88 \ 91 =$$

$$107 = 88 \ 94 =$$

$$108 = 89 \ 90 =$$

$$109 = 89 \ 91 =$$

$$110 = 89 \ 94 =$$

$$111 = 95 + 98 + 100 + 105 + 109 =$$

$$112 = -96 - 101 + 102 + 104 + 107 - 110 =$$

$$113 = 97 + 99 - 103 + 106 + 108 =$$

$$114 = \cos \theta_2 =$$

$$115 = \sin \theta_2 =$$

$$116 = \frac{r_2}{2 F_{F0} I_{F0_2}} =$$

$$117 = \frac{r_2}{2 A_{F0_2} F_{F0}} =$$

$$118 = \frac{r_2 \sin \theta_2}{2 A_{F0_2} \sin \theta_2} =$$

$$119 = .5 (\theta_2 - .5 \sin 2\theta_2) =$$

$$120 = .5 (\theta_2 + .5 \sin 2\theta_2) =$$

$$121 = 1 - \cos \theta_2 =$$

$$122 = \sin \theta_2 =$$

$$123 = .5 (1 - \cos 2\theta_2) =$$

$$124 = 116 \theta_2 =$$

$$125 = 2 \ 116 \ \theta_2 \ r_2 =$$

$$126 = 116 \ \theta_2 \ r_2^2 =$$

$$127 = 116 \ 119 \ r_2^2 =$$

$$128 = 116 \ 120 \ r_2^2 =$$

$$129 = 2 \ 116 \ 121 \ r_2^2 =$$

$$130 = 2 \ 116 \ 121 \ r_2^2 =$$

$$131 = 2 \ 116 \ 122 \ r_2^2 =$$

$$132 = 2 \ 116 \ 122 \ r_2^2 =$$

$$133 = 116 \ 123 \ r_2^2 =$$

$$134 = 117 \ 119 =$$



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$$\begin{aligned}135 &= 117 \ 120 = \\136 &= 117 \ 123 = \\137 &= 118 \ 119 = \\138 &= 118 \ 120 = \\139 &= 118 \ 123 = \\140 &= 126 + 128 - 132 + 135 + 137 = \\141 &= 127 + 134 + 138 = \\142 &= -125 + 131 = \\143 &= 130 - 133 - 136 + 139 = \\144 &= 124 \ 142 = \\145 &= 124 \ 9^2 = \\146 &= 2 \ 124 \ 9 \ 14 = \\147 &= 140 \ 115^2 = \\148 &= 141 \ 142 = \\149 &= 143 \ 114 \ 115 = \\150 &= 147 + 148 + 149 = \\151 &= 140 \ 142 = \\152 &= 141 \ 115^2 = \\153 &= 143 \ 114 \ 115 = \\154 &= 151 + 152 - 153 = \\155 &= 2 \ 140 \ 114 \ 115 = \\156 &= 2 \ 141 \ 114 \ 115 = \\157 &= 143 \ (114^2 - 115^2) = \\158 &= 155 - 156 + 157 = \\159 &= 142 \ 14 \ 115 = \\160 &= 129 \ 14 \ 114 = \\161 &= 159 - 160 = \\162 &= 142 \ 14 \ 114 = \\163 &= 129 \ 14 \ 115 = \\164 &= 162 + 163 = \\165 &= 142 \ 9 \ 115 = \\166 &= 129 \ 9 \ 114 = \\167 &= -165 + 166 = \\168 &= 142 \ 9 \ 114 = \\169 &= 129 \ 9 \ 115 = \\170 &= -168 - 169 = \\171 &= 144 + 154 + 164 = \end{aligned}$$



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$$172 = 145 + 150 + 167 =$$

$$173 = 7^2 150 + 6^2 154 - 6 \cdot 7 158 =$$

$$174 = -146 + 158 + 161 + 170 =$$

$$175 = 2 \cdot 6 154 - 7(158 + 161) + 6 164 =$$

$$176 = 6(158 + 170) - 7(2 \cdot 150 + 167) =$$

$$177 = a_6 - a_7 =$$

$$178 = 14 + a_{29} =$$

$$179 = a_5 + a_6 =$$

$$180 = 6 a_{29} + 7 179 =$$

$$181 = \frac{1}{2 E_{F0} I_{F0_3}} =$$

$$182 = \frac{a_7}{2 A_{F0_3} E_{F0}} =$$

$$183 = \frac{a_7 \chi_{F0_3}}{2 A_{F0_3} G_{F0}} =$$

$$184 = 181 a_7 =$$

$$185 = 184 a_7 =$$

$$186 = 185 \frac{a_7}{3} + 183 =$$

$$187 = 184 178^2 =$$

$$188 = 184 180^2 =$$

$$189 = 184 177^2 =$$

$$190 = 2 184 178 180 =$$

$$191 = 2 184 178 177 =$$

$$192 = 2 184 180 177 =$$

$$193 = 185 178 =$$

$$194 = 185 180 =$$

$$195 = 185 177 =$$

$$196 = 185 7 178 =$$

$$197 = 185 7 180 =$$

$$198 = 185 7 177 =$$

$$199 = 2 186 7 =$$

$$200 = 186 7^2 =$$

$$201 = 2 182 6 =$$

$$202 = 182 6^2 =$$

$$203 = 187 + 182 =$$

$$204 = 188 + 197 + 200 + 202 =$$



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$$205 = 189 + 195 + 186 =$$

$$206 = 190 + 196 + 201 =$$

$$207 = -191 - 193 =$$

$$208 = -192 - 194 - 198 - 199 =$$

$$209 = 85 + 111 + 121 + 203 =$$

$$210 = 173 + 204 =$$

$$211 = 59 + 113 + 172 + 205 =$$

$$212 = 125 + 206 =$$

$$213 = 58 + 112 + 174 + 207 =$$

$$214 = 176 + 208 =$$

$$300 = 209 + 46 =$$

$$301 = 2 \ 210 =$$

$$302 = 2 \ 26 =$$

$$303 = 2 \ 27 =$$

$$304 = 2 \ 300 =$$

$$305 = 2 \ 211 =$$

$$306 = 2 \ 48 =$$

$$UP_1^2 = + 301 \quad =$$

$$+ 2 \ 18 \ 214 \quad =$$

$$+ 305 (18^2 + 24^2) \quad =$$

$$- 2 \ 6 \ 34 \ 213 \quad =$$

$$+ 6^2 \ 304 \quad =$$

$$- 2 \ 6 \ 47 \ 52 \quad =$$

$$+ 306 (49^2 + 52^2) \quad =$$

$$UP_2 = + 214 (19 + 24) \quad =$$

$$+ 305 (1819 + 2224) \quad =$$

$$- 6 \ 213 (23 + 24) \quad =$$

$$+ 6^2 \ 304 \quad =$$

$$- 6 \ 212 \quad =$$

$$- 6 \ 47 (52 + 53) \quad =$$

$$+ 306 (4950 + 5253) \quad =$$



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$$\begin{aligned}
 \underline{UPQ}_1 &= +20\ 214 & = \\
 &+ 20\ 305 (18 + 24) & = \\
 &+ 213 (24 - 6\ 20) & = \\
 &- 6\ 304 & = \\
 &- 47 (6\ 51 - 52) & = \\
 &+ 306\ 51 (49 + 52) & =
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UPQ}_2 &= +6\ 13\ 47 & = \\
 &- 13\ 306 (49 + 52) & =
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UPR}_1 &= 20\ 214 & = \\
 &+ 20\ 305 (18 + 24) & = \\
 &+ 213 (24 - 6\ 20) & = \\
 &- 6\ 304 & = \\
 &- 47 (6\ 20 - 52) & = \\
 &+ 20\ 306 (49 + 52) & =
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UPR}_2 &= -5\ 214 & = \\
 &- 5\ 305 (18 - 24) & = \\
 &- 5\ 6\ 213 & = \\
 &- 5\ 6\ 47 & = \\
 &- 5\ 306 (49 + 52) & =
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UPR}_3 &= + 212 & = \\
 &+ 213 (18 - 24) & = \\
 &+ 6\ 304 & = \\
 &+ 47 (49 - 52) & =
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UP_2^1} &= + 301 & = & \\
 &+ 2 \ 23 \ 214 & = & \\
 &- 2 \ 6 \ 212 & = & \\
 &+ 305 (19^2 + 23^2) & = & \\
 &- 2 \ 6 \ 23 \ 213 & = & \\
 &+ 6^2 \ 304 & = & \\
 &- 2 \ 6 \ 47 \ 53 & = & \\
 &+ 306 (50^2 + 53^2) & = &
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UP_2^9} &= + 20 \ 214 & = & \\
 &+ 212 & = & \\
 &+ 20 \ 305 (19 + 23) & = & \\
 &+ 213 (23 - 6 \ 20) & = & \\
 &- 6 \ 304 & = & \\
 &- 47 (6 \ 51 - 53) & = & \\
 &+ 51 \ 306 (50 + 53) & = &
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UP_2^9} &= + 6 \ 13 \ 47 & = & \\
 &- 13 \ 306 (50 + 53) & = &
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UP_2^7} &= + 20 \ 214 & = & \\
 &+ 212 & = & \\
 &+ 20 \ 305 (19 + 23) & = & \\
 &+ 213 (23 - 6 \ 20) & = & \\
 &- 6 \ 304 & = & \\
 &- 47 (6 \ 20 - 53) & = & \\
 &+ 20 \ 306 (50 + 53) & = &
 \end{aligned}$$

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$$\begin{aligned}
 \underline{UP_2^8} &= + .5 \ 214 & = & \\
 &- .5 \ 305 (19 - 23) & = & \\
 &- .5 \ 6 \ 213 & = & \\
 &- .5 \ 6 \ 47 & = & \\
 &- .5 \ 306 (50 - 53) & = &
 \end{aligned}$$

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$$\begin{aligned} \underline{UR_2 R_3} &= -212 & = \\ &+ 213(19-23) & = \\ &+ 6304 & = \\ &+ 47(50-53) & = \end{aligned}$$

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$$\begin{aligned} \underline{UR_1^2} &= 220^2 305 & = \\ &+ 220213 & = \\ &+ 304 & = \\ &+ 24751 & = \\ &+ 251^2 306 & = \end{aligned}$$

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$$\begin{aligned} \underline{UR_1 R_2} &= -1347 & = \\ &- 21351306 & = \end{aligned}$$

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$$\begin{aligned} \underline{UR_1 R_3} &= 220^2 305 & = \\ &+ 220213 & = \\ &+ 304 & = \\ &+ 47(20+51) & = \\ &+ 22051306 & = \end{aligned}$$

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$$\begin{aligned} \underline{UR_1 R_2} &= .5213 & = \\ &+ .547 & = \end{aligned}$$

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$$\underline{UR_1 R_3} = -304 =$$

$$\underline{UR_2^2} = 213^2 306 =$$

$$\begin{aligned} \underline{UR_2 R_1} &= -21320306 & = \\ &- 1347 & = \end{aligned}$$

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$$\underline{UR_2 R_2} = 0 = \underline{UR_2 R_3}$$



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$$\begin{aligned}
 \underline{UR_1^2} &= + 302 & = & \\
 &+ 2 \underline{20^2 305} & = & \\
 &+ 2 \underline{20 213} & = & \\
 &+ \underline{304} & = & \\
 &+ 2 \underline{20 47} & = & \\
 &+ 2 \underline{20^2 306} & = &
 \end{aligned}$$

$$\underline{UR_1 R_2} = +.5(213 + 47) =$$

$$\underline{UR_1 R_3} = -304 =$$

$$\begin{aligned}
 \underline{UR_2^2} &= + 303 & = & \\
 &+ .5(\underline{305} + \underline{306}) & = &
 \end{aligned}$$

$$\underline{UR_2 R_3} = -213 - 47 =$$

$$\underline{UR_3^2} = 2 \underline{304} =$$

$$\begin{aligned}
 0 &= u_1 + \underline{UR_1^2} R_1 + \underline{UR_1 R_2} R_2 + \underline{UR_1 R_3} R_3 \\
 0 &= u_2 + \underline{UR_2 R_1} R_1 + \underline{UR_2^2} R_2 + \underline{UR_2 R_3} R_3 \\
 0 &= u_3 + \underline{UR_3 R_1} R_1 + \underline{UR_3 R_2} R_2 + \underline{UR_3^2} R_3
 \end{aligned}$$

$$u_i = \underline{UR_i P_1} + \underline{UR_i P_2} + \underline{UR_i Q_1} + \underline{UR_i Q_2}$$



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$$\delta_1 = \frac{UP_1^2}{P_1} + \frac{UP_2}{P_2} P_2 + \frac{UP_1 Q_1}{Q_1} + \frac{UP_2 Q_2}{Q_2} + \frac{UP_1 R_1}{R_1} + \frac{UP_2 R_2}{R_2} + \frac{UP_1 R_3}{R_3} R_3$$

$$\delta_2 = \frac{UP_2^2}{P_2} + \frac{UP_1}{P_1} P_2 + \frac{UP_2 Q_1}{Q_1} + \frac{UP_1 Q_2}{Q_2} + \frac{UP_2 R_1}{R_1} + \frac{UP_1 R_2}{R_2} + \frac{UP_2 R_3}{R_3} R_3$$

$$\delta_3 = \frac{UP_1 Q_1}{Q_1} + \frac{UP_2 Q_2}{Q_2} + \frac{UP_1^2}{P_1} P_1 + \frac{UP_2^2}{P_2} P_2 + \frac{UP_1 R_1}{R_1} + \frac{UP_2 R_2}{R_2} + \frac{UP_1 R_3}{R_3} R_3$$

$$\delta_4 = \frac{UP_1 Q_1}{Q_1} + \frac{UP_2 Q_2}{Q_2} + \frac{UP_1^2}{P_1} P_1 + \frac{UP_2^2}{P_2} P_2 + \frac{UP_1 R_1}{R_1} + \frac{UP_2 R_2}{R_2} + \frac{UP_1 R_3}{R_3} R_3$$

Station 742.5Spar Properties:

Drawing No 7-0162-156. Referring to report # 7-0562-4, "Distribution of the Fin Roll Moment",

$$\int_0^{\frac{1}{2}} \frac{x^2}{I_y} dx = 6560 \text{ in}^{-1} = \frac{1}{I_c} \int_0^{\frac{1}{2}} x^2 dx = \frac{1}{I_c} \frac{x^3}{3}$$

$$\therefore I_c = \frac{114^3}{24 \times 6560} = 9.42 \text{ in}^4 \text{ (Equivalent spar of constant "I")}$$

Say section 5, sheet 3) of ref. report for spar properties; i.e., $A = 2.1855$; $I = 9.3179$;

$$\alpha_{sp} = \frac{A}{I} A \bar{y}^2 = \frac{2.1855}{9.3179 \times 1.125} \left(1.755^2 \times 2.1765 + 1.25^2 \times 1.989 + 1.125^2 \times 2.0515 \times 1.025 \right) = 4.75$$

Former Properties: $A = 4 \times 1.25 + 1.6 \times 1 = .5 + .16 = .66 \text{ in}^2$

Dwg. 7-0158-133



$$I = \frac{1}{12} (2.5 \times 4 - 1.6^3 \times 1.15) = \frac{1}{12} (10 - 4.715) = .440 \text{ in}^4$$

$$\alpha_{F0} = \frac{.66}{.44 \times 1} \left(.225^2 \times 1.9 \times 2 \times 1.25 + .4 \times 1 \times 1 \times .8 \right) = 3.855$$

These properties for $F0_1$, $F0_2$, and $F0_3$.

Change 2" width to 4" for $F0_3$.

$$A_{F0_3} = .66 + 1 \times 2 = .86$$

$$I_{F0_3} = \frac{1}{12} (1.25 \times 14 - 1.15 \times 3.6^3) = \frac{1}{12} (80 - 53.7) = 2.19$$

$$\alpha_{F0_3} = \frac{.86}{2.19 \times 1} \left(1.9 \times 2 \times 1.25 + 0.95^2 \times 1 \times 1.9 \right) = 1.16$$

Strut Properties:

$$L = 42.5^\circ, \delta = 9.2^\circ$$

$$A_{ST} = .3136 \text{ in}^2 \text{ (} a_{20} = 1.75; .058 \text{ wall)}; E_{ST} = 29 \times 10^6 \text{ p.s.i.}$$

Geometry:

$a_2 = 21.8$	$a_7 = 11.32$	$a_{17} = 7.24$	$a_{29} = -2.1''$
$a_3 = 8.8$	$a_{11} = 5.4$	$a_{20} = 2.4$	
$a_4 = 7.5$	$a_{13} = 23''$	$a_{25} = 54.4$	
$a_5 = 17.4$	$a_{15} - a_{14} = -.32$	$a_{26} = 47.9''$	
$a_6 = 19.2$	$a_{16} = 56.4$	$a_{27} = .9$	



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$$\theta_1 = 180 - 2 \sin^{-1} \frac{a_2}{\sqrt{a_1^2 + a_3^2}} = 180 - 2 \sin^{-1} \frac{21.8}{\sqrt{495.24 + 77.44}} = 180 - 2 \sin^{-1} 9.2730 = 43^\circ 58' = .76736 \text{ rad.} \\ (2 \times 68^\circ)$$

$$r_1 = \frac{a_2}{\sin \theta_1} = \frac{21.8}{.69424} = 31.401''$$

$$\theta_3 = 180 - 2 \sin^{-1} \frac{a_5}{\sqrt{a_4^2 + a_6^2}} = 180 - 2 \sin^{-1} \frac{17.4}{\sqrt{302.76 + 56.25}} = 180 - 2 \sin^{-1} 9.1230 = 46^\circ 38' = .81390 \text{ rad.} \\ (2 \times 66^\circ)$$

$$r_2 = \frac{a_5}{\sin \theta_3} = \frac{17.4}{.72697} = 23.935''$$

$$\theta_2 = \theta_3 + \sin^{-1} \frac{a_6}{r_2} = 46^\circ 38' + \sin^{-1} \frac{19.2}{23.935} = 46^\circ 38' + \sin^{-1} .80217 = 94^\circ 58.3' = 1.74483 \text{ rad.} \\ (53^\circ 20.3')$$

$$\beta = \tan^{-1} \frac{r_2 + a_2 - a_4}{r_1 + a_5 - a_3} = \tan^{-1} \frac{23.935 + 21.8 - 7.5}{31.401 + 17.4 - 8.8} = \tan^{-1} .95545 = 43^\circ 42.4' \\ 46.001$$

$$3 = a_{25} - a_{20} = 54.4 - 2.4 = 52.0$$

$$4 = \frac{1}{2 \cos 8^\circ} = \frac{1}{2 \cos 9.2^\circ} = \frac{1}{2 \times .98714} = .506573$$

$$5 = \frac{1}{2 \sin 8^\circ} = \frac{1}{2 \sin 9.2^\circ} = \frac{1}{2 \times .15958} = 3.127345$$

$$6 = \cos \beta = \cos 43^\circ 42.4' = .72289$$

$$7 = \sin \beta = \sin 43^\circ 42.4' = .69096$$

$$8 = a_{20} + a_{17} + a_2 + a_{13} = 2.4 + 7.24 + 21.8 + 23 = 54.44$$

$$9 = a_3 - a_{16} + a_5 = 8.8 - .32 = 8.48$$

$$10 = 11 - 7 = .40083 - .69096 = -.29013$$

$$11 = \frac{6.8 + 7.9}{2 a_{16}} = \frac{.72289 \times 54.44 + .69096 \times 8.48}{2 \times 56.4} = .40083$$

$$12 = \frac{a_{25}}{2 a_{16}} = \frac{54.4}{112.8} = .48227$$

$$13 = \frac{1}{2 a_{16}} = \frac{1}{112.8} = .00886524$$

$$14 = a_2 + a_{13} = 21.8 + 23 = 44.8$$

$$15 = 2 a_{16} - 9 = 112.8 - 8.48 = 104.32$$

$$16 = 9 = 8.48$$

$$17 = a_{25} - a_{20} - a_{17} = 54.4 - 2.4 - 7.24 = 44.76$$

$$18 = \frac{7.5 - 6.14}{2 a_{16}} = \frac{.69096 \times 104.32 - .72289 \times 44.8}{112.8} = .351910$$

$$19 = -\frac{6.14 + 7.16}{2 a_{16}} = -\frac{.72289 \times 44.8 + .69096 \times 8.48}{112.8} = -.339050$$

$$20 = \frac{17}{2 a_{16}} = \frac{44.76}{112.8} = .3968085$$

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$$21 = 11 = .40083$$

$$22 = 10 = -.29013$$

$$23 = 19 + 7 = -.33905 + .69096 = .35191$$

$$24 = 18 - 7 = .35191 - .69096 = -.33905$$

$$25 = \frac{L}{2A_{ST} E_{ST}} = \frac{42.5}{2 \times 3136 \times 2.9 \times 10^7} = 23.366 \text{ g.} \quad (\text{g} = 10^{-2})$$

$$26 = 2 \cdot 25 \cdot 5^2 = 2 \times 23.366 \text{ g} \cdot 3.127345^2 = 457.052 \text{ g.}$$

$$27 = 2 \cdot 25 \cdot 4^2 = 2 \times 23.366 \text{ g} \cdot 5.06513^2 = 11.94893 \text{ g.}$$

$$28 = \tan \alpha = \tan 4^\circ = .06993$$

$$29 = \sin \alpha = \sin 4^\circ = .06976$$

$$30 = \cos \alpha = \cos 4^\circ = .99756$$

$$31 = \frac{a_{27}}{30} + a_{20} = \frac{.90}{.99756} + 2.4 = 3.3022$$

$$32 = \frac{1}{2 \cdot 30 E_{sp} I_{sp}} = \frac{1}{2 \times 99756 \times 10^7 \times 9.3179} = .0537914 \text{ g.}$$

$$33 = \frac{a_{16}}{2 \cdot 30 A_{sp} E_{sp}} = \frac{56.4}{2 \times 99756 \times 2.1855 \times 10^7} = 12.9348 \text{ g.}$$

$$34 = \frac{a_{16} a_{20}}{2 \cdot 30 A_{sp} G_{sp}} = \frac{56.4 \times 4.75}{2 \times 99756 \times 2.1855 \times 4 \times 10^7} = 153.6008 \text{ g.}$$

$$35 = 32 \cdot 31 \cdot a_{16} = .0537914 \text{ g} \cdot 3.3022^2 \times 56.4 = 33.0425 \text{ g}$$

$$36 = 32 \cdot 31 \cdot a_{16}^2 = .0537914 \text{ g} \cdot 3.3022 \times 56.4^2 = 565.0338 \text{ g}$$

$$37 = 36 \cdot 28 = 565.0338 \text{ g} \cdot .06993 = 39.5128 \text{ g}$$

$$38 = 32 \cdot \frac{a_{16}^2}{3} = .0537914 \text{ g} \cdot 33333 \times 56.4^2 = 3216.8325 \text{ g}$$

$$39 = 38 \cdot 28^2 = 3216.8325 \text{ g} \cdot .06993^2 = 15.72095 \text{ g}$$

$$40 = 33 \cdot 30^2 = 12.9348 \text{ g} \cdot .99756^2 = 12.8718 \text{ g}$$

$$299 = 2 \cdot 38 \cdot 28 = 2 \times 3216.8325 \text{ g} \cdot .06993 = 449.9062 \text{ g}$$

$$41 = 2 \cdot 33 \cdot 30 \cdot 29 = 2 \times 12.9348 \text{ g} \cdot .99756 \times .06976 = 1.8003 \text{ g}$$

$$42 = 23 \cdot 29^2 = 12.9348 \text{ g} \cdot .06976^2 = .062947 \text{ g}$$

$$43 = 34 \cdot 29^2 = 153.6008 \text{ g} \cdot .06976^2 = .747492 \text{ g}$$

$$44 = 2 \cdot 34 \cdot 29 \cdot 30 = 2 \times 153.6008 \text{ g} \cdot .06976 \times .99756 = 21.3741 \text{ g}$$

$$45 = 34 \cdot 30^2 = 153.6008 \text{ g} \cdot .99756^2 = 152.8521 \text{ g}$$

$$46 = 35 + 27 + 39 + 40 + 43 = \text{g}(33.0825 + 39.5128 + 15.73095 + 12.8718 + .7475) = 101.9456 \text{ g}$$

$$47 = 36 + 299 - 41 + 44 = \text{g}(565.0338 + 449.9062 - 1.8003 + 21.3741) = 1034.5178 \text{ g}$$

$$48 = 38 + 42 + 45 = \text{g}(3216.8325 + .0629 + 152.8521) = 3369.7475 \text{ g}$$

$$49 = 10 + 18 = -.29013 + .35191 = +.06178$$

$$50 = 11 + 19 = .40083 - .33905 = +.06178$$



AERO AIRCRAFT LIMITED

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$$51 = 20 - 12 = .376085 - .48227 = -.08546$$

$$52 = 21 + 24 = .40083 - .33905 = +.06178$$

$$53 = 22 + 23 = -.29013 + .35191 = +.06178$$

$$55 = \tan \Omega = \frac{a_{14} - a_{15}}{a_{13}} = \frac{.32}{.23} = .013913 \quad (\Omega = 0^\circ 44')$$

$$56 = \sin \Omega = .013913$$

$$57 = \cos \Omega = .999990$$

$$82 = \frac{a_3^2}{6 E_{FO} I_{FO} 57} = \frac{23^2}{6 \times 10^7 \times 440 \times .99999} = 4,609.16 \text{ g}$$

$$83 = \frac{a_3}{2 A_{FO} E_{FO} 57} = \frac{23}{2 \times .66 \times 10^7 \times .99999} = 17.4260 \text{ g}$$

$$84 = \frac{a_3 \sqrt{F_{FO}}}{2 A_{FO} G_{FO} 57} = \frac{.23 \times 3.855}{2 \times .66 \times 4 \times 10^7 \times .99999} = 167.9431 \text{ g}$$

$$85 = 82 + 83 \cdot 56^2 + 84 \cdot 57^2 = \text{g} (4609.16 + 17.4260 \times .013913^2 + 167.9431 \times .99999^2) = 4,777.07 \text{ g}$$

$$58 = 2 \cdot 55 \cdot 82 + 2 \cdot 56 \cdot 87 (84 - 83) = \text{g} [2 \times .013913 \times 4609.16 + 2 \times .013913 \times .99999 (167.9431 - 17.4260)] = 132.4424 \text{ g}$$

$$59 = 82 \cdot 55^2 + 83 \cdot 57^2 + 84 \cdot 56^2 = \text{g} (4609.16 \times .013913^2 + 17.4260 \times .99999^2 + 167.9431 \times .013913^2) = 18.3472 \text{ g}$$

$$86 = r_1 + a_{15} - a_{14} = 31.401 - .32 = 31.081$$

$$87 = \frac{r_1}{2 E_{FO} I_{FO}} = \frac{31.401}{2 \times 10^7 \times .44} = 35.6829 \text{ g}$$

$$88 = \frac{r_1}{2 A_{FO} E_{FO}} = \frac{31.401}{2 \times .66 \times 10^7} = 23.7886 \text{ g}$$

$$89 = \frac{r_1 \sqrt{F_{FO}}}{2 A_{FO} G_{FO}} = \frac{31.401 \times 3.855}{2 \times .66 \times 4 \times 10^7} = 229.2626 \text{ g}$$

$$90 = .5 (B_1 - .5 \sin 2B_1) = .5 (.76736 - .5 \times .99995) = .13344$$

$$91 = .5 (B_1 + .5 \sin 2B_1) = .5 (.76736 + .5 \times .99995) = .63352$$

$$92 = 1 - \cos B_1 = 1 - .71974 = .28026$$

$$93 = \sin B_1 = .69424$$

$$94 = .5 (1 - \cos 2B_1) = .5 (1 - .03606) = .48197$$

$$95 = 87 B_1 a_3^2 = 35.6829 \text{ g} \cdot 76736 \times 23^2 = 14,444.87 \text{ g}$$

$$96 = 2 \cdot 87 B_1 a_3 a_6 = 2 \times 35.6829 \text{ g} \cdot 76736 \times 23 \times 31.081 = 39,148.20 \text{ g}$$

$$97 = 87 B_1 a_6^2 = 35.6829 \text{ g} \cdot 76736 \times 31.081^2 = 26,451.41 \text{ g}$$

$$98 = 87 a_0 a_1^2 = 35.6829 \text{ g} \cdot 13344 \times 31.401^2 = 4,709.05 \text{ g}$$

$$99 = 87 a_1 a_1^2 = 35.6829 \text{ g} \cdot 63352 \times 31.401^2 = 22,289.86 \text{ g}$$

$$100 = 2 \cdot 87 \cdot 92 r_1 a_3 = 2 \times 35.6829 \text{ g} \cdot 28026 \times 31.401 \times 23 = 14,445.17 \text{ g}$$

$$101 = 2 \cdot 87 \cdot 92 r_1 a_6 = 2 \times 35.6829 \text{ g} \cdot 28026 \times 31.401 \times 31.081 = 19,520.44 \text{ g}$$

$$102 = 2 \cdot 87 \cdot 92 r_1 a_3 = 2 \times 35.6829 \text{ g} \cdot 69424 \times 31.401 \times 23 = 35,782.153 \text{ g}$$

$$103 = 2 \cdot 87 \cdot 92 r_1 a_6 = 2 \times 35.6829 \text{ g} \cdot 69424 \times 31.401 \times 31.081 = 48,354.64 \text{ g}$$

AVRO AIRCRAFT LIMITED

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$$104 = 87 \cdot 94 \cdot l^2 = 35.6829 \text{ q} \cdot 4.8197 \times 31.401^2 = 16,957.70 \text{ q}$$

$$105 = 88 \cdot 90 = 23.7886 \text{ q} \cdot 1.3384 = 3.18387 \text{ q}$$

$$106 = 88 \cdot 91 = 23.7886 \text{ q} \cdot 6.3352 = 15.07055 \text{ q}$$

$$107 = 88 \cdot 94 = 23.7886 \text{ q} \cdot 4.8197 = 11.4654 \text{ q}$$

$$108 = 89 \cdot 90 = 229.2626 \text{ q} \cdot 1.3384 = 30.6845 \text{ q}$$

$$109 = 89 \cdot 91 = 229.2626 \text{ q} \cdot 6.3352 = 145.2424 \text{ q}$$

$$110 = 89 \cdot 94 = 229.2626 \text{ q} \cdot 4.8197 = 110.4977 \text{ q}$$

$$111 = 95 + 98 + 180 + 105 + 109 = \text{q}(14,484.87 + 4,709.05 + 14,445.17 + 3.18 + 145.24) = 33,747.51 \text{ q}$$

$$112 = -96 - 101 + 102 + 104 + 107 - 110 = \text{q}(-39,148.20 - 19,520.44 + 35,782.53 + 16,957.70 + 11.47 - 110.50) = -6,027.44 \text{ q}$$

$$113 = 97 + 99 - 103 + 106 + 108 = \text{q}(26,481.41 + 23,289.86 - 48,354.64 + 15.07 + 30.68) = 432.38 \text{ q}$$

$$114 = \cos \theta_3 = .68666$$

$$115 = \sin \theta_3 = .72697$$

$$116 = \frac{F_2}{2E_{F0} I_{F02}} = \frac{23.935}{2 \times 10^7 \times .44} = 27.1984 \text{ q}$$

$$117 = \frac{F_2}{2A_{F02} E_{F0}} = \frac{23.935}{2 \times .66 \times 10^7} = 18.1326 \text{ q}$$

$$118 = \frac{F_2 G_{F02}}{2A_{F02} G_{F0}} = \frac{23.935 \times 3.855}{2 \times .66 \times .4 \times 10^7} = 174.7529 \text{ q}$$

$$119 = .5(\theta_2 - .5 \sin 2\theta_2) = .5(1.74483 - .5[-.34109]) = .957688$$

$$120 = .5(\theta_2 + .5 \sin 2\theta_2) = .5(1.74483 + .5[-.34109]) = .747143$$

$$121 = 1 - \cos \theta_2 = 1 - (-.17318) = 1.17318$$

$$122 = \sin \theta_2 = .98489$$

$$123 = .5(1 - \cos 2\theta_2) = .5(1 - [-.94075]) = .97037$$

$$124 = 116 \theta_2 = 27.1988 \text{ q} \cdot 1.74483 = 47.4573 \text{ q}$$

$$125 = 2 \cdot 116 \theta_2^2 l^2 = 2 \times 27.1988 \text{ q} \cdot 1.74483 \times 23.935^2 = 2,271.74 \text{ q}$$

$$126 = 116 \theta_2 l^2 = 27.1988 \text{ q} \cdot 1.74483 \times 23.935^2 = 27,187.52 \text{ q}$$

$$127 = 116 \cdot 118 l^2 = 27.1988 \text{ q} \cdot 95.7688 \times 23.935^2 = 14,922.46 \text{ q}$$

$$128 = 116 \cdot 120 l^2 = 27.1988 \text{ q} \cdot 174.7143 \times 23.935^2 = 12,265.07 \text{ q}$$

$$129 = 2 \cdot 116 \cdot 121 l^2 = 2 \times 27.1988 \text{ q} \cdot 1.17318 \times 23.935^2 = 1,527.49 \text{ q}$$

$$130 = 2 \cdot 116 \cdot 121 l^2 = 2 \times 27.1988 \text{ q} \cdot 1.17318 \times 23.935^2 = 36,560.42 \text{ q}$$

$$131 = 2 \cdot 116 \cdot 122 l^2 = 2 \times 27.1988 \text{ q} \cdot 98.489 \times 23.935^2 = 1,282.33 \text{ q}$$

$$132 = 2 \cdot 116 \cdot 122 l^2 = 2 \times 27.1988 \text{ q} \cdot 98.489 \times 23.935^2 = 30,692.60 \text{ q}$$

$$133 = 116 \cdot 123 l^2 = 27.1988 \text{ q} \cdot 97.003 \times 23.935^2 = 15,114.77 \text{ q}$$

$$134 = 117 \cdot 119 = 18.1326 \text{ q} \cdot 95.7688 = 17.3654 \text{ q}$$



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$$\begin{aligned}
 135 &= 117 \ 120 = 18,1326 \text{ g} \cdot 787143 = 14,2729 \text{ g} \\
 136 &= 117 \ 123 = 18,1326 \text{ g} \cdot 97003 = 17,5492 \text{ g} \\
 137 &= 118 \ 119 = 174,7529 \text{ g} \cdot 957668 = 167,3588 \text{ g} \\
 138 &= 118 \ 120 = 174,7529 \text{ g} \cdot 787143 = 137,5555 \text{ g} \\
 139 &= 118 \ 123 = 174,7529 \text{ g} \cdot 97003 = 169,5755 \text{ g} \\
 140 &= 126 + 138 - 132 + 135 + 137 = \text{g}(27,18752 + 12,26507 - 30,69260 + 14,27 + 167,36) = 8,941,62 \text{ g} \\
 141 &= 127 + 134 + 138 = \text{g}(14,92246 + 17,37 + 137,56) = 15,077,39 \text{ g} \\
 142 &= -125 + 131 = \text{g}(-2271,78 + 1,282,33) = -989,45 \text{ g} \\
 143 &= 130 - 123 - 126 + 139 = \text{g}(36,56042 - 15,114,77 - 17,54 + 169,52) = 21,597,58 \text{ g} \\
 144 &= 124 \ 14^2 = 47,4573 \text{ g} \cdot 44,8^2 = 95,248,70 \text{ g} \\
 145 &= 124 \ 9^2 = 47,4573 \text{ g} \cdot 8,48^2 = 3412,67 \text{ g} \\
 146 &= 2 \ 128 \ 9 \ 14 = 2 \times 47,4573 \text{ g} \cdot 8,48 \times 44,8 = 36058,44 \text{ g} \\
 147 &= 140 \ 115^2 = 8941,62 \text{ g} \cdot 72697^2 = 4,725,51 \text{ g} \\
 148 &= 141 \ 114^2 = 15,077,39 \text{ g} \cdot 68666^2 = 7,109,02 \text{ g} \\
 149 &= 143 \ 114 \ 115 = 21,597,58 \text{ g} \cdot 68666 \times 72697 = 10,781,10 \text{ g} \\
 150 &= 147 + 148 + 149 = 22,615,63 \text{ g} \\
 151 &= 140 \ 141^2 = 8941,62 \text{ g} \cdot 68666^2 = 4,215,99 \text{ g} \\
 152 &= 141 \ 115^2 = 15,077,39 \text{ g} \cdot 72697^2 = 7,968,17 \text{ g} \\
 153 &= 143 \ 114 \ 115 = 21,597,58 \text{ g} \cdot 68666 \times 72697 = 10,781,10 \text{ g} \\
 154 &= 151 + 152 - 153 = 1403,06 \text{ g} \\
 155 &= 2 \ 140 \ 114 \ 115 = 2 \times 8941,62 \text{ g} \cdot 68666 \times 72697 = 8926,97 \text{ g} \\
 156 &= 2 \ 141 \ 114 \ 115 = 2 \times 15,077,39 \text{ g} \cdot 68666 \times 72697 = 15,052,69 \text{ g} \\
 157 &= 143 \ (114^2 - 115^2) = 21,597,58 \text{ g} \cdot (68666^2 - 72697^2) = -1,230,69 \text{ g} \\
 158 &= 155 - 156 + 157 = -7356,41 \text{ g} \\
 159 &= 142 \ 14 \ 115 = -989,45 \text{ g} \times 44,8 \times 72697 = -32,224,66 \text{ g} \\
 160 &= 129 \ 14 \ 114 = 1527,49 \text{ g} \cdot 44,8 \times 68666 = 46,989,21 \text{ g} \\
 161 &= 159 - 160 = -79,213,87 \text{ g} \\
 162 &= 142 \ 14 \ 114 = -989,45 \text{ g} \cdot 44,8 \times 68666 = -30,437,83 \text{ g} \\
 163 &= 129 \ 14 \ 115 = 1527,49 \text{ g} \cdot 44,8 \times 72697 = 49,747,69 \text{ g} \\
 164 &= 162 + 163 = 19,309,86 \text{ g} \\
 165 &= 142 \ 9 \ 115 = -989,45 \text{ g} \cdot 8,48 \times 72697 = -6099,67 \text{ g} \\
 166 &= 129 \ 9 \ 114 = 1527,49 \text{ g} \cdot 8,48 \times 68666 = 8894,39 \text{ g} \\
 167 &= -165 + 166 = 14,994,06 \text{ g} \\
 168 &= 142 \ 9 \ 114 = -989,45 \text{ g} \cdot 8,48 \times 68666 = -5,761,45 \text{ g} \\
 169 &= 129 \ 9 \ 115 = 1527,49 \text{ g} \cdot 8,48 \times 72697 = 9,416,53 \text{ g} \\
 170 &= -168 - 169 = -3,655,08 \text{ g} \\
 171 &= 144 + 154 + 164 = \text{g}(95,24870 + 1,403,06 + 19,309,86) = 115,961,62 \text{ g}
 \end{aligned}$$



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$$172 = 145 + 150 + 167 = g(3412.67 + 22,615.63 + 14,994.06) = 41,022.36 g$$

$$173 = 7 \times 150 + 6 \times 154 - 67 \times 158 = g(1477426 + 22,615.63 + 5,22570 \times 1403.06 + 499288 \times 7356.44) = 15,2044.93 g$$

$$174 = -146 + 158 + 161 + 170 = g(-36,058.44 - 7156.41 - 79,213.87 - 3,655.08) = -126,283.80 g$$

$$175 = 2 \times 154 - 7(158 + 161) + 6 \times 164 = g(2028.52 - 11,44578 \times 1403.06 + 6,909628 \times 570.28 + 72289 \times 19,107.86) = 75,804.02 g$$

$$176 = 6(158 + 170) - 7(2 \times 150 + 167) = g(-72289 \times 11,011.49 - 69096 \times 60,226.32) = -49,573.39 g$$

$$177 = a_6 - a_3 = 56.4 - 11.32 = 45.08$$

$$178 = 14 + a_{39} = 44.8 - 2.1 = 42.7$$

$$179 = a_5 + a_6 = 17.4 + 19.2 = 36.6$$

$$180 = 6 a_{39} + 7 \times 179 = -72289 \times 2.1 + 69096 \times 36.6 = 23,771.07$$

$$181 = \frac{1}{2 E_{FO} I_{FO_3}} = \frac{1}{2 \times 10^3 \times 2.19} = .22831 g$$

$$182 = \frac{a_3}{2 A_{a_3} E_{FO}} = \frac{4.32}{2 \times .86 \times 10^3} = 6.58139 g$$

$$183 = \frac{a_3 \times a_{FO_3}}{2 A_{FO_3} G_{FO}} = \frac{11.32 \times 1.16}{2 \times .86 \times .44 \times 10^3} = 19.08603 g$$

$$184 = 181 a_3 = .22831 g \times 11.32 = 2.584470 g$$

$$185 = 184 a_3 = 2.584470 g \times 11.32 = 29.2562 g$$

$$186 = .185 \frac{a_3}{3} + 183 = 29.2562 \times 3.71333 g + 19.08603 g = 129.4793 g$$

$$187 = 184 \times 128^3 = 2.58447 g \times 1823.29 = 4712.24 g$$

$$188 = 184 \times 180^3 = 2.58447 g \times 565.0638 = 1460.39 g$$

$$189 = 184 \times 177^3 = 2.58447 g \times 2032.2064 = 5252.1765 g$$

$$190 = 2 \times 184 \times 178 \times 180 = 5.16894 \times 42.7 \times 23,771.07 = 5246.61 g$$

$$191 = 2 \times 184 \times 179 \times 177 = 5.16894 \times 42.7 \times 45.08 = 9,949.79 g$$

$$192 = 2 \times 184 \times 180 \times 177 = 5.16894 \times 23,771.07 \times 45.08 = 5,539.02 g$$

$$193 = 185 \times 178^3 = 29.2562 g \times 42.7 = 1249.24 g$$

$$194 = 185 \times 180 = 29.2562 g \times 23,771.07 = 695.45 g$$

$$195 = 185 \times 177 = 29.2562 g \times 45.08 = 1318.87 g$$

$$196 = 185 \times 178 = 29.2562 g \times 69096 \times 42.7 = 863.17 g$$

$$197 = 185 \times 180 = 29.2562 g \times 69096 \times 23,771.07 = 480.53 g$$

$$198 = 185 \times 177 = 20.21446 g \times 45.08 = 911.29 g$$

$$199 = 2 \times 186 \times 7 = 2 \times 129.4793 g \times 69096 = 178.93 g$$

$$200 = 186 \times 7^2 = 129.4793 g \times 47426 = 61.8168 g$$

$$201 = 2 \times 182 \times 6 = 2 \times 6.58139 g \times 72289 = 9.5152 g$$

$$202 = 182 \times 6^3 = 6.58139 g \times 522570 = 3.4392 g$$

$$203 = 187 + 182 = 4712.24 g + 6.58 g = 4718.82 g$$

$$204 = 188 + 197 + 200 + 202 = g(1460.39 + 480.53 + 61.82 + 3.44) = 2,006.18 g$$



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TECHNICAL DEPARTMENT (Aircraft)

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$$\begin{aligned}
 205 &= 189 + 195 + 186 = q(5252.1765 + 1318.87 + 129.48) = 6700.53 \text{ g} \\
 206 &= 190 + 196 + 201 = q(5246.61 + 863.17 + 9.52) = 6119.30 \text{ g} \\
 207 &= -181 - 193 = q(-9,949.79 - 1249.24) = -11,199.03 \text{ g} \\
 208 &= -192 - 194 - 198 - 199 = q(-5539.02 - 695.45 - 911.29 - 178.93) = -7224.69 \text{ g} \\
 209 &= 85 + 111 + 171 + 203 = q(4777.07 + 33,787.51 + 115,961.62 + 4718.82) = 159,245.02 \text{ g} \\
 210 &= 173 + 204 = q(15,204.93 + 2006.16) = 17,211.11 \text{ g} \\
 211 &= 59 + 113 + 172 + 205 = q(18.35 + 932.36 + 41,022.36 + 6700.53) = 48,173.62 \text{ g} \\
 212 &= 125 + 206 = q(75,804.02 + 6,119.30) = 81,923.32 \text{ g} \\
 213 &= 58 + 112 + 174 + 207 = q(132.44 - 6027.44 - 126,283.80 - 11,199.03) = -143,377.83 \text{ g} \\
 214 &= 126 + 208 = q(-49,573.39 - 7324.69) = -56,898.08 \text{ g} \\
 300 &= 208 + 46 = q(159,245.02 + 101.95) = 159,346.97 \text{ g} \\
 301 &= 2 \ 210 = 34,422.22 \text{ g} \\
 302 &= 2 \ 26 = 914,1040 \text{ g} \\
 303 &= 2 \ 27 = 23,9786 \text{ g} \\
 304 &= 2 \ 200 = 318,693.94 \text{ g} \\
 305 &= 2 \ 211 = 96,347.24 \text{ g} \\
 306 &= 2 \ 48 = 8739.50 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UP}_1^2 &= +301 = 34,422.22 \\
 + 2 \ 18 \ 214 &= 2 \times 35191(-56,898.08 \text{ g}) = -40,046.01 \\
 + 305 (18^2 + 24^2) &= 96,347.24 \text{ g} (123841 + 114965) = 23,007.34 \\
 - 2 \ 6 \ 24 \ 213 &= -2 \times 72289(-33905)(-143,377.83) \text{ g} = -70,282.62 \\
 + 6^2 \ 304 &= .522570 \times 318,693.94 \text{ g} = 166,539.89 \\
 - 2 \ 6 \ 47 \ 52 &= -1.44578 \times 1034.5178 \text{ g} \times 0.6178 = -92.40 \\
 + 306 (44^2 + 52^2) &= 6739.5 \text{ g} (.00351677 \times 2) = 51.45 \\
 &= 113,599.87 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UP}_2 &= +214(19+24) = -56,898.08 \text{ g} (-33905 \times 2) = 38,582.59 \\
 + 305 (1819 + 2324) &= 96,347.24 \text{ g} (-119315 \times 2) = -22,991.34 \\
 - 6 \ 213 (37 + 24) &= -72289 (-143,377.83)(.01286) = 1232.89 \\
 + 6^2 \ 304 &= 166,539.89 \\
 - 6 \ 212 &= -72289 \times 81,923.32 = -59,221.55 \\
 - 6 \ 47 (52 + 53) &= -2 \ 6 \ 47 \ 52 = -92.40 \\
 + 306 (44^2 + 52^2) &= 306 (44^2 + 52^2) = 51.45 \\
 &= 124,201.53 \text{ g}
 \end{aligned}$$



AERO AIRCRAFT LIMITED

TECHNICAL DEPARTMENT (Aircraft)

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energy of struts and
spar.

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$$\begin{aligned}
 205 &= 189 + 195 + 186 = \text{Same} \\
 206 &= 190 + 196 + 201 = \\
 207 &= -191 - 193 = \\
 208 &= -192 - 194 - 198 - 199 = \\
 209 &= 85 + 111 + 171 + 203 = \\
 210 &= 173 + 204 = \\
 211 &= 59 + 113 + 122 + 205 = \\
 212 &= 175 + 206 = \text{Same} \\
 213 &= 58 + 112 + 174 + 207 = \text{Same} \\
 214 &= 126 + 208 = \text{Same} \\
 300 &= 209 + 46 = 209 = 158,245.02 \text{ g} \\
 301 &= 2 \cdot 210 = \text{Same} \\
 302 &= 2 \cdot 26 = 0 \\
 303 &= 2 \cdot 27 = 0 \\
 304 &= 2 \cdot 300 = 318,490.04 \text{ g} \\
 305 &= 2 \cdot 211 = \text{Same} \\
 306 &= 2 \cdot 48 = 0
 \end{aligned}$$

$$\begin{aligned}
 \underline{UP_1^2} &= +301 &= &\checkmark &= &34,422.22 \\
 &+ 2 \cdot 18 \cdot 214 &= &\checkmark &= &-40,046.01 \\
 &+ 305 (18^2 + 214^2) &= &\checkmark &= &23,007.34 \\
 &- 2 \cdot 6 \cdot 24 \cdot 213 &= &\checkmark &= &-70,282.62 \\
 &+ 6^2 \cdot 304 &= &.522570 \times 318,490.04 &= &166,433.34 \\
 &- 2 \cdot 6 \cdot 47 \cdot 52 &= & &= &- \\
 &+ 306 (47^2 + 52^2) &= & &= &- \\
 &&&&&= &\underline{113,534.27 \text{ g}}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UP_2} &= +214 (19 + 24) &= &\checkmark &= &38,582.59 \\
 &+ 205 (18 \cdot 19 + 23 \cdot 24) &= &\checkmark &= &-22,991.34 \\
 &- 6 \cdot 213 (23 + 24) &= &\checkmark &= &1332.89 \\
 &+ 6^2 \cdot 304 &= & &= &166,433.34 \\
 &- 6 \cdot 47 \cdot 52 &= &\checkmark &= &-59,221.55 \\
 &- 6 \cdot 47 (52 + 53) &= & &= &- \\
 &+ 306 (47 \cdot 50 + 52 \cdot 53) &= & &= &- \\
 &&&&&= &\underline{124,125.93 \text{ g}}
 \end{aligned}$$

$$\begin{aligned}
 \infty &= A_{ST} = A_{SP} = I_{SP} \\
 0 &= 26 = 27 = 302 = \underline{303} \\
 &= 32 = 33 = \underline{34} \\
 &= 46 = 47 = \underline{48} \\
 &= \underline{306}
 \end{aligned}$$



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$$\begin{aligned}
 \underline{UPQ}_1 &= +20 \ 214 &= .3968085 (-56,899.05) &= -22,577.64 \\
 &+20 \ 305 (18+24) &= .3968085 \times 96,347.24 \times .01286 &= 491.66 \\
 &+213 (24-6 \ 20) &= -143,377.83 (-37905 - 72289 \times .3968085) &= 89,740.04 \\
 &-6 \ 304 &= -.72289 \times 318,693.84 &= -230,380.66 \\
 &-47 (6 \ 51-52) &= -1034.5178 (-72289 \times .08546 - .06178) &= 127.83 \\
 &+206 \ 51 (49+52) &= 6739.5 (-.08546 \times .06178 \times 2) &= -71.17 \\
 &&&= \underline{-162,667.94} \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UPQ}_2 &= +6 \ 13 \ 47 &= .72289 \times .00886524 \times 1034.5178 &= 6.62805 \\
 &-13 \ 306 (49+52) &= -.07486524 \times 6739.5 \times .12356 &= -7.382378 \\
 &&&= \underline{-.754298} \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UPR}_1 &= 20 \ 214 &= &= -22,577.64 \\
 &+20 \ 305 (18+24) &= &= 491.66 \\
 &+213 (24-6 \ 20) &= &= 89,740.04 \\
 &-6 \ 304 &= &= -230,380.66 \\
 &-47 (6 \ 20-52) &= -1034.5178 (72289 \times .3968085 - .06178) &= -232.84 \\
 &+20 \ 306 (49+52) &= .3968085 \times 6739.5 \times .12356 &= 330.44 \\
 &&&= \underline{-162,629.00} \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UPR}_2 &= -5 \ 214 &= &= 28,449.04 \\
 &-5 \ 305 (18-24) &= -48,123.62 (35191 + .37905) &= -33,286.04 \\
 &-5 \ 6 \ 213 &= -.361445 (-143,377.83) &= 51,823.20 \\
 &-5 \ 6 \ 47 &= -.361445 \times 1034.5178 &= -373.92 \\
 &-5 \ 306 (49-52) &= &= 0.0 \\
 &&&= \underline{46,612.28} \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 \underline{UPR}_3 &= +212 &= &= 81,923.32 \\
 &+213 (18-24) &= -143,377.83 \times .69086 &= -99,068.35 \\
 &+6 \ 304 &= &= 230,380.66 \\
 &+47 (49-52) &= &= 0.0 \\
 &&&= \underline{213,235.63} \text{ g}
 \end{aligned}$$



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$$\begin{array}{rcl}
 \underline{UPR_0} = +20 \ 214 & = & \checkmark \\
 + 30 \ 305 (18+24) & = & \checkmark \\
 + 213 (24-6 \ 20) & = & \checkmark \\
 - 6 \ 304 & = & -72289 \times 318, 490.04 \\
 - 47 (6 \ 51-52) & = & \\
 + 306 \ 51 (49+52) & = & \\
 & = & -22,577.64 \\
 & = & 491.66 \\
 & = & 89,740.04 \\
 & = & -230,233.27 \\
 & = & \underline{\hspace{2cm}} \\
 & = & -162,579.21 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UPR_2} = +6 \ 13 \ 47 & = & \\
 -13 \ 306 (49+52) & = & \\
 & = & \underline{\hspace{2cm}} \\
 & = & \underline{\hspace{2cm}}
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UPR_1} = 20 \ 214 & = & \checkmark \\
 + 20 \ 305 (18+24) & = & \checkmark \\
 + 213 (24-6 \ 20) & = & \checkmark \\
 - 6 \ 304 & = & \\
 - 47 (6 \ 20-52) & = & \\
 + 20 \ 306 (49+52) & = & \\
 & = & -162,579.21 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UPR_2} = -5 \ 214 & = & \checkmark \\
 -5 \ 305 (18-24) & = & \checkmark \\
 -5 \ 6 \ 213 & = & \checkmark \\
 -5 \ 6 \ 47 & = & \\
 -5 \ 306 (49-52) & = & \\
 & = & 28,449.04 \\
 & = & -33,246.04 \\
 & = & 51,423.20 \\
 & = & \underline{\hspace{2cm}} \\
 & = & 46,986.20 \ g
 \end{array}$$

$$\begin{array}{rcl}
 \underline{UPR_3} = +212 & = & \checkmark \\
 + 213 (18-24) & = & \checkmark \\
 + 6 \ 304 & = & \\
 + 47 (49-52) & = & \\
 & = & 81,923.32 \\
 & = & -99,068.35 \\
 & = & 230,233.27 \\
 & = & \underline{\hspace{2cm}} \\
 & = & 213,088.24 \ g
 \end{array}$$



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$$\begin{aligned}
 UP_2^2 &= + 301 &= &= 34,422.22 \\
 + 2 \ 23 \ 214 &= 2 \times .35191 (-56,844.08) &= &= -40,046.01 \\
 - 2 \ 6 \ 212 &= -1.44578 \times 81,923.32 &= &= -118,443.10 \\
 + 305 (19^2 + 23^2) &= 96,347.24 (.33905^2 + .35191^2) &= &= 23,007.34 \\
 - 2 \ 6 \ 23 \ 213 &= -1.44578 \times .35191 (-143,377.83) &= &= 72,948.35 \\
 + 6^2 \ 304 &= &= &= 166,539.89 \\
 - 2 \ 6 \ 47 \ 53 &= &= &= - 92.40 \\
 + 306 (50^2 + 53^2) &= &= &= 51.45 \\
 & & &= 138,387.74 \checkmark
 \end{aligned}$$

$$\begin{aligned}
 UP_2 Q_1 &= + 20 \ 214 &= &= -22,577.64 \\
 + 212 &= &= &= 81,923.32 \\
 + 20 \ 305 (19 + 23) &= &= &= 491.66 \\
 + 213 (23 - 6 \ 20) &= -143,377.83 (.35191 - .28688) &= &= -9,328.15 \\
 - 6 \ 304 &= &= &= -230,380.66 \\
 - 47 (6 \ 51 - 53) &= &= &= 127.83 \\
 + 51 \ 306 (50 + 53) &= &= &= - 71.17 \\
 & & &= -179,814.81 \checkmark
 \end{aligned}$$

$$\begin{aligned}
 UP_2 R_1 &= + 6 \ 13 \ 47 &= &= 6.629805 \\
 - 13 \ 306 (50 + 53) &= &= &= - 7.382378 \\
 & & &= - .752573 \checkmark
 \end{aligned}$$

$$\begin{aligned}
 UP_2 R_2 &= + 20 \ 214 &= &= -22,577.64 \\
 + 212 &= &= &= 81,923.32 \\
 + 20 \ 305 (19 + 23) &= &= &= 491.66 \\
 + 213 (23 - 6 \ 20) &= &= &= - 9,328.15 \\
 - 6 \ 304 &= &= &= -230,380.66 \\
 - 47 (6 \ 20 - 53) &= &= &= - 232.84 \\
 + 20 \ 206 (50 + 53) &= &= &= 320.44 \\
 & & &= -179,773.87 \checkmark
 \end{aligned}$$

$$\begin{aligned}
 UP_2 R_3 &= + .5 \ 214 &= &= -28,449.04 \\
 - .5 \ 305 (19 - 23) &= &= &= 33,286.04 \\
 - .5 \ 6 \ 213 &= &= &= 51,823.20 \\
 - .5 \ 6 \ 47 &= &= &= - 373.92 \\
 - .5 \ 306 (50 - 53) &= &= &= 0.0 \\
 & & &= 56,286.28 \checkmark
 \end{aligned}$$

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$$\begin{aligned}
 U_{P_2 R_3} &= -212 &= &= -81,923.32 \\
 &+ 213(19-23) &= &= 99,088.35 \\
 &+ 6 \underline{304} &= &= 230,380.66 \\
 &+ 47(50-53) &= &= 0.0 \\
 &&&= 247,525.69 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 U_{Q_1} &= 2 \underline{20^2 305} &= &= 2 \times .3968085^2 \times 96,347.24 &= &= 30,341.09 \\
 &+ 2 \underline{20 213} &= &= 2 \times .3968085 (-143,377.83) &= &= -113,797.08 \\
 &+ \underline{304} &= &= &= &= 318,693.94 \\
 &+ 2 \underline{47 51} &= &= 2 \times 1034.5178 (-0.08546) &= &= -176.82 \\
 &+ 2 \underline{51^2 306} &= &= 2 \times .08546^2 \times 6739.50 &= &= 98.44 \\
 &&&&&= &= 235,164.57 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 U_{Q_1 Q_2} &= -13 \underline{47} &= &= -.00886524 \times 1034.5178 &= &= -9.171249 \\
 &- 2 \underline{13 51 306} &= &= -2 \times .00886524 (-0.08546)(6739.5) &= &= 10,212.007 \\
 &&&&&= &= 1.040758 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 U_{Q_1 R_1} &= 2 \underline{20^2 305} &= &= &= &= 30,341.09 \\
 &+ 2 \underline{20 213} &= &= &= &= -113,747.08 \\
 &+ \underline{304} &= &= &= &= 318,693.94 \\
 &+ \underline{47(29+51)} &= &= 1034.5178(.3968085 - .08546) &= &= 322.10 \\
 &+ 2 \underline{20 51 306} &= &= 2 \times .3968085 (-0.08546)(6739.5) &= &= -457.09 \\
 &&&&&= &= 235,112.96 \text{ g}
 \end{aligned}$$

$$\begin{aligned}
 U_{Q_1 R_2} &= .5 \underline{213} &= &= &= &= -71,688.92 \\
 &+ .5 \underline{47} &= &= &= &= 517.26 \\
 &&&&&= &= -71,171.66 \text{ g}
 \end{aligned}$$

$$U_{Q_1 R_3} = -304 = -318,693.94 \text{ g}$$

$$U_{Q_2} = 2 \underline{13^2 306} = 2 \times .00886524^2 \times 6739.5 = 1.057348 \text{ g}$$

$$\begin{aligned}
 U_{Q_2 R_1} &= -2 \underline{13 20 306} &= &= -2 \times .00886524 \times .3968085 \times 6739.5 &= &= -47,416.61 \\
 &- \underline{13 47} &= &= -.00886524 \times 1034.5178 &= &= -4,171.249 \\
 &&&&&= &= -56,587.71 \text{ g}
 \end{aligned}$$

$$U_{Q_2 R_2} = 0 = U_{Q_2 R_3}$$



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$$\begin{aligned} \underline{UR_1} &= -212 & = \\ &+ 213(19-23) & = \\ &+ 6 \underline{304} & = \\ &+ 47(50-53) & = \end{aligned}$$

$$\begin{aligned} &= -81,923.32 \\ &= 99,068.35 \\ &= 230,233.27 \\ &= \underline{\hspace{2cm}} \\ &= 247,378.30 \text{ g} \end{aligned}$$

$$\begin{aligned} \underline{UR_1^t} &= 2 \underline{20^2 305} & = \\ &+ 2 \underline{20 213} & = \\ &+ \underline{304} & = \\ &+ 2 \underline{47 51} & = \\ &+ 2 \underline{51^2 306} & = \end{aligned}$$

$$\begin{aligned} &= 30,341.09 \\ &= -113,787.08 \\ &= 318,490.04 \\ &= \underline{\hspace{2cm}} \\ &= \underline{\hspace{2cm}} \\ &= 235,044.05 \text{ g} \end{aligned}$$

$$\begin{aligned} \underline{UR_2} &= -13 \underline{47} & = \\ &- 2 \underline{13 51 306} & = \end{aligned}$$

$$\begin{aligned} &= \underline{\hspace{2cm}} \\ &= \underline{\hspace{2cm}} \\ &= \underline{\hspace{2cm}} \end{aligned}$$

$$\begin{aligned} \underline{UR_1} &= 2 \underline{20^2 305} & = \\ &+ 2 \underline{20 213} & = \\ &+ \underline{304} & = \\ &+ 47(20+51) & = \\ &+ 2 \underline{20 51 306} & = \end{aligned}$$

$$\begin{aligned} &= 30,341.09 \\ &= -113,787.08 \\ &= 318,490.04 \\ &= \underline{\hspace{2cm}} \\ &= \underline{\hspace{2cm}} \\ &= 235,044.05 \text{ g} \end{aligned}$$

$$\begin{aligned} \underline{UR_2} &= .5 \underline{213} & = \\ &+ .5 \underline{47} & = \end{aligned}$$

$$\begin{aligned} &= -71,688.92 \\ &= \underline{\hspace{2cm}} \\ &= -71,688.92 \text{ g} \end{aligned}$$

$$\underline{UR_3} = -304 = -318,490.04$$

$$\underline{UR_2^t} = 2 \underline{13^2 306} = 0$$

$$\begin{aligned} \underline{UR_2 R_1} &= -2 \underline{13 20 306} & = \\ &- 13 \underline{47} & = \end{aligned}$$

$$\begin{aligned} &= \underline{\hspace{2cm}} \\ &= \underline{\hspace{2cm}} \\ &= \underline{\hspace{2cm}} \end{aligned}$$

$$\underline{UR_2 R_2} = 0 = \underline{UR_2 R_3}$$

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Same as 742.5-A except
that struts have zero
stiffness.
 $\infty = 26 = 27 = 302 = 303$

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$$\begin{aligned} \underline{UR}_1^2 &= + 302 &= \infty \\ &+ 2 \underline{20}^2 \underline{305} &= \\ &+ 2 \underline{20} \underline{213} &= \\ &+ 304 &= \\ &+ 2 \underline{20} \underline{47} &= \\ &+ 2 \underline{20}^2 \underline{306} &= \end{aligned}$$

$$\begin{aligned} &= \infty \\ &= \\ &= \\ &= \\ &= \\ &= \\ &= \end{aligned}$$

$$\underline{UR}_1 R_2 = +.5(\underline{213} + \underline{47}) = -71,688.92 \text{ g}$$

$$\underline{UR}_1 R_3 = -304 = -318,490.04$$

$$\begin{aligned} \underline{UR}_2^2 &= + 303 &= \infty \\ &+ .5(\underline{305} + \underline{306}) &= \end{aligned}$$

$$\begin{aligned} &= \infty \\ &= \\ &= \end{aligned}$$

$$\underline{UR}_2 R_3 = -213 - 47 = 143,377.53 \text{ g}$$

$$\underline{UR}_3^2 = 2 \underline{304} = 636,980.08 \text{ g}$$

$$\begin{aligned} 0 &= u_1 + \underline{UR}_1^2 R_1 + \underline{UR}_1 R_2 R_2 + \underline{UR}_1 R_3 R_3 \\ 0 &= u_2 + \underline{UR}_2 R_1 R_1 + \underline{UR}_2^2 R_2 + \underline{UR}_2 R_3 R_3 \\ 0 &= u_3 + \underline{UR}_3 R_1 R_1 + \underline{UR}_3 R_2 R_2 + \underline{UR}_3^2 R_3 \end{aligned}$$

$$u_i = \underline{UR}_1 P_i + \underline{UR}_2 Q_i + \underline{UR}_3 R_i + \underline{UR}_4 Q_i$$

$$0 = u_1 + \infty R_1 - .007168892 R_2 - .031849004 R_3 \quad \therefore R_1 = 0$$

$$0 = u_2 - .007168892 R_1 + \infty R_2 + .014337783 R_3 \quad \therefore R_2 = 0$$

$$0 = u_3 - .031849004 R_1 + .014337783 R_2 + .063698008 R_3$$

$$\therefore R_3 = -15.69907 u_3$$

$$\begin{aligned} &= -15.69907 (.021308824 P_1 + .024737830 P_2 - .031849004 Q_1) \\ &= \underline{-.3345287 P_1 - .3883609 P_2 + .5000000 Q_1} \end{aligned}$$

Next page is 742.5-F15B



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$$-42.54521 U_1 - 139.4915 U_2 - .3669791 R_2 = .6449774 R_3$$

$$\begin{aligned} R_3 &= -65.96384 U_1 - 216.27334 U_2 - .5689794 R_2 \\ &= -65.96384 U_1 - 216.27334 U_2 + 357.83297 U_2 - 80.54465 U_3 \\ &= -65.96384 U_1 + 141.55963 U_2 - 80.54465 U_3 \end{aligned}$$

$$\begin{aligned} R_1 &= 31.39815 U_3 + .4501799 R_2 + 2 R_3 \\ &= + 31.39815 U_3 \\ & - 283.11958 U_2 + 63.727408 U_3 \\ &= -131.92768 U_1 + 283.11926 U_2 - 161.089300 U_3 \\ &= -131.92768 U_1 - .00032 U_2 - 65.96374 U_3 \end{aligned}$$

$$\begin{aligned} U_1 &= -.016257921 P_1 - .017972408 P_2 + .023504405 Q_1 \\ U_2 &= .004698620 P_1 + .005666020 P_2 - .007168892 Q_1 \\ U_3 &= .021308824 P_1 + .024737830 P_2 - .031849004 Q_1 \end{aligned}$$

$$\begin{aligned} R_1 &= +2.144870 P_1 + 2.371058 P_2 - 3.100882 Q_1 \\ &= .000001 P_1 - .000001 P_2 + .000002 Q_1 \\ &= -1.405610 P_1 - 1.621800 P_2 + 2.100879 Q_1 \\ &= .739259 P_1 + .739257 P_2 - 1.000001 Q_1 \end{aligned}$$

$$\begin{aligned} R_2 &= -2.954977 P_1 - 3.563378 P_2 + 4.508539 Q_1 \\ &= +3.016474 P_1 + 3.501884 P_2 - 4.508541 Q_1 \\ &= .061497 P_1 - .061494 P_2 \end{aligned}$$

$$\begin{aligned} R_3 &= +1.072435 P_1 + 1.185528 P_2 - 1.550441 Q_1 \\ &= +.665135 P_1 + .802080 P_2 - 1.014826 Q_1 \\ &= -1.716312 P_1 - 1.992500 P_2 + 2.565267 Q_1 \\ &= .021258 P_1 - .004892 P_2 \end{aligned}$$



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$$\begin{aligned} \delta_1 &= \underline{UP_1^2 P_1} + \underline{UP_1 P_2 P_2} + \underline{UP_1 Q_1 Q_1} + \underline{UP_1 Q_2 Q_2} + \underline{UP_1 R_1 R_1} + \underline{UP_1 R_2 R_2} + \underline{UP_1 R_3 R_3} \\ \delta_2 &= \underline{UP_2^2 P_1} + \underline{UP_2^2 P_2} + \underline{UP_2 Q_1 Q_1} + \underline{UP_2 Q_2 Q_2} + \underline{UP_2 R_1 R_1} + \underline{UP_2 R_2 R_2} + \underline{UP_2 R_3 R_3} \\ \delta_3 &= \underline{UP_3 Q_1 P_1} + \underline{UP_3 Q_2 P_2} + \underline{UQ_1^2 Q_1} + \underline{UQ_1 Q_2 Q_2} + \underline{UQ_1 P_1 P_1} + \underline{UQ_1 P_2 P_2} + \underline{UQ_1 P_3 P_3} \\ \delta_4 &= \underline{UQ_2 P_1 P_1} + \underline{UQ_2 P_2 P_2} + \underline{UQ_2 Q_1 Q_1} + \underline{UQ_2^2 Q_2} + \underline{UQ_2 R_1 R_1} + \underline{UQ_2 R_2 R_2} + \underline{UQ_2 R_3 R_3} \end{aligned}$$

$$\begin{aligned} \delta_1 &= .011359987 P_1 + .012420153 P_2 - .016266994 Q_1 - .0000000752573 Q_2 \\ &- .011420760 P_1 - .011420720 P_2 + .01544876 Q_1 - .0000115383 Q_2 \quad \text{---+} \\ &+ .000237597 P_1 - .000237564 P_2 \quad \text{---} \\ &+ .000110861 P_1 - .000550894 P_2 + .000533752 Q_1 + .00000756446 Q_2 \quad \text{+---+} \\ &= .00028769 P_1 + .00021098 P_2 - .0002845 Q_1 - .0000040491 Q_2 \end{aligned}$$

$$\begin{aligned} \delta_2 &= .012420153 P_1 + .012420153 P_2 - .017481481 Q_1 - .0000000752573 Q_2 \\ &- .01262477 P_1 - .01262473 P_2 + .017077345 Q_1 - .0000127547 Q_2 \quad \text{---+} \\ &+ .000286908 P_1 - .0002868 P_2 \quad \text{---} \\ &+ .000128649 P_1 - .000637483 P_2 + .00061954 Q_1 + .000008780522 Q_2 \quad \text{+---+} \\ &= .00021098 P_1 + .00028769 P_2 - .0002845 Q_1 - .0000040491 Q_2 \end{aligned}$$

$$\begin{aligned} \delta_3 &= -.016266994 P_1 - .017981481 P_2 + .023516957 Q_1 + .0000001040758 Q_2 \\ &+ .016510996 P_1 + .016510949 P_2 - .022334297 Q_1 + .0000166810 Q_2 \quad \text{++---} \\ &- .000362783 P_1 + .000362733 P_2 \quad \text{---} \\ &- .000165689 P_1 + .000823346 P_2 - .000797723 Q_1 - .0000113055 Q_2 \quad \text{+---} \\ &= -.0002845 P_1 - .0002845 P_2 + .0003849 Q_1 + .000005480 Q_2 \end{aligned}$$

$$\begin{aligned} \delta_4 &= -.00000007526 P_1 - .00000007526 P_2 + .000000104076 Q_1 + .000000105935 Q_2 \\ &- .00000397392 P_1 - .00000397391 P_2 + .000005375487 Q_1 - .00000004015 Q_2 \quad \text{---+} \\ &= -.0000040491 P_1 - .0000040491 P_2 + .0000054796 Q_1 + .000000010192 Q_2 \end{aligned}$$

Check: When $P_1 = P_2 = P$ and $Q = \delta_3 = Q_2$,

$$0 = -.0005690 P + .0003849 Q$$

$$\therefore Q_1 = 1.4783 P$$

$$\text{Then } \delta_1 = P(.00028769 + .00021098 - .0002845 \times 1.4783) = .00007809 P$$

When $-P_1 = P_2 = -P$, and $Q = Q_1 = Q_2$,

$$\delta_1 = P(.00028769 - .00021098) = .00007671 P. \quad \text{OK}$$



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$$\begin{aligned} \delta_1 &= \underline{UP_1^2 P_1} + \underline{UP_1 P_2} + \underline{UP_1 Q_1} + \underline{UP_1 Q_2} + \underline{UP_1 R_1} + \underline{UP_1 R_2} + \underline{UP_1 R_3} \\ \delta_2 &= \underline{UP_2 P_1} + \underline{UP_2^2 P_2} + \underline{UP_2 Q_1} + \underline{UP_2 Q_2} + \underline{UP_2 R_1} + \underline{UP_2 R_2} + \underline{UP_2 R_3} \\ \delta_3 &= \underline{UP_3 P_1} + \underline{UP_3 P_2} + \underline{UP_3^2 Q_1} + \underline{UP_3 Q_2} + \underline{UP_3 R_1} + \underline{UP_3 R_2} + \underline{UP_3 R_3} \\ \delta_4 &= \underline{UP_4 P_1} + \underline{UP_4 P_2} + \underline{UP_4 Q_1} + \underline{UP_4^2 Q_2} + \underline{UP_4 R_1} + \underline{UP_4 R_2} + \underline{UP_4 R_3} \end{aligned}$$

$$\begin{aligned} \delta_1 &= .011353427 P_1 + .012413593 P_2 - .016257921 Q_1 \\ &- .012018814 P_1 - .012018782 P_2 + .016257921 Q_1 \\ &+.0002889510 P_1 - .0002889369 P_2 \\ &+.0004529830 P_1 - .0001042428 P_2 \\ &= .000076547 P_1 + .00001631 P_2 \end{aligned}$$

$$\begin{aligned} \delta_2 &= .012413593 P_1 + .013832214 P_2 - .017972408 Q_1 \\ &- .013286264 P_1 - .013286228 P_2 + .017972408 Q_1 \\ &+.000348443 P_1 - .000348426 P_2 \\ &+.000525877 P_1 - .000121017 P_2 \\ &= .00001649 P_1 + .000026543 P_2 \end{aligned}$$

$$\begin{aligned} \delta_3 &= -.016257921 P_1 - .017972408 P_2 + .023504405 Q_1 \\ &+.017375843 P_1 + .017375796 P_2 - .023504405 Q_1 \\ &- .000440865 P_1 + .000440844 P_2 \\ &- .000677046 + .000155605 P_2 \\ &= \underline{\text{Zero}} \end{aligned}$$

$$\delta_4 = \underline{\text{Zero}}$$

Check: When $P_1 = P_2 = P$, $\delta_1 = .00007818 P$ ok.
When $-P_1 = P_2 = -P$, $\delta_1 = .00007491 P$ ok.



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$$\begin{aligned} \delta_1 &= \frac{UP^2}{P} + \frac{UPR_1}{P_2} + \frac{UPR_2}{P_2} + \frac{UPR_3}{P_2} + \frac{UPR_4}{P_2} + \frac{UPR_5}{P_2} + \frac{UPR_6}{P_2} + \frac{UPR_7}{P_2} \\ \delta_2 &= \frac{UPR_1}{P_2} + \frac{UPR_2}{P_2} + \frac{UPR_3}{P_2} + \frac{UPR_4}{P_2} + \frac{UPR_5}{P_2} + \frac{UPR_6}{P_2} + \frac{UPR_7}{P_2} \\ \delta_3 &= \frac{UPR_1}{P_2} + \frac{UPR_2}{P_2} + \frac{UPR_3}{P_2} + \frac{UPR_4}{P_2} + \frac{UPR_5}{P_2} + \frac{UPR_6}{P_2} + \frac{UPR_7}{P_2} \\ \delta_4 &= \frac{UPR_1}{P_2} + \frac{UPR_2}{P_2} + \frac{UPR_3}{P_2} + \frac{UPR_4}{P_2} + \frac{UPR_5}{P_2} + \frac{UPR_6}{P_2} + \frac{UPR_7}{P_2} \end{aligned}$$

$$\begin{aligned} \delta_1 &= +.011353427 P_1 + .012413593 P_2 - .016257921 Q_1 \\ &\quad - .007128413 P_1 - .008275514 P_2 + .010654412 Q_1 \\ &= .00422501 P_1 + .004138079 P_2 - .005603509 Q_1 \end{aligned}$$

$$\begin{aligned} \delta_2 &= .012413593 P_1 + .013832214 P_2 - .017972408 Q_1 \\ &\quad - .008275514 P_1 - .009607206 P_2 + .012368915 Q_1 \\ &= .004138079 P_1 + .00422501 P_2 - .005603493 Q_1 \end{aligned}$$

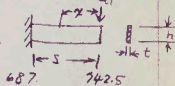
$$\begin{aligned} \delta_3 &= -.016257921 P_1 - .017972408 P_2 + .023504405 Q_1 \\ &\quad + .010654406 P_1 + .012368908 P_2 - .015924502 Q_1 \\ &= -.005603515 P_1 - .005603500 P_2 + .007579903 Q_1 \end{aligned}$$

Checks: 1. When $P_1 = P_2 = P$, and $Q = Q_1$, $\delta_1 = .00836309 P$ (119.6 #/in.)
 $-P_1 = P_2 = -P$, $Q = Q_1$, $\delta_1 = .000086931 P$ (1,503. #/in.)

2. When $P_1 = P_2 = P$, and $\delta_3 = 0$, $Q_1 = 1.47852 P$
 then, $\delta_1 = (.00836309 - .00828490) P = .00007819 P$ (12,789 #/in.)

3. If $-Q_1$ is the force at the end of a cantilever beam,

$$\alpha = \frac{Q}{EI} \int y dA = \frac{th}{\frac{t}{2} h^3 t} \times \frac{th}{2} h = 1.5 ; S = 55.5''$$



$$U = \int_0^S \left\{ \frac{Q^2 x^2}{2EI} + \frac{\alpha Q^2 x}{2AG} \right\} dx = Q^2 \left(\frac{S^3}{6EI} + \frac{\alpha S}{2AG} \right)$$

Suppose $h = 2(Q_1 + Q_2) = 61.4''$
 and $t = .031''$

$$\frac{\partial U}{\partial Q} = -\delta_3 = Q_1 \left(\frac{S^3}{3EI} + \frac{\alpha S}{AG} \right) = Q_1 \left(\frac{12 \times 55.5^3 \times 32}{3 \times 10^6 \times 61^3} + \frac{1.5 \times 55.5}{\frac{t}{2} \times 4 \times 10^8} \right) = .00002056 Q_1$$

Then $-.0112070 P = -.007579903 P - .00002056 Q_1 = -.007600463 Q_1$
 $Q_1 = 1.474516 P$

Then, $\delta_1 = (.00836309 P - .00826246 P) = .00010063 P$ (9,937 #/in.)



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Referring to the equations on page 742.5-F15, considering the force Q_1 due to the skin underneath the former as described on page 742.5-F15B, and letting $Q_2 = 0$, $0 = \delta_2 - \delta_3$

$$-.0002845P - .0002845P + .0003849Q_1 + .00002056Q_1 = 0$$

$$Q_1 = 1.4032P$$

$$\delta_1 = P(.00028769 + .00021098 - .00039921) = .00009946P$$

(10,054 #/in.)

Partial Summary of Results: Station 742.5

Conditions: $F_1 = F_2 = P$; $Q_2 = 0$.

I_{F0}	I_{SP}	A_{ST}	t	Spec.	γ/δ
.440	9.42	.3136	0	$0 = Q_1 = Q_2$	2,005. #/in.
"	"	"	$\frac{1}{32}$	$\delta_2 = -.00002056Q_1$	10,054.
"	"	"	∞	$\delta_3 = 0$; 0	12,806.
"	∞	∞	0	$Q_1 = 0 = Q_2$	12,790.
"	∞	0	0	$Q_1 = 0 = Q_2$	119.
"	∞	0	∞	$\delta_3 = 0$.	12,789.
"	∞	0	$\frac{1}{32}$	$\delta_2 = -.00002056Q_1$	9937.

SNAP-LOCK BINDER, 1½ inch capacity

NO. 4872-W

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THE CHAS. CHAPMAN CO. LIMITED
London - Canada

